

**FUNCTIONAL SPECIFICATION
FOR HLCAC
PROPULSOR ASSEMBLY**

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P.O. # N61331-03-F-1609

Data Item A002

THIS SPECIFICATION HAS BEEN PREPARED BASED ON A CFD ANALYSIS AND THE RESULTING DESIGN ANALYSES CONDUCTED BY CDI MARINE SDD. IT MAY BE UPDATED AS A RESULT OF THE MODEL AND FULL-SCALE FAN TEST RESULTS.

FOREWORD

The work described in this working paper was prepared by CDI Marine Systems Development Division (CDIM-SDD). This document presents a functional specification for the propeller shroud assembly for the HLCAC hovercraft.

CDIM-SDD points of contact for this task were Mr. Brian Forstell and Mr. Alan Becnel.

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1.0 SCOPE

This specification presents the physical and performance requirements for a propeller assembly and for a propeller shroud assembly that will be employed on the HLCAC. These two elements compose the propulsor assembly and will provide primary propulsion for the craft. This specification also defines the geometry of the propeller shroud assembly through the inclusion of figures and tables of offsets as necessary.

For the purposes of this specification, a propeller shroud assembly consists of the propeller shroud, two aft rudders, seven stators connecting the shroud to the propeller hub centerbody, and the foundations by which the propeller shroud is attached to the deck. The propeller assembly consists of the propeller hub, composite propeller blades, a control module, a transfer housing, oil transfer tubes, a propeller support shaft, a drive adapter and a spinner. All components of the propeller assembly shall be documented by manufacturer-supplied interface control drawings.

2.0 APPLICABLE DOCUMENTS

Documents are applicable to the extent they are referred to in Sections 3.0, 4.0 and 5.0 of this specification. Unless stated otherwise, the current revision of the following documents will apply.

2.1 Government Documents

HLCAC Propulsor Interface Control Document.

2.2 Other Documents

INS 0041 – Subcontractors requirements (NFT document).

ISO 1940/73/05/01 – Mechanical vibration. Balance quality requirements of rigid rotors.

A Handbook of Balancing, AiResearch Mfg. Co., 1960.

An Introduction to Fatigue in Metals and Composites, R.L. Carlson and G.A. Kardomateas, Chapman & Hall, 1996.

MIL-T-31000 – Military Specification, Technical Data Package

Drawings:

Dwg. #755-05 HLCAC Propulsor

3.0 REQUIREMENTS

3.1 Geometry

The HLCAC hovercraft includes two aft-mounted propulsor assemblies, one to port and the other to starboard.

The geometry of a single propulsor assembly is defined in the following paragraphs. Unless otherwise noted, the port and starboard propulsors are identical.

3.1.1 General Arrangement

Figures 3-1 and 3-2 show the general arrangement of the propeller, propeller shroud, stators, and rudders which make up the propulsor assembly. Detailed geometry is presented in the following sections.

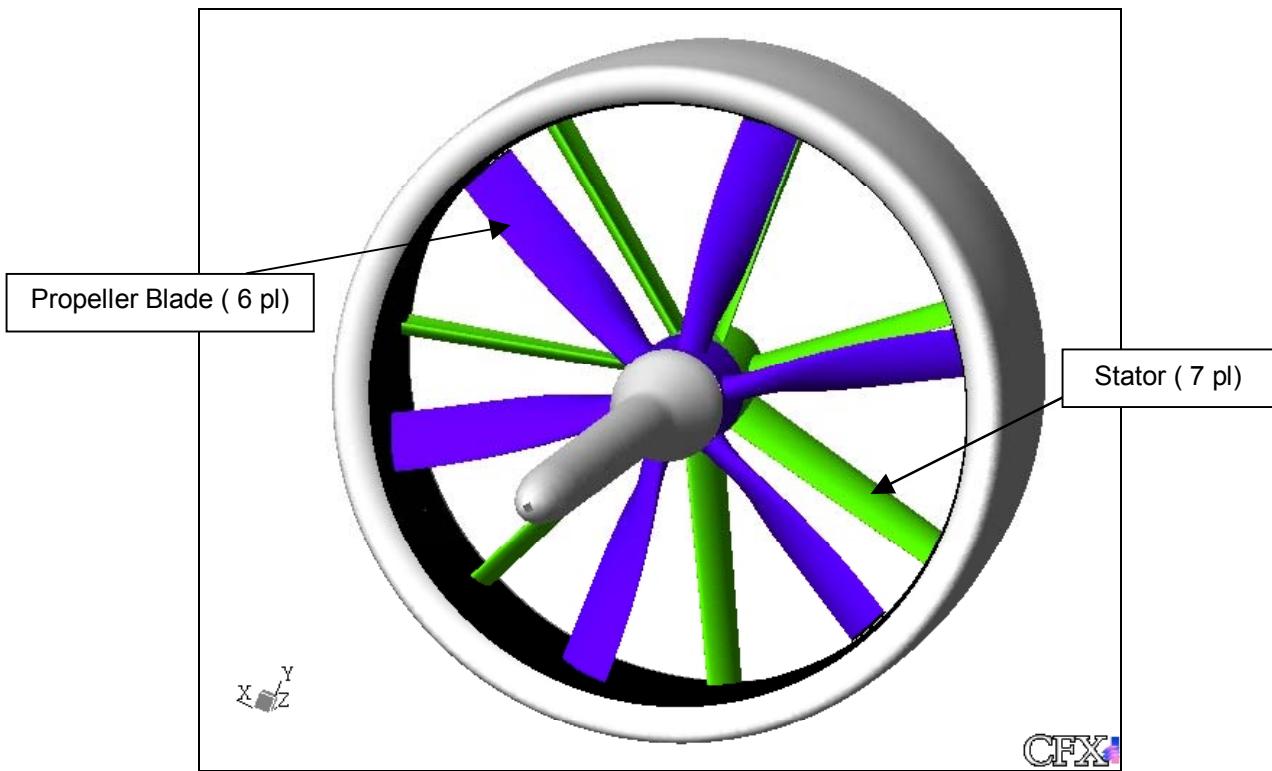


Figure 3-1. General Arrangement of the Propulsor Assembly (Port Assembly, View From Front)

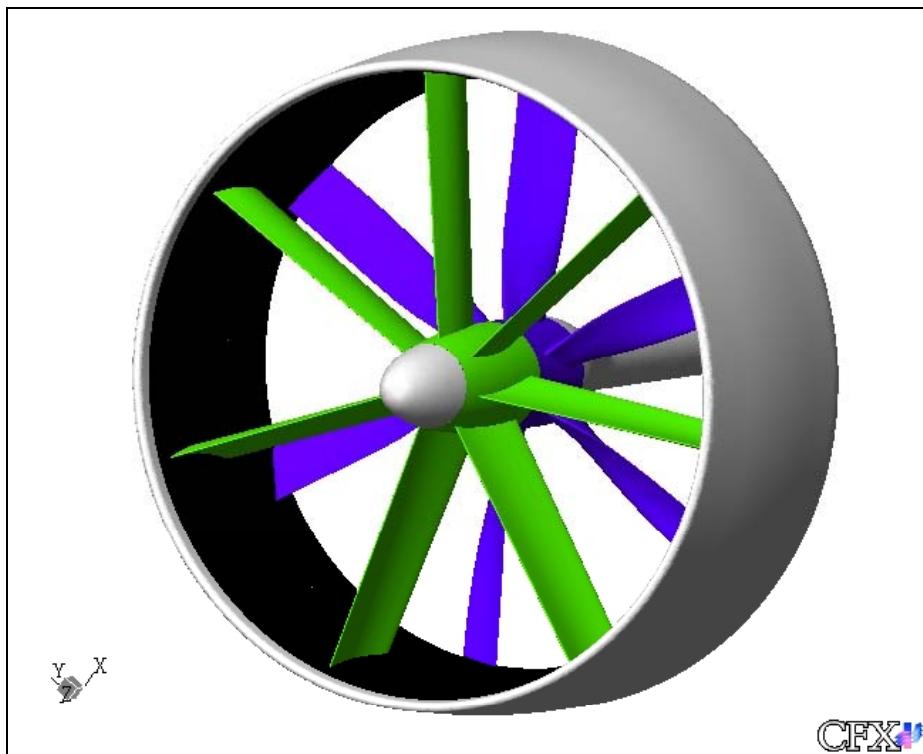


Figure 3-2. General Arrangement of the Propulsor Assembly (Port Assembly, View From Aft)

3.1.1.1 Propeller

Both the port and starboard propellers have a right-handed rotation. The propeller centerline is located 355.6 mm (14 inches) aft of the leading edge of the propeller shroud. The maximum diameter of the propeller shall be 3581.4 mm (141 inches) at the maximum operating condition. This maximum diameter should include the effects of the centrifugal load, thermal expansion, and all manufacturing tolerances. The detailed offsets for the propeller blade at different spans are presented in Appendix A.

3.1.1.2 Shroud

The shroud is a cylindrical airfoil surrounding the propeller. Figure 3-3 presents a cross-section of the shroud. The shroud has an inner diameter of 3606.8 mm (142 inches), providing a nominal 12.7 mm (0.5-inch) tip clearance with a 3581.4 mm (141 inch) six-bladed propeller. The upper and lower surfaces of the shroud inlet lip are elliptical. The two elliptical surfaces meet at the shroud leading edge radius of 1998.9 mm (78.695 inches). The outer elliptical surface shall extend to 116.8 mm (4.6 inches) aft of the leading edge, and the inner elliptical surface shall extend to 236.2 mm (9.3 inches) aft of the leading edge.

The outer surface of the shroud contains a circular arc transition section starting 482.6 mm (19 inches) aft of the leading edge. The arc radius is 5993.4 mm (235.96 inches) and transitions to a straight line 990.6 mm (39 inches) aft of the leading edge.

The inner surface of the shroud also contains a circular arc transition, starting 609.6 mm (24 inches) aft of the leading edge with a radius of 751.8 mm (29.6 inches). This transitions to a straight line section 736.6 mm (29 inches) aft of the leading edge. Appendix F provides a table of shroud offset points.

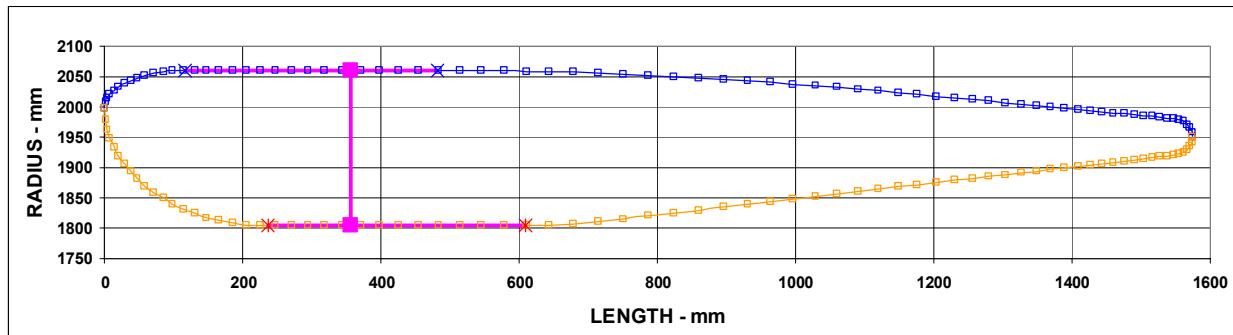


Figure 3-3. Propeller Shroud Airfoil Section

3.1.1.3 Stators

There are seven identical stator blades located aft of each propeller. The stators provide support for the propeller centerbody and carry the thrust and torque produced by the propeller out to the shroud where it is reacted to the deck. The stators are equally spaced at 0, 51.43, 102.86, 154.29, 205.71, 257.14, and 308.57 degrees, with 0 degrees representing top dead center. The geometry of an individual stator blade at the shroud and hub intersections (100 and 0 percent span) is shown in Figure 3-4. The geometry of the stator varies along the span, but the geometry is a ruled section between the presented hub and tip sections. The detailed offsets for the stator blade at different spans are presented in Appendix B. Appendix G provides a table of offsets to define the prop and stator hub shape that is needed to match the 0-percent span sections of both.

The (0, 0, 0) point of the stator is the centerline of the propeller, which is 355.6 mm (14 inches) from the leading edge of the shroud.

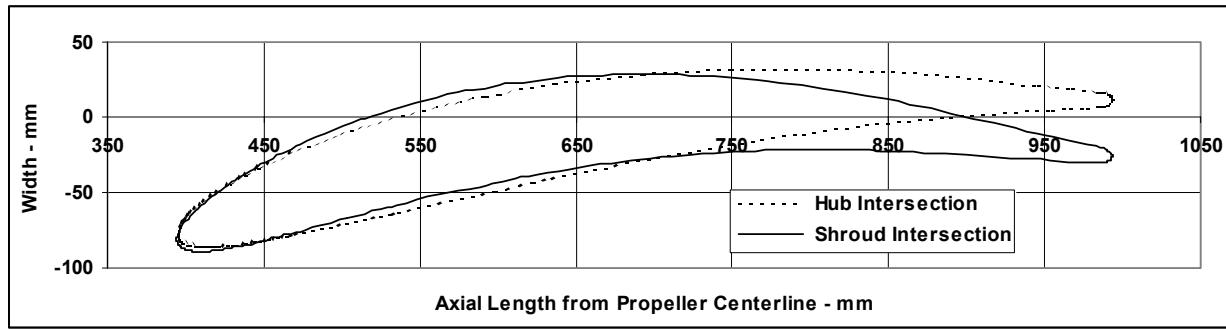


Figure 3-4. Stator Blade Geometry

3.1.1.4 Rudders

There are two rudders located behind each propeller shroud. The two rudders will be Government Furnished Equipment and are described in the HLCAC Propulsor Interface Control Document.

3.2 Performance Characteristics

3.2.1 General Performance

Performance characteristics have been determined at an ambient temperature of 37.7°C (100°F) and a nominal propeller rotational speed of 1257 rpm. All performance has been analyzed and predicted under government sponsorship and has resulted in the propeller blade section, blade geometry, and shroud geometry that is documented in this functional specification.

3.2.2 Minimum Rating

The aerodynamic performance rating of the propeller when operating in the duct, whose nominal discharge area ratio is 1.2, has been determined by the government. The government analysis has concluded that the bollard thrust of the propulsor unit will be at least 87 kN (19,556 lb) static thrust at 5057 kW (6781 shp) with an ambient air temperature of 37.3°C (100°F) at sea level and still air, a propeller speed of 1257 rpm, and a propeller pitch not to exceed 40 degrees measured at the three-quarter radius position with erosion protection system installed.

3.2.3 Other Guaranteed Rating Points

Table 3-1 provides the two primary aerodynamic performance rating points that have been evaluated by the government.

3.2.4 Attitude Conditions

The propeller system shall meet the performance requirements of the specification under the following conditions:

- Level craft position and the craft rolled 5 degrees to either side.
- Level craft position and the craft rolled 20 degrees to either side for 30 seconds.
- Craft at 0 to 12 degrees pitch in either direction with up to 5 degrees roll to either side.
- Craft at 0 to 12 degrees pitch in either direction with up to 20 degrees roll to either side for 30 seconds.

Table 3-1
Required Performance Points

Nominal Thrust at Bollard Conditions	
Thrust, kN	87
Propeller Input Power, kW	5057
Propeller Speed, rpm	1257
Air Temperature, °C	37.7
Relative Air Speed, kts	0
Nominal Thrust at Cruise Conditions	
Thrust, kN	76.5
Propeller Input Power, kW	5057
Propeller Speed, rpm	1257
Air Temperature, °C	37.7
Relative Air Speed, kts	25

3.2.5 Air Speed and Altitude

The propeller system shall perform in accordance with the requirements of the specification at sea level for relative air velocities up to 75 kts and ambient temperatures from -23 to 37.7°C.

3.2.6 Power Profile

For design purposes, the power profile for the HLCAC propeller shall be as defined in Table 3-2.

Table 3-2
Operational Load Spectrum

	Case 1 Idle	Case 2 Low	Case 3 Cruise
Propeller Power, kW	100	1300	5057
Speed, kts	0	5	25
Time at Load Case L, %	15	10	75

3.2.7 Rotational Speed

The steady speeds of rotation for the propeller system shall be as follows:

- a) Normal: Up to 1257 rpm
- b) Maximum: Up to 1313 rpm

3.2.8 Over Speed

The propeller shall be capable of withstanding an over speed to 1400 rpm for at least 10 seconds.

3.2.9 Crosswind Performance

The Aq (the angle of the air approaching the propeller duct inlet plane due to the combination of cross wind and craft forward speed multiplied by its dynamic pressure) shall correspond to a vehicle forward speed of 25 kts in the presence of a 16-kt cross wind. At 1257 rpm and 5057 kW (6781 shp), at sea level and 37.7°C (100°F), the estimated shaft bending moment and the estimated normal force shall be determined and provided to the USN.

3.2.10 Blade Activity Factor

The calculated blade activity factor was determined by the government analysis, determined in conjunction with the blade section properties to meet the specified performance. The blade design was based on using a total activity factor of 1002.

3.2.11 Polar Moment of Inertia

The contractor is responsible for calculating the polar moment of inertia of the propeller about the axis of propeller rotation.

3.2.12 Propeller System Life

The propeller system shall operate in the environment specified in Section 5.0 for a minimum of 6000 operating hours with the spectrum given in Table 3-2. The blades should provide a minimum of 500 hours of operation before servicing, except for replacement of leading edge protection that shall have a minimum of 250 hours of operation.

3.2.13 Propeller Loads

The propeller loads for the maximum power condition are shown in Appendix D. The averaged pressure distributions around the blade are tabulated for 0 to 100-percent span at 10-percent span increments.

3.2.14 Flutter

The propeller shall be free from flutter under all normal operating conditions that encompass static thrust, up through a maximum relative air speed of 75 kts.

3.2.15 Hydraulic Power

Loss of external hydraulic power to the propeller system shall result in the blades remaining at the pitch existing at the time of the failure when operating in the forward pitch range. If in reverse pitch, the blades shall move to a small negative angle established by the blade twisting moment associated with the propeller operating conditions.

3.2.16 Propeller System Weight

The propeller designer/builder is responsible for determining and reporting the weight of the propeller system.

3.2.17 Blade Pitch

The propeller shall incorporate means of adjusting the individual blade angle in increments of 0.1 degrees. The propeller shall either be so constructed that when the blades are assembled in the hub, the difference between their mean aerodynamic angle shall not exceed 0.1 degrees throughout the entire operating range as well as against the limiting pitch stops, or the propeller shall incorporate means of adjusting the individual blade angle in increments of 0.1 degrees.

The propeller control system shall function to provide control of the propeller in response to electrical signals from the craft control system. The control shall operate throughout the blade angle range specified. A block diagram of the required control system shall be provided and shown to be consistent with the HLCAC Propulsor Interface Control Document. The time from full forward to full reverse shall not exceed two seconds and the velocity gain shall not be less than 20 degrees per second.

3.2.18 Blade Angle Range

The maximum blade angle shall be 40° . The minimum blade angle shall be -35° .

3.3 Structural Design Requirements

The propulsor assembly shall satisfy the design criteria and loading requirements specified in this paragraph.

Loads are presented in metric units. When loads are presented as “ \pm ” cases, all possible combinations must be checked.

3.3.1 Propeller System Loads

3.3.1.1 Static Strength

The propeller system (propeller assembly, shaft adapter assembly, and associated mountings) shall withstand, without excessive deformation, yielding (i.e. 0.2% proof strain), or loss of utility, the loadings resulting from the critical combination of the following limit loads multiplied in each case by the yield factor of safety. In addition, the propeller system shall withstand, without failure, the loadings resulting from the critical combination of the following limit loads, multiplied in each case by the ultimate factor of safety:

a) Limit Loads

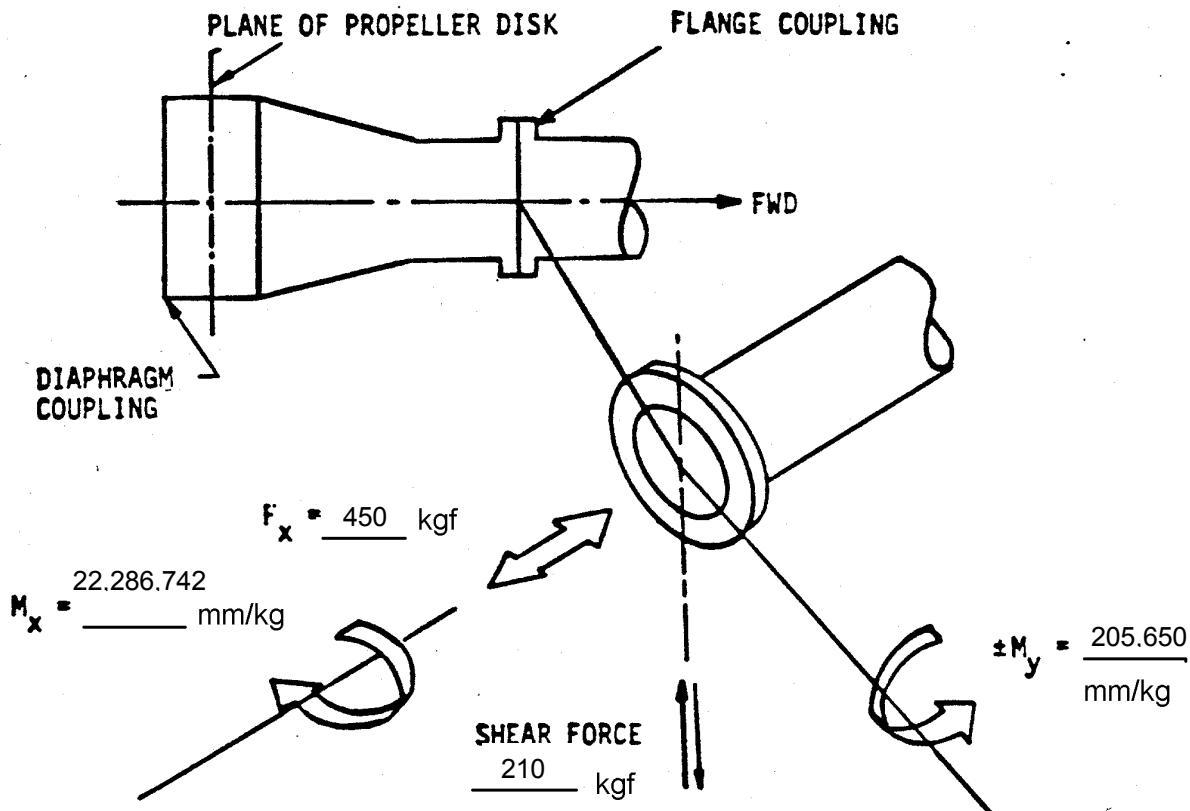
1. The limit torque of Figure 3-5.
2. The inertia load factors of Figure 3-6 multiplied by the mass of the components.
3. Loads due to static and dynamic deflections of the hull under hogging, sagging, and impact load conditions, in combination with the axial stiffness of associated couplings.
4. Loads due to thermal expansion differences between hull structure and propeller system components, in combination with the axial stiffness of the associated couplings.
5. Loads due to initial installation tolerances and interferences, in conjunction with the axial stiffness of the associated couplings.
6. Loads due to gyroscopics; at maximum-rated speed (rpm), the system shall withstand gyroscopic moments resulting from the following:
 - i. Craft rotational velocity of $\theta = \pm 0.5$ rad/sec (pitch) combined with the bow impact inertia loads of Figure 3-2.
 - ii. Craft rotational velocity of $\psi = \pm 0.5$ rad/sec (yaw) combined with inertia loads of $N_z = 1.0$ down, $N_y = \pm 1.76$, and $N_x = \pm 1.25$.
7. Thrust loads from propeller and bevel gears, where applicable.

b) Factors of Safety

1. Yield = $1.0 \times$ limit.
2. Ultimate = $1.5 \times$ limit.

CASE 1: Torsional Only - $M_x = 22,286,742 \text{ mm/kg}$ is subject to shaft criteria where equivalent limit load = 2 x Operational Steady State Maximum Torque.

CASE 2: Combined Limit Loads.



CASE 3: Fatigue Loads. $M_x = 22,286,742 \text{ mm/kg} \pm 11,143,371 \text{ mm/kg}$ for Infinite Life.

Note that Ultimate Load = 1.5 x Limit Load.

Figure 3-5. Shaft Adapter Loads

Inertia Factor			Design Condition
N_x	N_y	N_z	
-1.0	± 1.0	+4.6	Stern Impact
+2.0	± 1.0	-3.2	Bow Impact
+8.0, -3.0	± 3.0	+4.5, -3.0	Crash Case 3

1. Inertia load factor is defined as N where:

$$N = \frac{\text{Force on Body}}{\text{Weight of Body}} = -\frac{\text{Acceleration of Body}}{9}$$

2. Variations of loading due to the velocity inflow gradients shall be considered for static, dynamic, and fatigue requirement.

3) These are ultimate craft load factors to be applied separately for ultimate load design.

Positive Sign Convention

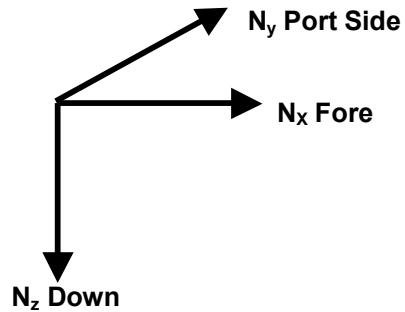


Figure 3-6. Inertia Load Factors

3.3.1.2 Repeated Load Strength

The propeller system shall withstand, without loss of structural integrity, the loads resulting from the most critical combination of the loadings. For substantiating fatigue life, allowable strengths shall be based on reference data with appropriate stress factors. Where applicable, allowable fatigue shall not be more than the average test specimen life divided by a factor of 2.0.

Analysis shall be conducted to show a positive margin of life after 6000 hours of craft operation. The maximum steady operating torque with $\pm 50\%$ alternating shall be used at the maximum power rating for the operational load spectrum of Table 3-2. In addition, the stresses resulting from craft inertia, gyroscopic, centrifugal thermal growth, hull deflections, installation tolerances, and thrust loads described in Section 3.3.1.1 shall be combined with, or superimposed on, the stresses resulting from torsional loading.

In lieu of the preceding, an analysis may be conducted to show infinite life for the application of the maximum steady operating torques with $\pm 50\%$ alternating, shown in Figure 3-5. The effects of maximum inertia loads, hull deflections, thermal expansions, installation tolerances, gyroscopic, centrifugal and thrust loads described in Section 3.3.1.1 shall be superimposed on, or combined with, the stresses from torsional effects.

3.3.1.3 Provisions

Provisions shall be made for attaching balance weights in one plane. Points of attachment must be readily accessible with the propeller mounted on the craft.

3.3.1.4 Drive Shaft Adapter

The drive shaft adapter shall be dynamically balanced. All balancing shall be accomplished by the removal of metal, and care shall be taken to prevent subsequent corrosion.

3.3.2 Load Cases

Table 3-3 identifies the required load cases. The shroud assembly shall be capable of withstanding the associated loads in accordance with the design criteria specified in Paragraph 3.3.3. Specific component loadings corresponding to these design cases are presented in Paragraphs 3.3.4 through 3.3.6. In many cases, the components must also react loads generated by other components, including the propeller. The details of the structural design will dictate how these additional loads are applied.

Table 3-3
Propulsion Duct Assembly Load Cases

Case	1	6		
Descriptor	Ahead, Bollard	Max Design Load Speed Case		
Ship Speed, Kts	0	25		
Engine Power Setting	MCR	MCR		
Propeller Power, kW	5056.6	5056.6		
Propeller Speed, rpm	1257	1257		
Rudder Defl., deg (Note 2)	± 20	± 20		
Sideslip Angle, deg (Note 2)	0	± 40		
Total Thrust, kN (lb)	86 (19,556)	76.5 (17,190)		
Propeller Thrust, kN (lb)	45.3 (10,178)	39.8 (8,947)		
Propeller Torque, N-M (lb-Ft)	38,413 (28,332)	38,413 (28,332)		
Duct Thrust, kN (lb)	41.7 (9,378)	36.7 (8,243)		
Temperature ($^{\circ}$ C)	37.8	37.8		
Ambient Pressure, Bar (psi)	1.014 (14.7)	1.014 (14.7)		
Seas	Calm	Max.		
Wave Slap	N/A	Yes		
Inertial Load (Note 3)	N/A	$+2.0, -1.0 \text{ N}_x,$ $\pm 1.0 \text{ N}_y, +4.6,$ -3.2 N_z		
Propeller Gyroscopic Moment, kN-M (in.-lb) (Note 4)	N/A	$\pm 8.2 (\pm 72,590)$		
Notes:				
1. Loading scenarios for this design case are presented the loads section for each component, where applicable.				
2. All combinations of \pm rudder angle and \pm sideslip angle should be checked for maximum side forces and generated moments. 20° represents the angle of maximum rudder force, not the maximum deflection angle.				
3. Sign convention is $+N_x$ forward, $+N_y$ to port, $+N_z$ down.				
4. Gyroscopic moments are present for both pitch rate and yaw rate motions, but will not occur simultaneously. The effect of gyroscopic moments shall be computed for both positive and negative moments. (Yaw moment occurs with pitch rate; pitch moment occurs with yaw rate).				

3.3.3 Design Criteria

Loads defined in this document represent limit loads, that is, the maximum load expected under the stated operating conditions. The propeller and shroud assembly shall withstand these limit loads subject to the design criteria contained within this paragraph. If, on the basis of the structural analysis of Paragraph 8.1.2, the contractor believes that other more severe loads exist that will drive the design, the contractor must inform the USN and provide justification for those other loads.

3.3.3.1 Design Margins

The following rules for load cases apply.

- Yield = 1.0 * Limit
- Ultimate = 1.5 * Limit
- Yield loads are defined as limit loads times the yield factor of safety, while ultimate loads are defined as the limit loads times the ultimate factor of safety. The load cases defined herein, unless otherwise noted, represent limit loads.
- When subjected to yield loads, the structure shall not interfere with normal operation of the craft. When subjected to ultimate loads, the structure shall not fail.

All static load calculations must show a positive margin of safety when the margins of safety are defined as:

$$M.S.(\text{Yield}) = \frac{\text{Allowable Yield Load (Stress)}}{\text{Design Yield Load (Stress)}} - 1$$

$$M.S.(\text{Ultimate}) = \frac{\text{Allowable Ultimate Load (Stress)}}{\text{Design Ultimate Load (Stress)}} - 1$$

Material properties shall be used to define allowable loads (stresses) for yield and ultimate conditions.

All fittings and joints shall have an additional special safety factor of at least 1.15 added to the safety factors above applied to the fitting, the means of attachment, and the bearing on the joined parts.

3.3.3.2 Fatigue Criteria

The margin of safety for fatigue shall be positive when calculated using the following:

$$M.S.(\text{Fatigue}) = \frac{\text{Material Fatigue Strength, at } 2N \text{ Cycles}}{\text{Design Load}} - 1$$

Material fatigue strength per the above definition shall depend on whether the fatigue load is applied as oscillations between plus and minus loading or between plus and zero loading (i.e. the appropriate R value must be considered). N is the cumulative number of cycles defined by the load spectrum and frequency. A factor of 2.0 is applied for margin. For composite materials, an equivalent definition of fatigue strength shall be used.

A craft operating life of 30 years with 200 operating hours per year shall be assumed. The operational load spectrum shall be as defined in Table 3-2. Specific information on fatigue loads is presented, where applicable, for each component.

3.3.3.3 Allowable Deflections and Deformations

The following requirements define the allowable deformations of the structure under loads:

- The maximum allowable axial deflection under load of the structure is 41 mm. The maximum deflection is defined at the uppermost point of the duct structure. Displacements at lower elevations shall be less than 41 mm.
- The maximum allowable eccentricity of the duct under load is 7 mm radial.
- All deflections of the structure shall be such that proper operation of the propeller, duct, and rudder are not hindered.

3.3.4 Duct Loads

The pressure distributions around the duct are determined for the maximum power case at the 25-knot design speed. Pressure distributions for the inside and outside of the duct are shown in Appendix C. The pressure distributions are shown for every 30 degrees of rotation around the duct.

Loads corresponding for each load case are presented in Table 3-3.

The duct shall react stator loads, rudder loads, and propeller thrust loads in addition to the loads defined in Table 3-4.

Table 3-4

Duct Loads

Case	1	2
Descriptor	Ahead, Bollard	Max Design Load Speed Case
Inertial Loads		
Shock Load	N/A	+2.0, -1.0 N _x , ±1.0 N _y , +4.6, -3.2 N _z
Wave Slap Loads (Note1)		
Wave Slap, kPa (lb/in ²)	0	Figure 3-7
Thrust Loads		
Duct Thrust, kN (lb)	41.7 (9,378)	36.7 (8,243)
Aerodynamic Loads (Note 2)		
Side Force, kN (lb)	0	52.27 (11,751)
Notes:		
1. Wave slap loads are applied normal to the surface of the duct. The wave slap load varies from a maximum at the lower surface to zero at the upper surface, as shown in Figure 3-7.		
2. Side force loads shall be assumed to act at the quarter chord of the shroud airfoil cross-section, at the level of the propeller longitudinal axis.		

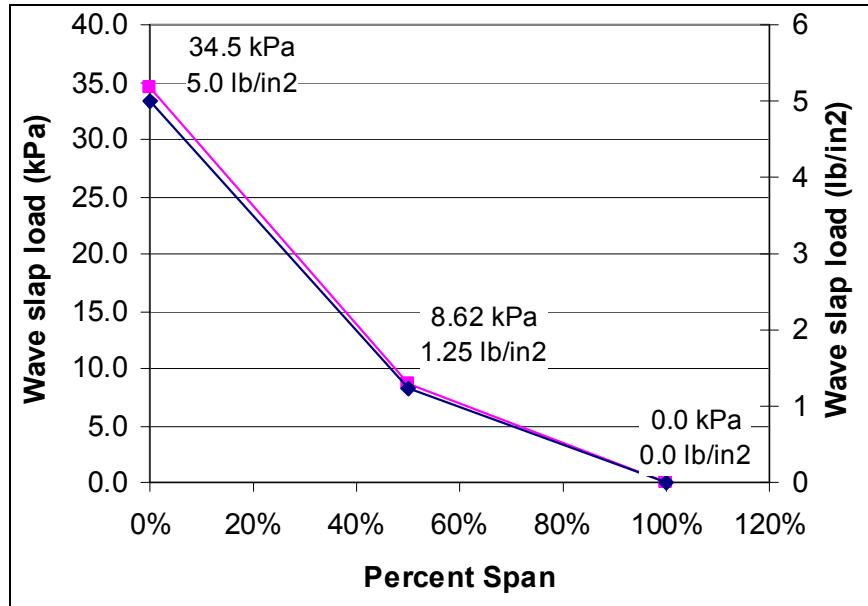


Figure 3-7. Duct Wave Slap Load Distribution

3.3.4.1 Duct Fatigue Loads

The duct will experience fatigue loading as a result of pressure pulses arising from the close passage of the propeller blade. Appendix C provides information on the peak blade passage pressures and the pressures midway between adjacent blades.

The frequency of the load will be at the primary blade passing frequency of 125.7 Hz (6P rate at 1257 rpm). Cycle life and operational loading shall be per that defined in Paragraph 3.3.3.2. (The idle case is assumed to not contribute to fatigue life.)

The oscillatory loads shall be applied in combination with the steady-state thrust loads as follows:

Lower Power:	Propeller Thrust = 24 kN	Duct Thrust = 20 kN
Maximum Power:	Propeller Thrust = 45.3 kN	Duct Thrust = 41.7 kN

3.3.5 Stator Loads

The pressure distributions for the stators are presented in Appendix E for the maximum power design point of 5056.6 kW and 25-knot craft speed. The pressure distributions represent the circumferential averaging of the pressures around the stator blades at different spans. Data is presented in both tabular form for 10-percent stator span increments and in graphic form for 0, 50, and 100-percent span. Table 3-5 summarizes additional stator loads.

3.3.5.1 Stator Fatigue Loads

Fatigue loading on the stators stems from the propeller crossing in front of the stator blade and from the varying thrust produced by each propeller blade during a single rotation (due to upstream flow blockage). Both loads will occur at a primary frequency of 125.7 Hz. Blade passage loads will oscillate between zero load and a maximum load (applied in addition to the steady aerodynamic loads on the stator). Propeller thrust variations will create a moment that oscillates between positive and negative maximum values.

These oscillatory stator loads must be included in the stress analysis of paragraph 8.1.2.

Table 3-5**Stator Loads**

Case	1	2
Descriptor	Ahead, Bollard	Max Design Speed Load Case
Inertial Loads		
Shock Load	0	+2.0, -1.0 N _x , ±1.0 N _y , +4.6, -3.2 N _z
Wave Slap Loads (Note 1)		
Wave Slap, kPa (lb/in ²)	0	Figure 3-8
Propeller Loads (Note 2)		
Thrust Per Stator, kN (lb)	12.94 (2908)	11.37 (2556)
Aerodynamic Loads (Note 3)		
Side Force Per Stator, kN (lb)	3.97 (893)	3.97 (893)
Thrust Force Per Stator, kN (lb)	1.89 (424)	1.66 (373)
Maintenance Loads (Note 4)		
Midspan Normal Load, kg (lb)	114 (250)	N/A
Axial Load, kg (lb)	114 (250)	N/A
Notes:		
1. For purposes of design commonality, all stators shall be designed to withstand the wave slap in Figure 3-8. The maximum wave slap occurs at the interface to the shroud, i.e., the lowest point of the stator located at bottom dead center. For purposes of analysis of the entire propeller shroud assembly under simultaneous loading, stator wave slap loads shall not be included. (The maximum duct wave slap load represents a normal impact, which would effectively block the stators from receiving the wave slap load). In Figure 3-8, 0% span represents the stator location closest to the shroud (for the stator located at bottom dead center).		
2. Propeller thrust loads stated in the table are twice the propeller thrust divided by the number of propeller blades. Each stator (and their connections with the shroud and propeller centerbody) shall be designed to withstand this 'twice average' load, but when performing analysis of the entire shroud assembly under simultaneous loading for the specified load cases, one half of this load should be applied to each stator, thus representing the actual propeller thrust load.		
3. Aerodynamic loads should be assumed to act at 70% of the span (with 0% at the centerbody interface and 100% at the duct interface) and at 25% of the chord.		
4. These loads are not applied during craft operation. However, they are included in the table to ensure the stator design can accommodate these loads, while the craft is non-operational. The axial load shall be applied to the connections between the stator and centerbody and the stator and shroud.		

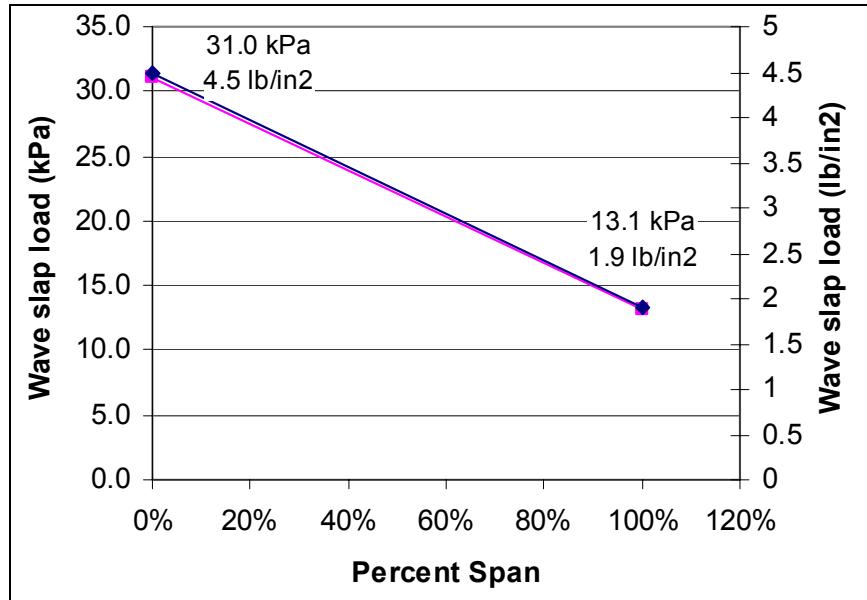


Figure 3-8. Stator Blade Wave Slap Load Distribution

3.3.6 Rudder Loads

The loads for the rudders are presented in Table 3-6. There are no critical fatigue loads for the rudder other than continued operation in the propeller wake for the operational life of the craft. The loads in Table 3-6 represent the loads on a single rudder (of the pair behind the propeller shroud). The design shall accommodate the loads created by two rudders deflected in tandem.

Table 3-6

Rudder Loads

Case	1	2	3	4
Descriptor	Ahead, Bollard	Astern, Bollard (Est)	Wave Slap @ Max Thrust	Max Design Speed Load Case
Inertial Loads				
Shock Load	N/A	N/A	N/A	+2.0, -1.0 N _x , ±1.0 N _y , +4.6, -3.2 N _z
Wave Slap Loads (Note 1)				
Wave Slap, kPa (lb/in ²)	0	0	Figure 3-9	0
Aerodynamic Loads (Note 2)				
Side Force, N (Per Single Rudder)	3333	4000	3333	3800
Drag Force, N (Per Single Rudder)	-102	2000	-102	117
Hinge Moment, N-m (Per Single Rudder)	158	1400	158	181
Rudder Defl., deg	± 20	± 20	± 20	± 20

Table 3-6 (continued)

Notes:

1. Wave slap shall be applied per Figure 3-9. For design purposes, all of the wave slap load shall be reacted by the lower rudder hinge alone. 0% span represents the lowest point on the rudder.
2. Aerodynamic loads shall be assumed to act at the quarter chord of the rudder, at midspan (except for the specified hinge moment). The hinge moment for the reverse bollard thrust load case does not imply a requirement to produce this hinge moment. The linkage shall be designed for ahead conditions, and will simply provide whatever it can produce in astern conditions, the listed value represents the maximum possible load. There is no operational requirement to maneuver the vehicle in astern conditions. Sixteen degrees represents the load corresponding to maximum rudder side force.

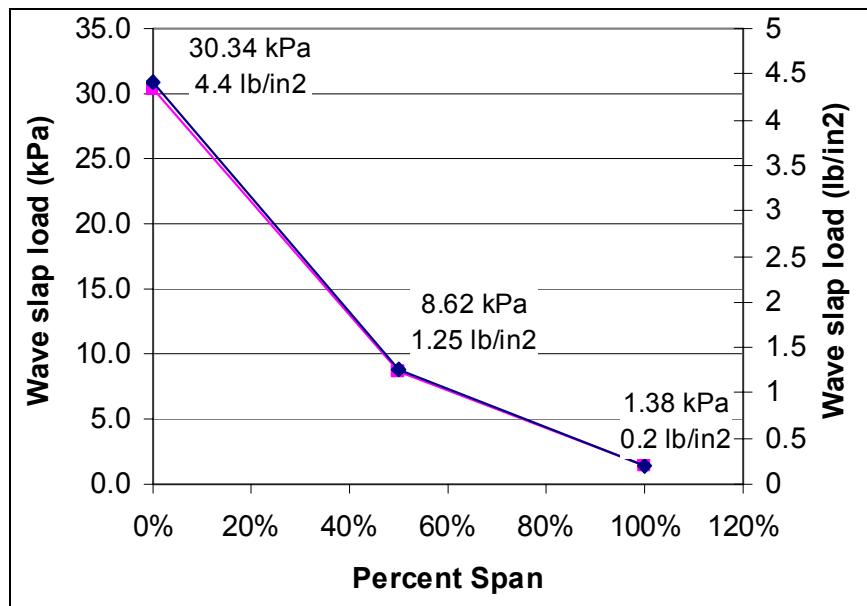


Figure 3-9. Rudder Wave Slap Distribution

3.3.7 Rudder Movement

3.3.7.1 Range of Deflection

The rudders shall be designed to provide a maximum deflection of $\pm 20^\circ$. The actuator stroke shall be designed to prevent the rudders from being deflected beyond these limits.

3.3.7.2 Slew Rate

The maximum slew rate for the rudders shall be approximately 20 degrees per second, based on the actuator interface specifications presented in Paragraph 3.5.2.

3.3.8 Balance

3.3.8.1 Blade

Each blade shall be balanced against a master balance blade. The blade shall be the complete and finished blade, to include any cuff, coating, and polyurethane leading edge strip. The blade shall balance horizontally and vertically at any two blade angles 90 degrees apart. It is permissible to use equipment of

equivalent accuracy in which the imbalance is determined with the blade centerline, located in either a horizontal or vertical position. Where balancing equipment other than knife edges is used, the unbalanced moment specified above may be applied as scale-reading deviations from the specified master moments. The above balance shall be accomplished on a finished composite blade, including cuff, coating, and polyurethane leading edge strip. The detachable rubber-covered leading-edge guards will be mass balanced.

3.3.8.2 Spinner

The means used to support the spinner for balancing shall provide the same degree of accuracy of centering and aligning the spinner as will be obtained when the spinner is installed on the propeller.

3.3.8.3 Dynamic

The entire propeller assembly shall be balanced to meet a balance quality grade of G6.3 or better.

3.4 Functional Requirements

3.4.1 De-Icing

The leading edge of the propeller shall be provided with a means for de-icing. (Note: The prototype craft will not be equipped with de-icing equipment, but the design shall accommodate its inclusion on production articles.)

3.4.2 Lift Points

The upper surface of the propeller shroud shall be equipped with lifting points so that the entire propulsor assembly can be suspended from these points.

The lift points and their mounts shall be designed with a yield factor of safety of 3.0.

3.4.3 Access Panels

Access panels shall be provided, as needed, in the duct surface to support replacement of stator blades and propeller blades.

3.5 Interfaces

3.5.1 Propeller Centerbody Interfaces

The stators shall interface with the propeller centerbody (hub assembly) as defined in Appendix G.

For information purposes only, the propeller centerbody is cylindrical in shape in the plane of the propeller, with a radius of 338.074 mm (13.31 inch), transitioning to an ellipsoidal shape 254 mm (10 inches) aft of the centerline of the propeller. The ellipsoid has minor axes of 338.074 mm (13.31 inches) radially and 1354 mm (53.307 inches) axially.

Details of this interface will be mutually agreed to between the supplier and the USN.

3.5.1.1 Hydraulics

The propeller shroud assembly shall provide hydraulic interfaces to support propeller blade angle control that is shown to be consistent with the appropriate HLCAC Propulsor Interface Control Document..

The hydraulic lines will be routed from the deck surface, through the propeller shroud foundations, propeller shroud, and a single stator to the centerbody. The hydraulic lines shall be routed internally within the propeller shroud assembly structure.

Details of the hydraulic fittings are provided in the HLCAC Propulsor Interface Control Document.

3.5.1.2 Load Sharing

The interface between the propeller centerbody and the stators shall support the load definitions provided elsewhere in this specification.

3.5.1.3 Electrical

The propeller shroud assembly shall provide sufficient internal wiring space to accommodate the wiring necessary for command and control of the variable pitch propeller mechanism. This wiring shall be located adjacent to the hydraulic lines to the extent possible.

3.5.2 Rudder Actuator Interface

Deflection of the rudders shall be via hydraulic control. Each rudder linkage shall interface with a hydraulic actuator whose performance is defined in the following paragraphs.

3.5.2.1 Actuator Stroke

Each rudder, linkage, and rudder attachment to the propeller shroud shall be designed to accommodate the maximum deflection angle that is specified in the appropriate HLCAC Propulsor Interface Control Document.

3.5.2.2 Actuator Fitting Interface Description

Refer to the HLCAC Propulsor Interface Control Document for details on the rudder and actuator interface to the propulsor shroud.

3.5.3 Deck Interfaces

The propeller shroud assembly shall be mounted on foundations such that the longitudinal centerline of the propeller is located 2190.75 mm above deck level.

The foundations shall be bolted to the deck.

3.5.4 Command and Control Interfaces

The propeller shroud assembly shall provide any necessary signal and electrical interfaces necessary for control of the rudders and the variable pitch propeller.

4.0 CONSTRUCTION

4.1 Nameplates and Product Marking

Equipment, assemblies, and parts shall be uniformly and legibly marked for identification. Label plates fitted in the weather shall be corrosion-resisting steel, QQ-S-766, Class 302, Finish 26. Information on the identification plate shall be as follows:

a) Propeller. Propellers shall be marked with the following information:

1. Model Designation
2. Serial Number
3. Manufacturer Identification

- b) Hub. The following information shall be stamped or etched in a location such as to be readily visible when installed in the HLCAC propeller shroud, but not in a location to produce critical stress concentrations:
 - 1. Model Designation (Propeller)
 - 2. Serial Number
 - 3. Part Number
- c) Blades
 - 1. Each blade shall have the following information stamped or etched on the base or ferrule, as applicable, in such a manner as not to affect the balance or strength of the blade:
 - i. Model Designation
 - ii. Serial Number
 - iii. Part Number
 - 2. The following information shall be enameled, labeled, or painted on the blade, parallel to the longitudinal axis of the blade, and on the cambered face. This marking shall be given one coat of clear varnish or lacquer.
 - i. Model Designation
 - ii. Serial Number
 - iii. Part Number
 - iv. Manufacturer Identification
- d) Spinner. The following markings shall be placed on the inside of the spinner shell or bulkhead at an appropriate and accessible location to ensure that this information will not be obliterated if outside surfaces of the spinner are painted. Locate markings in minimum erosion areas.
 - 1. Part Number
 - 2. Serial Number
 - 3. Manufacturer Identification
- e) Control. Controls shall be marked with the following information:
 - 1. Model Designation
 - 2. Serial Number
 - 3. Manufacturer Identification.
- f) Drive Shaft Adapter. The drive shaft adapter shall be marked with a part number and serial number, but not in a critical stress area.

4.1.1 Workmanship

Workmanship shall be in accordance with high industry standards.

4.1.2 Interchangeability

All parts having the same manufacturer part number shall be directly and completely interchangeable with each other with respect to installation and performance.

4.1.3 Safety

The propulsor system shall be designed to provide safety for operations and personnel.

4.2 Aerodynamic Surfaces and Contours

The aerodynamic surfaces of the propeller shroud assembly shall satisfy the following requirements:

4.2.1 Deviation From Flowlines

Deviation from the contours defined in Paragraph 3.1 shall not exceed 1.5 mm on the stator surfaces and the shroud internal surface, and shall not exceed 3.0 mm on the shroud external surface.

4.2.2 Discontinuities

Table 4-1 defines the allowable surface discontinuities for permanent joints on the propeller shroud assembly. The maximum allowable surface discontinuities for quick removal panels and hinged doors are less stringent, but shall be no greater than 1.5 times the values listed in Table 4-1.

Table 4-1

Surface Discontinuity Limits

Type	Allowable Discontinuity
Internal Shroud Surface and Stator Surfaces	
Forward Facing Step	0.8 mm
Aft Facing Step	1.2 mm
Longitudinal Step	1.6 mm
Transverse Gap	1.6 mm
Longitudinal Gap	1.6 mm
External Shroud Surface	
Forward Facing Step	1.6 mm
Aft Facing Step	1.6 mm
Longitudinal Step	1.6 mm
Transverse Gap	1.6 mm
Longitudinal Gap	2.4 mm

4.3 Water Intrusion

The propeller shroud assembly shall be designed to prohibit water intrusion. Sources of water intrusion include waves, spray, and rain, and include water shed from the propeller. Skin surfaces as well as joints shall be designed to prohibit water intrusion and absorption.

All internal and external surfaces shall be designed to prevent absorption of water.

4.3.1 Drain Plugs

If water intrusion cannot be eliminated, the propeller shroud assembly shall be fitted with drainage paths and drainage plugs which will allow the captured water to be drained.

4.4 Inner Duct Skin Joint Locations

To minimize fatigue risk, no inner duct skin joints are permitted in the plane of the propeller.

5.0 ENVIRONMENTAL REQUIREMENTS

5.1 Temperature

The propeller shroud assembly shall satisfy the requirements of Paragraph 3.0 in ambient temperatures ranging from -23°C to 49°C . Also, the propeller shall not suffer any adverse impact from temperature extremes, which include a -34°C cold soak and a 54°C heat exposure.

5.2 Humidity

The propeller shroud assembly shall satisfy the requirements of Paragraph 3.0 in relative humidity ranging from 10% to 100%.

5.3 Shock

The propeller shroud assembly shall satisfy the requirements of Paragraph 3.0 while experiencing shock loads.

5.4 Vibration

Propeller assembly must be balanced to meet or exceed the requirements of section 3.3.8.3.

5.5 Erosion – Wave Spray

The leading edges of the shroud, stators and rudders shall be designed to withstand erosion effects associated with operation in the marine environment (spray, blown sand, ice particles, etc.).

The leading edge of the rudders and stators shall be provided with replaceable erosion tape, or the equivalent.

6.0 FAILSAFE PROVISIONS

Failsafe provisions shall include physical stops that limit the maximum positive and maximum negative blade angles in the event of control system failure. In the event of a loss of electrical power to the locking device, the blade-angle servo actuator shall incorporate features to lock the propeller blade-angle input at the last actual position. A hydraulic lock shall be incorporated in the propeller to prevent reduction in pitch due to electrical or hydraulic failure when operating in forward pitch range.

7.0 USEFUL LIFE

The propeller shroud assembly shall be designed for a 20-year operating life with 200 operating hours per year. Normal preventive maintenance cycles shall be assumed. The blades should provide a minimum of 500 hours of operation before servicing, except for replacement of leading edge protection which shall have a minimum of 250 hours of operation.

8.0 DOCUMENTATION

8.1 Engineering Reports and Data

8.1.1 Engineering Analysis Report

An engineering analysis report shall be prepared summarizing the engineering analysis leading to the design of the propeller.

8.1.2 Stress Analysis

The report shall contain a summary of static load criteria, static stresses, and margins of safety for all major components. It shall include, as a minimum, the results of fatigue analyses of the shroud, propeller blade and hub, including design criteria, stresses and stress cycles, and material fatigue allowables. The report shall also indicate assumptions made on stress amplification due to blade dynamic response.

8.1.3 Acoustic Analysis

The report shall present an explanation of the analysis and methods used to predict the acoustic characteristics of the propeller.

8.1.4 Mechanical Operation and Control Analysis

The report shall contain a description of the pitch change mechanism and the method of control. It shall present a description of the analysis performed to show that the propeller mechanism meets the requirements of this specification.

8.1.5 Erosion Protection System Analysis

The report shall describe the analysis leading up to the selection of the erosion protection system, including the impact of the erosion protection system on performance and structural aspects of the design.

8.1.6 Propulsor Contractor Drawings

The propulsor manufacturer shall provide the following drawings and diagrams:

- Propeller assembly drawing.
- Blade and hub detail drawing.
- Propeller system installation drawing, including clearances for maintenance, checking adjustment, and removal of accessories and components subject to separate removal.
- Propeller accessory and component interface requirements (for units not mounted on the propeller).
- Electrical interface diagram showing the circuits internal and external to the propeller required for installation.
- Hydraulic installation interconnecting diagram showing interconnection between components.
- Electrical schematic of the propeller system.
- Craft interface drawing.

Drawings shall be prepared and maintained in accordance with MIL-T-31000 standards for mechanical engineering or equivalent.

8.1.7 Parts List

The parts list for the first propulsor system delivered shall constitute the approved parts list for any subsequent system of the same model to be delivered. Changes to the approved system parts list shall be approved by the USN.

8.1.8 Weight Status Reports

Weight status reports shall be submitted as weight changes occur, and attention shall be directed to any weight variations. The actual metric weight of each propeller system shall be determined and furnished prior to delivery.

8.1.9 Weight of Residual Fluids

The estimated metric weight of residual fluids remaining in a propeller system after operation and drainage shall be specified in the engineering analysis report.

8.1.10 Quality Assurance (Q/A) Reports

8.1.10.1 Dimensional Survey Reports

For the First Article and subsequent propulsors, a dimensional survey report, in metric units, shall be completed showing nominal (design drawing) dimensions and actual measured dimensions. Out-of-tolerance dimensions shall be highlighted. Out-of-tolerance items shall be resolved by means of correction or by waiver, depending on the sensitivity of the dimensions in question. Waivers will only be granted with the consent of the USN.

8.1.10.2 Material Reports

For each batch of material used, a report of the physical properties shall be submitted to the USN. Material documentation shall be traceable to point of manufacture. Where practicable, these properties should be confirmed by physical test of test specimens. Relevant dates shall be included.

8.1.10.3 Balancing Report

8.1.10.3.1 Blade Mass Properties

For the first article only, the weight and location of the center of gravity of each of the six blades shall be determined and recorded by serial number.

8.1.10.3.2 Propulsor Mass Properties

For the first article only, the mass properties, in metric units, including moment of inertia about the propeller centerline of the propeller assembly, shall be determined and recorded.

The report shall also contain:

- The weight of the major components of the shroud.
- The weight and CG location of the complete shroud assembly.
- The weight and CG location of the complete propulsor assembly.

8.1.10.3.3 Static Balance

During assembly, the manufacturer shall use reasonable diligence to ensure that each propeller is fairly well balanced statically by the use of selective blade assembly (by weight).

Static balance shall be determined and recorded prior to dynamic balancing.

8.1.10.3.4 Dynamic and Static Balance Report

Following completion of balancing in compliance with paragraph 3.3.8, a balancing report shall be submitted. This report shall include the blade mass properties report of paragraphs 8.1.8 and 8.1.9 and the moment of inertia of the propeller assembly as it exists prior to delivery, expressed in kg m² about the axis of rotation.

8.1.11 Delivery Reports

When the propulsor is ready for delivery, it shall be accompanied by a Delivery Report. This package shall contain copies of the dimensional survey report(s), balancing reports, and the weight report and dates of manufacture of components and date of assembly.

8.1.12 Technical Manuals

Source information shall be prepared to enable the USN to prepare technical manuals covering technical description, handling, installation and maintenance.

8.1.13 Test Plans and Procedures

Test plans and procedures shall be prepared in contractor format for development tests and qualification tests. An Acceptance Test Procedure shall be prepared and submitted to the USN for approval.

8.1.14 Configuration Control Requirements

The contractor shall maintain a configuration control system in accordance with DOD-STD-480 and MIL-STD-483 or equivalent.

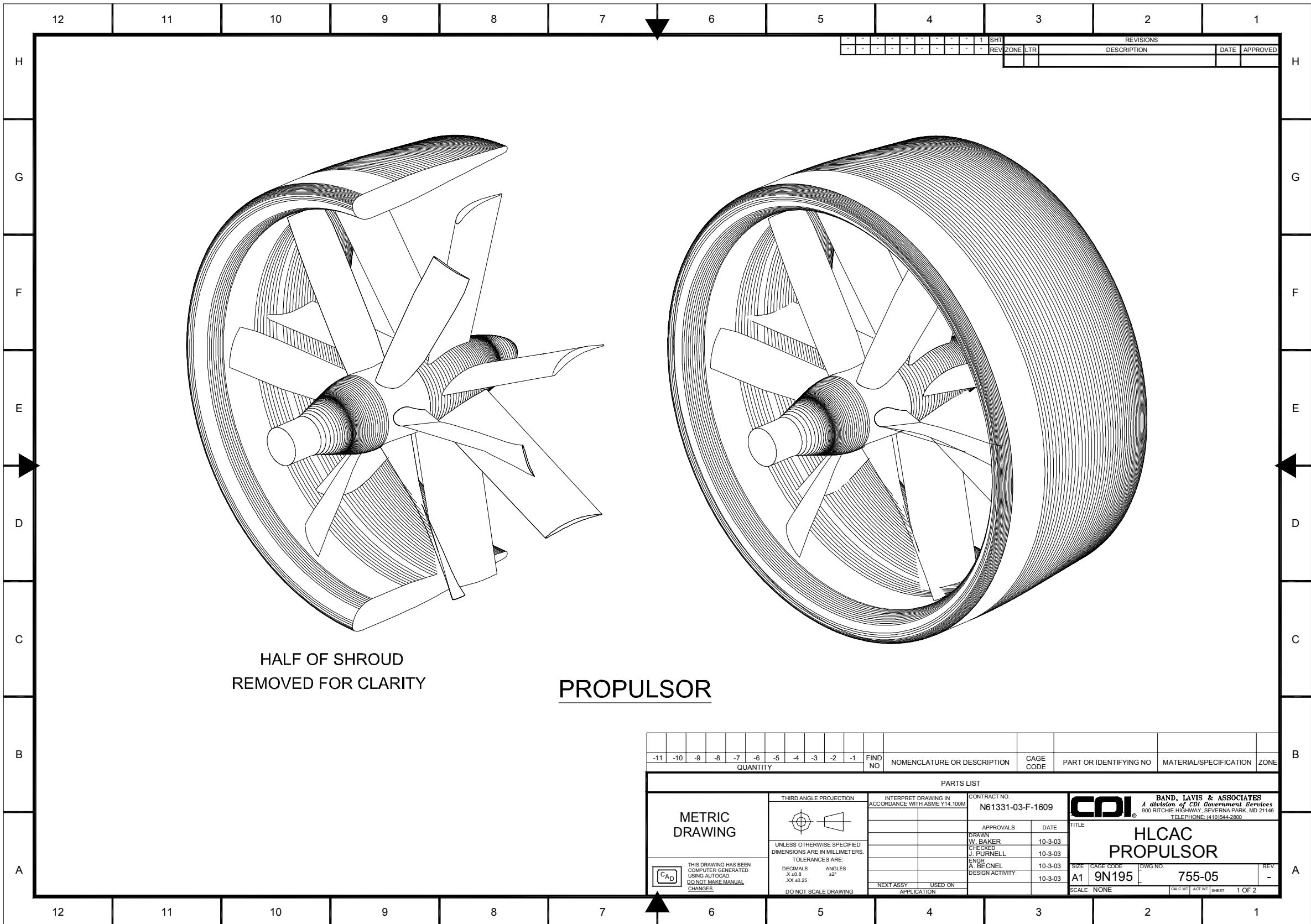
8.2 Logistics

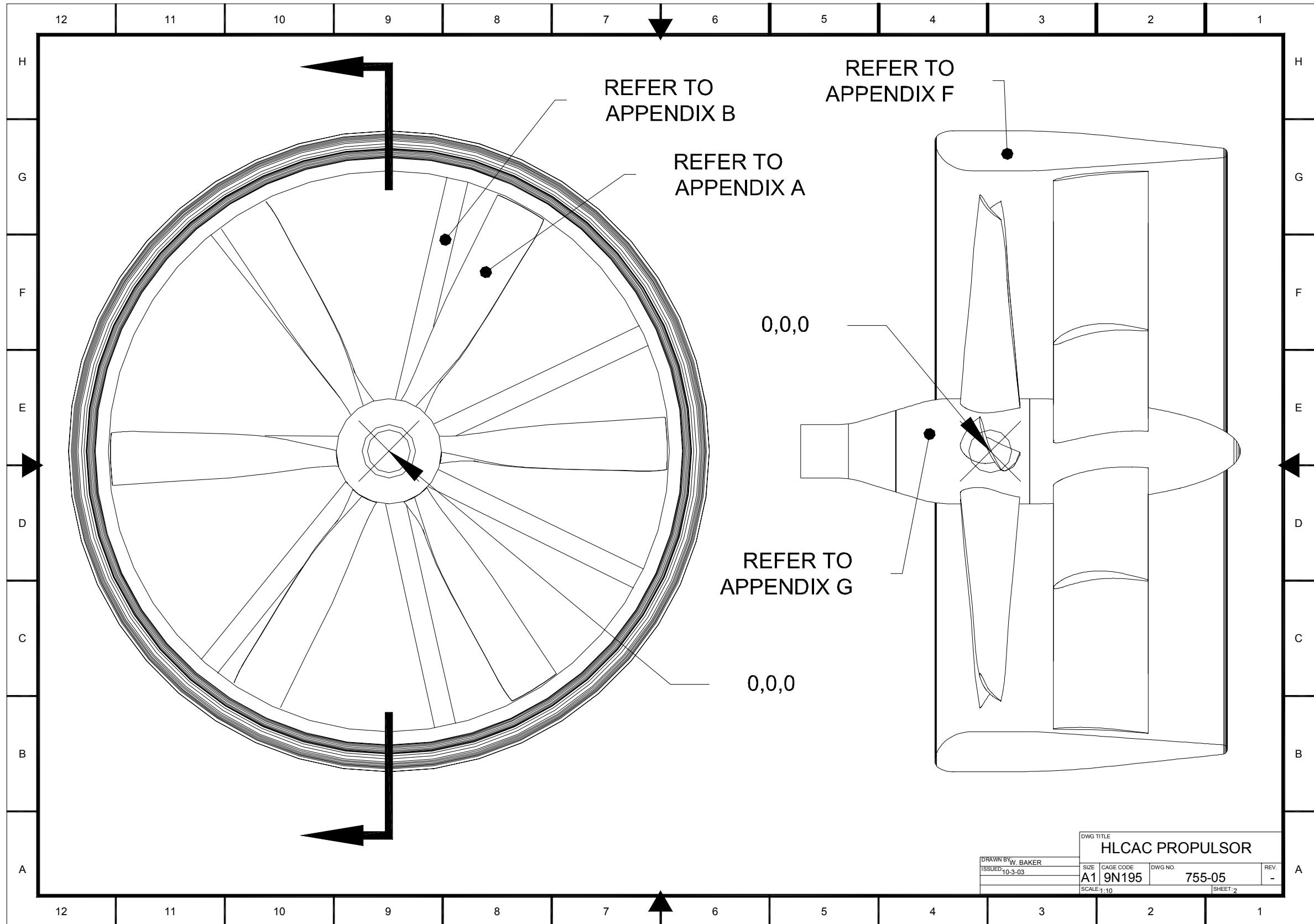
8.2.1 Maintenance

The propulsor shall be designed to be maintainable and provide for easy regular inspections, periodic replacement of erosion protection film, and replacement of damaged blades.

8.2.2 Supply

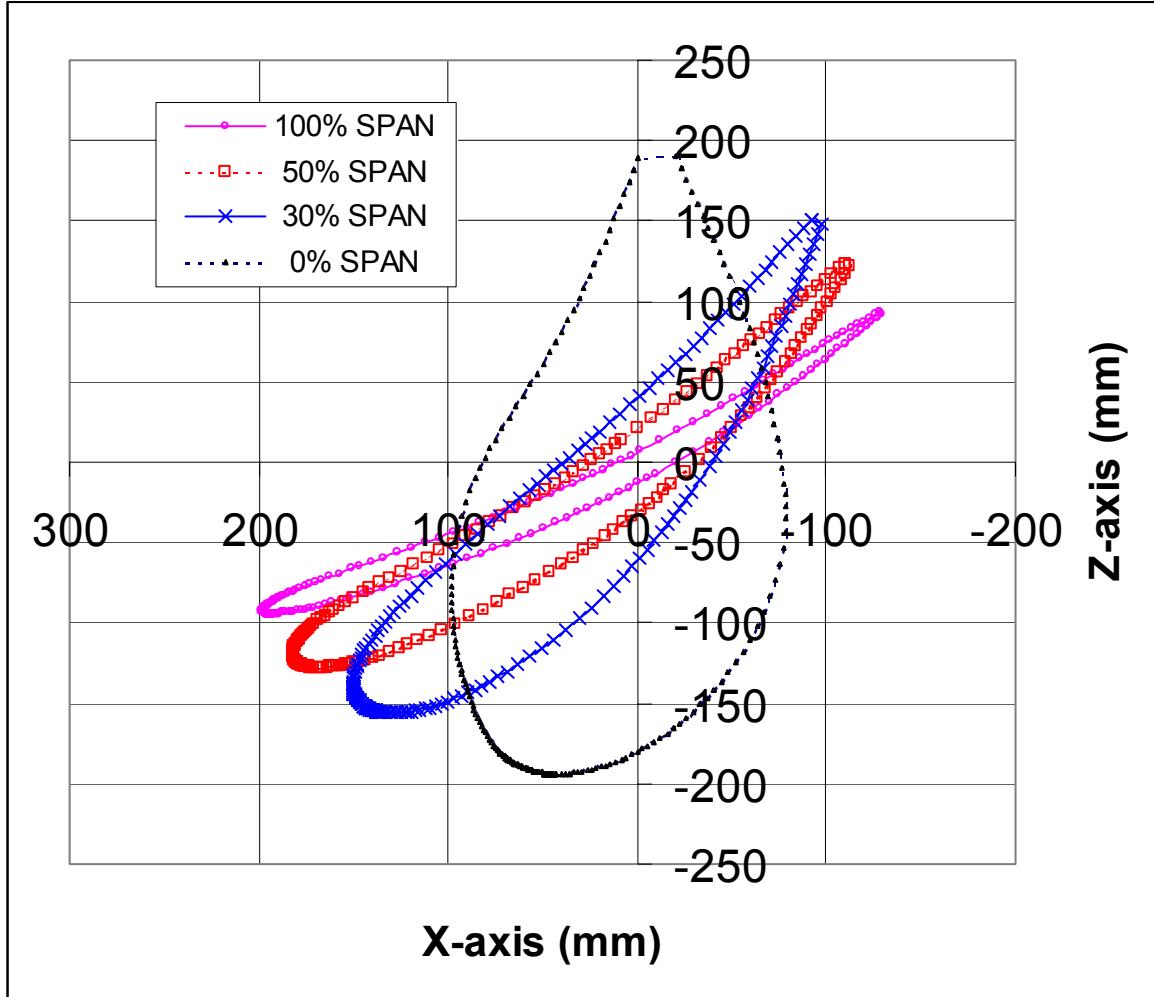
The contractor shall propose methods to ensure economic availability of support resources concurrent with delivery of the equipment for test and evaluation, if required, for operational use.





APPENDIX A
PROPELLER OFFSETS

The unwrapped propeller sections at several spans are shown in the Figure below. The X, Y, Z coordinates around the propeller blade from 0 to 100-percent span at 10-percent span increments is shown in the tables below. The data follows the right hand rule with the positive Z-axis being along the shaft. The propeller rotation as presented is for a counter-clockwise rotation when looking into the propeller inlet. All dimensions are presented in millimeters. Six identical propeller blades are used with a space of 60 degrees between each. The coordinate (0, 0, 0) represents the centerline location of the propeller.



0% SPAN			10% SPAN			20% SPAN			30% SPAN		
X	Y	Z	X	Y	Z	X	Y	Z	X	Y	Z
-0.5126	338.0736	189.4600	-33.2054	482.1947	176.8987	-65.8900	625.1363	164.3461	-91.6011	768.4213	150.9140
1.9957	338.0681	181.9460	-30.1212	482.3972	170.1615	-62.2331	625.5110	158.3855	-87.5730	768.8909	145.7383
4.5037	338.0440	174.4310	-27.0316	482.5802	163.4154	-58.5658	625.8649	152.4083	-83.5245	769.3411	140.5316
7.0112	338.0013	166.9250	-23.9362	482.7436	156.6658	-54.8873	626.1984	146.4152	-79.4575	769.7719	135.2980
9.5230	337.9399	159.4360	-20.8365	482.8873	149.9149	-51.2047	626.5103	140.4022	-75.3840	770.1814	130.0310
12.0473	337.8593	151.9920	-17.7351	483.0111	143.1889	-47.5325	626.7996	134.3942	-71.3255	770.5679	124.7509
14.5985	337.7587	144.6120	-14.6206	483.1154	136.5081	-43.8611	627.0671	128.4126	-67.2752	770.9320	119.4791
17.1875	337.6368	137.2960	-11.4787	483.2003	129.8706	-40.1734	627.3143	122.4534	-63.2155	771.2756	114.2122
19.8220	337.4924	130.0450	-8.3015	483.2653	123.2774	-36.4617	627.5410	116.5179	-59.1380	771.5988	108.9502
22.5087	337.3239	122.8630	-5.0829	483.3100	116.7332	-32.7201	627.7471	110.6113	-55.0371	771.9022	103.6982
25.2528	337.1295	115.7550	-1.8167	483.3332	110.2419	-28.9420	627.9326	104.7365	-50.9059	772.1856	98.4583
28.0584	336.9076	108.7190	1.5024	483.3343	103.8027	-25.1207	628.0970	98.8939	-46.7363	772.4493	93.2313
30.9284	336.6563	101.7540	4.8783	483.3120	97.4150	-21.2516	628.2399	93.0833	-42.5223	772.6927	88.0170
33.8656	336.3735	94.8569	8.3149	483.2651	91.0770	-17.3298	628.3603	87.3043	-38.2580	772.9154	82.8150
36.8706	336.0574	88.0256	11.8146	483.1923	84.7876	-13.3513	628.4575	81.5566	-33.9377	773.1174	77.6252
39.9451	335.7058	81.2525	15.3800	483.0919	78.5415	-9.3123	628.5303	75.8373	-29.5555	773.2972	72.4452
43.0886	335.3169	74.5306	19.0127	482.9626	72.3339	-5.2099	628.5777	70.1438	-25.1042	773.4545	67.2706
46.3011	334.8884	67.8486	22.7141	482.8026	66.1571	-1.0411	628.5984	64.4719	-20.5762	773.5883	62.0945
49.5820	334.4184	61.1939	26.4853	482.6105	60.0022	3.1964	628.5911	58.8169	-15.9641	773.6971	56.9108
52.9305	333.9048	54.5502	30.3270	482.3843	53.8583	7.5052	628.5544	53.1729	-11.2604	773.7799	51.7134
56.3451	333.3456	47.8981	34.2396	482.1224	47.7127	11.8868	628.4868	47.5340	-6.4553	773.8349	46.4928
59.8250	332.7386	41.2138	38.2306	481.8223	41.5375	16.3570	628.3863	41.8685	-1.5236	773.8603	41.2149
63.3690	332.0819	34.4655	42.3178	481.4805	35.2827	20.9524	628.2499	36.1077	3.5788	773.8535	35.8155
66.9770	331.3731	27.6103	46.5185	481.0928	28.8960	25.7084	628.0732	30.1899	8.8949	773.8107	30.2398
70.6490	330.6097	20.5842	50.8425	480.6552	22.3279	30.6443	627.8519	24.0802	14.4556	773.7268	24.4608
74.3856	329.7890	13.2875	55.2912	480.1637	15.5228	35.7618	627.5811	17.7673	20.2751	773.5963	18.4807
76.9414	329.2021	8.1364	58.2856	479.8094	10.8364	39.1638	627.3781	13.5455	23.9488	773.4911	14.7275
79.5261	328.5873	2.7090	61.3299	479.4298	5.9580	42.6354	627.1517	9.2161	27.7183	773.3652	10.9103
82.1338	327.9452	-3.1340	64.4196	479.0244	0.8030	46.1730	626.9011	4.7498	31.5836	773.2170	7.0319
84.7350	327.2828	-9.6927	67.5366	478.5949	-4.8168	49.7711	626.6258	0.0700	35.5482	773.0449	3.0882
87.2862	326.6116	-16.9973	70.6568	478.1443	-10.9253	53.4257	626.3248	-4.8407	39.6186	772.8470	-0.9289
89.6949	325.9584	-25.0103	73.7252	477.6807	-17.5052	57.1219	625.9984	-9.9853	43.8028	772.6211	-5.0299
91.7762	325.3785	-33.5334	76.6283	477.2237	-24.4406	60.8237	625.6496	-15.3304	48.1114	772.3648	-9.2285
93.3982	324.9166	-42.0171	79.2891	476.7888	-31.3998	64.5129	625.2800	-20.7626	52.5606	772.0747	-13.5448
94.6182	324.5634	-50.0889	81.7557	476.3719	-38.1679	68.2280	624.8855	-26.2255	57.1778	771.7466	-18.0135
95.5126	324.3014	-57.6337	84.0963	475.9644	-44.7030	72.0280	624.4590	-31.7500	62.0065	771.3737	-22.6944
96.1210	324.1216	-64.6177	86.3691	475.5573	-51.0359	75.9891	623.9893	-37.4322	67.1215	770.9454	-27.6953
96.4673	324.0187	-71.0328	88.6158	475.1437	-57.2120	80.1699	623.4659	-43.3708	72.5955	770.4493	-33.1349
96.6063	323.9772	-76.9538	90.8589	474.7199	-63.2906	84.5589	622.8859	-49.6102	78.4249	769.8777	-39.0646
96.5925	323.9814	-82.4766	93.0392	474.2973	-69.2385	88.9787	622.2699	-55.9866	84.4039	769.2452	-45.2965
96.4657	324.0191	-87.6824	95.0865	473.8911	-74.9720	93.2456	621.6448	-62.2513	90.2840	768.5771	-51.5194
96.2545	324.0819	-92.6339	96.9762	473.5081	-80.4593	97.2797	621.0262	-68.2774	95.9601	767.8892	-57.5663
95.9786	324.1637	-97.3757	98.6844	473.1551	-85.6842	101.0123	620.4302	-73.9877	101.3684	767.1940	-63.3512
95.6419	324.2632	-101.9180	100.1805	472.8406	-90.6184	104.3772	619.8729	-79.3151	106.4466	766.5059	-68.8094
95.2478	324.3792	-106.2710	101.4384	472.5722	-95.2396	107.3181	619.3705	-84.2048	111.1294	765.8409	-73.8810
94.8006	324.5102	-110.4530	102.4592	472.3519	-99.5572	109.8334	618.9294	-88.6574	115.3783	765.2124	-78.5401
94.3132	324.6522	-114.5000	103.2858	472.1719	-103.6254	111.9971	618.5415	-92.7456	119.1743	764.6304	-82.7920
93.7873	324.8045	-118.4270	103.9506	472.0261	-107.4840	113.8743	618.1988	-96.5347	122.4693	764.1096	-86.6232
93.2211	324.9675	-122.2350	104.4755	471.9102	-111.1559	115.5110	617.8950	-100.0707	125.2576	763.6574	-90.0422
92.6107	325.1420	-125.9200	104.8753	471.8214	-114.6565	116.9412	617.6260	-103.3874	127.6406	763.2628	-93.1289
91.9351	325.3336	-129.4310	105.1470	471.7609	-117.9631	118.1792	617.3903	-106.4899	129.7398	762.9087	-95.9704
91.2782	325.5185	-132.4310	105.4620	471.6907	-121.0565	119.4838	617.1390	-109.6773	132.0567	762.5111	-99.1921
90.5794	325.7137	-135.3140	105.6724	471.6436	-123.9975	120.6195	616.9181	-112.6767	134.1240	762.1501	-102.1898
89.8446	325.9171	-138.0930	105.7995	471.6151	-126.8107	121.6237	616.7209	-115.5245	135.9972	761.8182	-105.0099
89.0794	326.1271	-140.7820	105.8680	471.5997	-129.5242	122.5401	616.5396	-118.2628	137.7315	761.5065	-107.7002
88.3057	326.3535	-143.4300	105.9125	471.5898	-132.1891	123.4164	616.3647	-120.9451	139.3735	761.2078	-110.3062
87.5413	326.5433	-146.0820	105.9266	471.5865	-134.8207	124.2216	616.2028	-123.5569	140.8914	760.9282	-112.8064
86.7781	326.7470	-148.7270	105.8832	471.5963	-137.3945	124.9089	616.0640	-126.0597	142.2444	760.6764	-115.1708
86.0099	326.9500	-151.3570	105.7689	471.6219	-139.8994	125.4583	615.9523	-128.4397	143.4139	760.4568	-117.3875
85.2299	327.1542	-153.9640	105.5853	471.6630	-142.3361	125.8800	615.8662	-130.7063	144.4104	760.2682	-119.4672
84.4302	327.3615	-156.5350	105.3437	471.7171	-144.7105	126.2044	615.7998	-132.8840	145.2680	760.1047	-121.4385
83.6035	327.5736	-159.0610	105.0508	471.7823	-147.0278	126.4531	615.7488	-134.9926	146.0125	759.9622	-123.3241
82.7421	327.7923	-161.5280	104.7079	471.8587	-149.2837	126.6355	615.7114	-137.0378	146.6610	759.8373	-125.1359
81.8320	328.0206	-163.9090	104.3112	471.9465	-151.4662	126.7592	615.6859	-139.0219	147.2310	759.7271	-126.8874
80.8717	328.2587	-166.1980	103.8635	472.0453	-153.5744	126.8309	615.6710	-140.9495	147.7321	759.6298	-128.5863
79.8622	328.5058	-168.3960	103.3648	472.1546	-155.6081	126.8493	615.6673	-142.8193	148.1667	759.5450	-130.2333

33.6591	336.3943	-193.1200	66.5836	478.7285	-181.0865	99.4764	620.6782	-169.0599	129.0916	763.0187	-155.9756
31.3479	336.6175	-192.7310	64.5062	479.0128	-180.8203	97.6277	620.9717	-168.9166	127.4798	763.2896	-155.9358
28.9993	336.8280	-192.2640	62.3839	479.2939	-180.4862	95.7273	621.2674	-168.7157	125.8026	763.5679	-155.8432
26.6140	337.0248	-191.7220	60.2190	479.5706	-180.0885	93.7786	621.5646	-168.4624	124.0609	763.8528	-155.6996
24.1933	337.2072	-191.1080	58.0140	479.8423	-179.6300	91.7853	621.8621	-168.1597	122.2553	764.1438	-155.5061
21.7351	337.3746	-190.4200	55.7632	480.1091	-179.1062	89.7382	622.1608	-167.8006	120.3744	764.4424	-155.2535
19.2280	337.5268	-189.6320	53.4551	480.3716	-178.4967	87.6259	622.4617	-167.3701	118.4036	764.7501	-154.9301
16.6687	337.6628	-188.7340	51.0756	480.6304	-177.7855	85.4238	622.7677	-166.8462	116.3139	765.0707	-154.5133
14.0553	337.7817	-187.7190	48.6096	480.8860	-176.9539	83.1035	623.0817	-166.1986	114.0712	765.4083	-153.9704
11.3865	337.8822	-186.5770	46.0362	481.1393	-175.9784	80.6246	623.4073	-165.3899	111.6326	765.7678	-153.2657
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5.8649	338.0231	-183.8340	40.5284	481.6345	-173.5137	75.1319	624.0931	-163.2041	106.1107	766.5524	-151.3010
3.0018	338.0607	-182.1850	37.6063	481.8714	-172.0071	72.1533	624.4445	-161.8400	103.0564	766.9690	-150.0571
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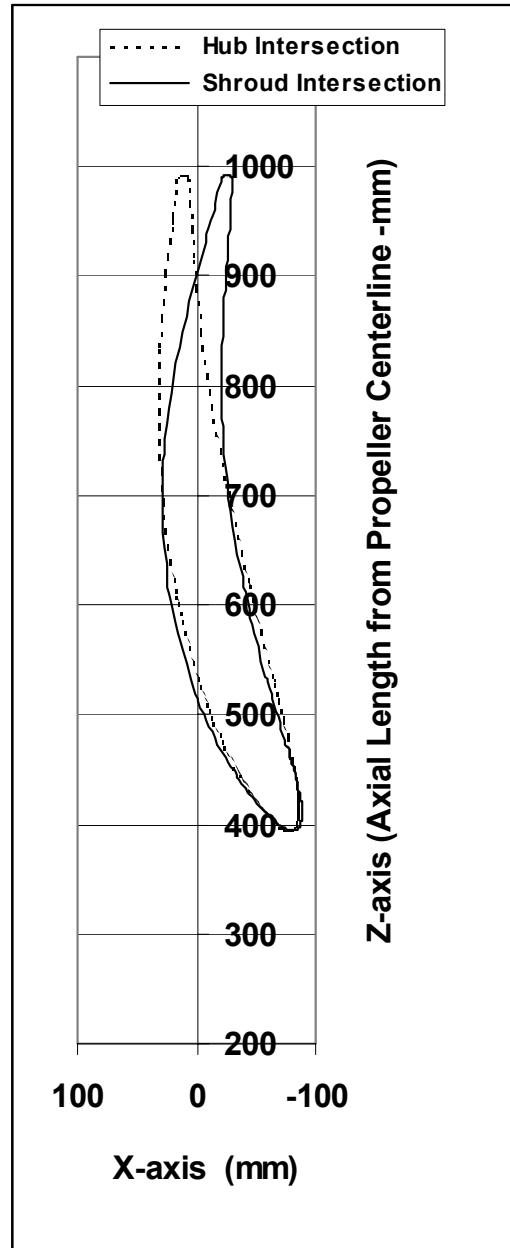
80% SPAN			90% SPAN			100% SPAN		
X	Y	Z	X	Y	Z	X	Y	Z
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-54.6230	1499.1801	49.4383	-58.4056	1644.4005	48.6106	-62.2049	1789.6193	47.5322
-49.3326	1499.3634	45.3702	-53.0625	1644.5817	44.6801	-56.8087	1789.7987	43.7756
-43.6545	1499.5396	41.0483	-47.2148	1644.7599	40.4031	-50.9709	1789.9744	39.7331
-37.5499	1499.7047	36.4641	-40.8287	1644.9308	35.7808	-44.5382	1790.1460	35.3079
-30.9748	1499.8550	31.6107	-33.8908	1645.0883	30.8357	-37.4199	1790.3091	30.4619
-23.8797	1499.9847	26.4808	-26.3797	1645.2260	25.5887	-29.5638	1790.4559	25.1940
-16.2074	1500.0873	21.0678	-18.2645	1645.3361	20.0565	-20.9197	1790.5779	19.5132
-7.8963	1500.1541	15.3674	-9.5021	1645.4100	14.2502	-11.4457	1790.6635	13.4407
1.1195	1500.1744	9.3789	-0.0372	1645.4375	8.1764	-1.1116	1790.6997	7.0123
5.2661	1500.1655	6.6921	4.3059	1645.4318	5.4583	3.4606	1790.6967	4.2348
9.5759	1500.1443	3.9429	8.8017	1645.4138	2.6893	8.1872	1790.6813	1.4066
14.0650	1500.1089	1.1250	13.4569	1645.3823	-0.1305	13.0680	1790.6523	-1.4674
18.7451	1500.0577	-1.7644	18.2756	1645.3361	-2.9988	18.0996	1790.6084	-4.3808
23.6226	1499.9888	-4.7240	23.2607	1645.2731	-5.9126	23.2779	1790.5487	-7.3263
28.6989	1499.9003	-7.7484	28.4150	1645.1921	-8.8687	28.5984	1790.4717	-10.2967
33.9716	1499.7902	-10.8295	33.7414	1645.0914	-11.8636	34.0562	1790.3761	-13.2851
39.4364	1499.6565	-13.9584	39.2425	1644.9694	-14.8941	39.6483	1790.2610	-16.2855
45.0892	1499.4971	-17.1259	44.9218	1644.8241	-17.9570	45.3781	1790.1250	-19.2965
50.9295	1499.3099	-20.3258	50.7824	1644.6537	-21.0491	51.2518	1789.9664	-22.3187
56.9600	1499.0930	-23.5543	56.8281	1644.4557	-24.1679	57.2758	1789.7838	-25.3530
63.1840	1498.8436	-26.8087	63.0626	1644.2285	-27.3107	63.4567	1789.5753	-28.4000
69.6025	1498.5593	-30.0849	69.4895	1643.9695	-30.4759	69.8004	1789.3391	-31.4607
76.2174	1498.2374	-33.3794	76.1123	1643.6760	-33.6619	76.3123	1789.0732	-34.5360
83.0303	1497.8754	-36.6900	82.9319	1643.3462	-36.8669	82.9963	1788.7756	-37.6268
90.0430	1497.4702	-40.0153	89.9479	1642.9771	-40.0886	89.8549	1788.4442	-40.7333
97.2546	1497.0190	-43.3532	97.1584	1642.5665	-43.3245	96.8868	1788.0770	-43.8550
104.6611	1496.5195	-46.7018	104.5575	1642.1121	-46.5719	104.0859	1787.6725	-46.9899
112.2500	1495.9694	-50.0571	112.1234	1641.6128	-49.8245	111.4284	1787.2297	-50.1315
119.9854	1495.3689	-53.4078	119.6953	1641.0781	-53.0326	118.8645	1786.7507	-53.2628
127.7930	1494.7219	-56.7294	127.0577	1640.5245	-56.1171	126.3109	1786.2396	-56.3548
135.4958	1494.0433	-59.9622	134.3182	1639.9460	-59.1117	133.6253	1785.7074	-59.3574
142.6273	1493.3794	-62.9452	141.1907	1639.3687	-61.9223	140.3901	1785.1884	-62.1250
148.5833	1492.7986	-65.4628	147.3232	1638.8289	-64.4315	146.1474	1784.7261	-64.4944
153.2853	1492.3230	-67.4918	152.4593	1638.3591	-66.5531	150.8245	1784.3370	-66.4425
159.6034	1491.6606	-70.2693	159.1944	1637.7184	-69.3779	157.1684	1783.7894	-69.1278
164.2580	1491.1552	-72.3955	164.3706	1637.2069	-71.5976	162.1589	1783.3428	-71.2796
168.3681	1490.6967	-74.2965	168.5520	1636.7817	-73.4325	166.2884	1782.9624	-73.0925
172.0504	1490.2762	-76.0252	172.0505	1636.4177	-75.0094	169.8026	1782.6311	-74.6674
175.3125	1489.8959	-77.5925	175.0546	1636.0990	-76.4047	172.8368	1782.3395	-76.0604
178.1446	1489.5599	-79.0001	177.7003	1635.8138	-77.6697	175.4782	1782.0813	-77.3074
180.6075	1489.2634	-80.2736	180.0823	1635.5533	-78.8411	177.7874	1781.8524	-78.4325
182.7893	1488.9972	-81.4453	182.2522	1635.3130	-79.9375	179.8173	1781.6488	-79.4553
184.7493	1488.7552	-82.5382	184.2332	1635.0911	-80.9684	181.6181	1781.4661	-80.3951
186.5318	1488.5330	-83.5679	186.0416	1634.8861	-81.9394	183.2335	1781.3007	-81.2678
188.1716	1488.3265	-84.5469	187.6918	1634.6976	-82.8550	184.7030	1781.1489	-82.0876
189.6963	1488.1331	-85.4850	189.1936	1634.5244	-83.7188	186.0596	1781.0077	-82.8661
191.1146	1487.9515	-86.3845	190.5632	1634.3652	-84.5355	187.3157	1780.8761	-83.6077
192.4268	1487.7823	-87.2462	191.8130	1634.2191	-85.3093	188.4769	1780.7534	-84.3149
193.6369	1487.6254	-88.0703	192.9536	1634.0848	-86.0444	189.5491	1780.6396	-84.9893

194.7462	1487.4806	-88.8577	193.9953	1633.9614	-86.7439	190.5371	1780.5342	-85.6329
195.7570	1487.3479	-89.6091	194.9452	1633.8484	-87.4107	191.4460	1780.4366	-86.2473
196.6727	1487.2271	-90.3257	195.8115	1633.7449	-88.0472	192.2801	1780.3469	-86.8338
197.4970	1487.1178	-91.0087	196.6000	1633.6501	-88.6557	193.0423	1780.2643	-87.3941
198.2353	1487.0197	-91.6598	197.3157	1633.5638	-89.2379	193.7370	1780.1888	-87.9291
198.5375	1486.9792	-91.9401	197.6104	1633.5282	-89.4880	194.0160	1780.1586	-88.1522
198.8241	1486.9409	-92.2147	197.8929	1633.4940	-89.7334	194.2828	1780.1294	-88.3710
199.0949	1486.9048	-92.4838	198.1633	1633.4613	-89.9743	194.5396	1780.1014	-88.5856
199.3518	1486.8704	-92.7475	198.4205	1633.4301	-90.2110	194.7842	1780.0746	-88.7959
199.5950	1486.8376	-93.0061	198.6667	1633.4001	-90.4433	195.0171	1780.0492	-89.0022
199.8254	1486.8068	-93.2601	198.9017	1633.3716	-90.6716	195.2403	1780.0247	-89.2044
200.0438	1486.7773	-93.5096	199.1257	1633.3442	-90.8959	195.4520	1780.0015	-89.4027
200.2512	1486.7494	-93.7551	199.3389	1633.3182	-91.1166	195.6535	1779.9792	-89.5972
200.4497	1486.7227	-93.9971	199.5427	1633.2933	-91.3336	195.8448	1779.9583	-89.7878
200.6391	1486.6971	-94.2363	199.7360	1633.2697	-91.5472	196.0264	1779.9382	-89.9748
200.8231	1486.6724	-94.4729	199.9212	1633.2471	-91.7574	196.1983	1779.9193	-90.1580
201.0018	1486.6482	-94.7080	200.0975	1633.2253	-91.9647	196.3604	1779.9014	-90.3376
201.1754	1486.6248	-94.9410	200.2658	1633.2048	-92.1689	196.5124	1779.8846	-90.5138
201.3398	1486.6024	-95.1706	200.4265	1633.1851	-92.3705	196.6556	1779.8689	-90.6866
201.4914	1486.5819	-95.3955	200.5778	1633.1665	-92.5687	196.7891	1779.8541	-90.8555
201.6281	1486.5634	-95.6151	200.7188	1633.1492	-92.7632	196.9127	1779.8405	-91.0212
201.7473	1486.5472	-95.8283	200.8468	1633.1334	-92.9531	197.0278	1779.8278	-91.1833
201.8495	1486.5333	-96.0347	200.9618	1633.1193	-93.1383	197.1327	1779.8162	-91.3419
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201.9981	1486.5131	-96.4281	201.1471	1633.0966	-93.4925	197.3134	1779.7960	-91.6482
202.0471	1486.5065	-96.6157	201.2156	1633.0881	-93.6610	197.3884	1779.7878	-91.7959
202.0812	1486.5020	-96.7973	201.2690	1633.0815	-93.8239	197.4539	1779.7805	-91.9397
202.1012	1486.4991	-96.9740	201.3067	1633.0769	-93.9817	197.5092	1779.7743	-92.0798
202.1092	1486.4979	-97.1461	201.3314	1633.0737	-94.1342	197.5545	1779.7693	-92.2163
202.1051	1486.4985	-97.3140	201.3417	1633.0725	-94.2821	197.5886	1779.7654	-92.3487
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202.0688	1486.5035	-97.6394	201.3303	1633.0740	-94.5658	197.6240	1779.7616	-92.6010
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201.5675	1486.5717	-98.8103	200.8915	1633.1279	-95.5604	197.2025	1779.8083	-93.4017
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199.8766	1486.7999	-99.8836	199.2243	1633.3320	-96.4021	195.6311	1779.9818	-94.0142
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199.3441	1486.8713	-100.0269	198.7310	1633.3922	-96.5131	195.2001	1780.0292	-94.0935
199.0625	1486.9091	-100.0932	198.4722	1633.4237	-96.5642	194.9705	1780.0542	-94.1277
198.7755	1486.9474	-100.1567	198.2046	1633.4563	-96.6121	194.7312	1780.0804	-94.1585
198.4842	1486.9865	-100.2191	197.9292	1633.4896	-96.6573	194.4816	1780.1077	-94.1856
198.1857	1487.0261	-100.2793	197.6448	1633.5242	-96.6989	194.2230	1780.1360	-94.2092
197.8800	1487.0669	-100.3355	197.3506	1633.5597	-96.7373	193.9545	1780.1653	-94.2290
197.5644	1487.1089	-100.3889	197.0464	1633.5963	-96.7718	193.6756	1780.1957	-94.2450
197.2378	1487.1522	-100.4379	196.7309	1633.6344	-96.8025	193.3868	1780.2271	-94.2570
196.8993	1487.1971	-100.4831	196.4055	1633.6736	-96.8293	193.0860	1780.2596	-94.2653
196.5470	1487.2437	-100.5235	196.0680	1633.7141	-96.8521	192.7751	1780.2933	-94.2693
196.1819	1487.2919	-100.5589	195.7194	1633.7559	-96.8706	192.4541	1780.3281	-94.2694
195.8027	1487.3419	-100.5884	195.3589	1633.7990	-96.8848	192.1208	1780.3641	-94.2652
195.4071	1487.3939	-100.6119	194.9859	1633.8436	-96.8944	191.7759	1780.4012	-94.2568
194.9966	1487.4478	-100.6306	194.5992	1633.8896	-96.8994	191.4197	1780.4396	-94.2440
194.0156	1487.5760	-100.6491	193.6789	1633.9990	-96.8926	190.5464	1780.5332	-94.1969
192.9467	1487.7151	-100.6360	192.6848	1634.1166	-96.8592	189.6049	1780.6337	-94.1244
191.7872	1487.8650	-100.5891	191.6128	1634.2427	-96.7978	188.5916	1780.7413	-94.0258

190.5326	1488.0262	-100.5061	190.4574	1634.3776	-96.7060	187.5040	1780.8562	-93.8991
189.1816	1488.1986	-100.3878	189.2113	1634.5225	-96.5813	186.3377	1780.9786	-93.7431
187.7298	1488.3824	-100.2336	187.8662	1634.6776	-96.4210	185.0873	1781.1090	-93.5562
186.1763	1488.5775	-100.0407	186.4137	1634.8438	-96.2218	183.7488	1781.2476	-93.3366
184.5198	1488.7837	-99.8111	184.8422	1635.0223	-95.9799	182.3164	1781.3948	-93.0822
182.7588	1489.0009	-99.5429	183.1395	1635.2139	-95.6908	180.7833	1781.5510	-92.7909
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174.2587	1490.0195	-97.9449	174.7456	1636.1321	-93.9742	173.1707	1782.3071	-91.0818
171.5498	1490.3339	-97.3327	172.1607	1636.4061	-93.3721	170.6828	1782.5470	-90.4390
168.4610	1490.6862	-96.5786	169.3055	1636.7041	-92.6720	167.8586	1782.8152	-89.6699
164.9239	1491.0817	-95.6539	166.0790	1637.0345	-91.8341	164.6335	1783.1158	-88.7486
160.9408	1491.5168	-94.5529	162.3383	1637.4098	-90.8022	160.9145	1783.4554	-87.6366
156.4992	1491.9894	-93.2609	157.8802	1637.8457	-89.4908	156.5590	1783.8430	-86.2717
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144.8168	1493.1686	-89.4996	145.3535	1639.0048	-85.4016	144.7454	1784.8405	-82.2702
139.8953	1493.6378	-87.7685	140.0316	1639.4680	-83.5229	139.9314	1785.2244	-80.5314
133.7146	1494.2037	-85.4876	133.7212	1639.9948	-81.2140	134.0392	1785.6764	-78.3331
126.3491	1494.8446	-82.6482	126.6623	1640.5552	-78.5495	127.1373	1786.1810	-75.6779
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102.4086	1496.6754	-72.7574	103.8717	1642.1555	-69.4269	104.5341	1787.6461	-66.5103
94.5953	1497.1895	-69.3324	96.1206	1642.6276	-66.1220	97.0615	1788.0676	-63.3297
86.9633	1497.6520	-65.8895	88.5389	1643.0536	-62.7975	89.7348	1788.4503	-60.1350
79.5264	1498.0656	-62.4372	81.1472	1643.4353	-59.4661	82.5754	1788.7950	-56.9375
72.2908	1498.4320	-58.9806	73.9526	1643.7748	-56.1334	75.5912	1789.1039	-53.7427
65.2574	1498.7548	-55.5236	66.9576	1644.0746	-52.8032	68.7832	1789.3785	-50.5529
58.4257	1499.0366	-52.0680	60.1631	1644.3372	-49.4786	62.1490	1789.6212	-47.3697
51.7952	1499.2805	-48.6169	53.5689	1644.5652	-46.1627	55.6851	1789.8340	-44.1933
45.3665	1499.4888	-45.1742	47.1735	1644.7611	-42.8583	49.3865	1790.0188	-41.0245
39.1402	1499.6642	-41.7441	40.9743	1644.9271	-39.5678	43.2480	1790.1777	-37.8629
33.1146	1499.8092	-38.3301	34.9687	1645.0657	-36.2946	37.2636	1790.3124	-34.7087
27.2887	1499.9265	-34.9368	29.1545	1645.1791	-33.0419	31.4273	1790.4243	-31.5619
21.6646	1500.0184	-31.5700	23.5288	1645.2692	-29.8130	25.7330	1790.5151	-28.4224
16.2488	1500.0868	-28.2386	18.0893	1645.3380	-26.6113	20.1779	1790.5864	-25.2918
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6.0622	1500.1626	-21.7168	7.7582	1645.4191	-20.3053	9.5050	1790.6748	-19.0772
1.2940	1500.1742	-18.5413	2.8618	1645.4349	-17.2080	4.3988	1790.6947	-16.0058
-3.2648	1500.1713	-15.4277	-1.8587	1645.4363	-14.1526	-0.5468	1790.7000	-12.9673
-7.6269	1500.1555	-12.3752	-6.4070	1645.4249	-11.1419	-5.3279	1790.6920	-9.9677
-11.8100	1500.1283	-9.3790	-10.7900	1645.4021	-8.1767	-9.9434	1790.6724	-7.0124
-20.8556	1500.0299	-2.6616	-20.2998	1645.3123	-1.5188	-20.3362	1790.5845	-0.1349
-29.1225	1499.8921	3.7815	-29.0439	1645.1812	4.8877	-29.8040	1790.4519	6.4144
-36.6782	1499.7264	9.9438	-37.0799	1645.0195	11.0309	-38.3781	1790.2887	12.5935
-43.5878	1499.5415	15.8236	-44.4520	1644.8368	16.8956	-46.1044	1790.1064	18.3754
-49.9134	1499.3442	21.4236	-51.1940	1644.6407	22.4620	-53.0377	1789.9144	23.7465
-55.7101	1499.1401	26.7484	-57.3287	1644.4385	27.7077	-59.2373	1789.7200	28.7047
-61.0270	1498.9331	31.8037	-62.8739	1644.2357	32.6094	-64.8009	1789.5271	33.2844
-65.9091	1498.7263	36.5972	-67.8681	1644.0371	37.1645	-69.8919	1789.3356	37.5783
-70.4130	1498.5215	41.1516	-72.4104	1643.8434	41.4245	-74.6454	1789.1436	41.6691
-74.6135	1498.3181	45.5106	-76.6053	1643.6532	45.4545	-79.1608	1788.9495	45.6195
-78.5682	1498.1160	49.7084	-80.5469	1643.4648	49.3168	-83.5046	1788.7520	49.4710
-82.3178	1497.9147	53.7677	-84.3116	1643.2759	53.0651	-87.6924	1788.5515	53.2284
-85.8899	1497.7141	57.7029	-87.9357	1643.0859	56.7219	-91.7263	1788.3491	56.8867
-89.2994	1497.5146	61.5178	-91.4375	1642.8949	60.2957	-95.6089	1788.1459	60.4414
-92.5526	1497.3171	65.2103	-94.8298	1642.7025	63.7920	-99.3425	1787.9424	63.8891
-95.6628	1497.1216	68.7852	-98.1230	1642.5092	67.2139	-102.9313	1787.7393	67.2279
-98.6520	1496.9276	72.2582	-101.3237	1642.3148	70.5632	-106.3773	1787.5376	70.4567
-101.5434	1496.7343	75.6459	-104.4380	1642.1198	73.8401	-109.6842	1787.3376	73.5748
-104.3503	1496.5411	78.9543	-107.4682	1641.9242	77.0433	-112.8599	1787.1399	76.5828
-107.0867	1496.3478	82.1896	-110.4211	1641.7283	80.1707	-115.9112	1786.9447	79.4834
-109.7625	1496.1539	85.3602	-113.2905	1641.5327	83.2197	-118.8377	1786.7524	82.2797
-112.3740	1495.9601	88.4691	-116.0597	1641.3392	86.1812	-121.6352	1786.5642	84.9722
-114.9200	1495.7666	91.5192	-118.7075	1641.1498	89.0407	-124.2988	1786.3809	87.5614
-117.4048	1495.5737	94.5154	-121.2097	1640.9669	91.7771	-126.8313	1786.2029	90.0472
-119.8531	1495.3795	97.4802	-123.5302	1640.7939	94.3587	-129.2321	1786.0306	92.4294
-118.5767	1495.4812	98.8337	-122.4250	1640.8767	95.6216	-128.4031	1786.0905	93.4609

APPENDIX B
STATOR OFFSETS

The following is a list of offsets for the stator blade geometry, with all dimensions in millimeters (mm) and are oriented for a right hand rule with positive Z being along the shaft axis or the axis of rotation for the prop. The stator blade at 0 and 100-percent span are shown in the figure below. Data points for the 0, 50, and 100-percent span locations are shown with the 0-percent span being the hub intersection and the 100-percent span being the shroud or tip intersection. The points are shown starting from the trailing edge location ($Z = 939.079$ mm) of the blade going forward around the suction side of the blade to the leading edge location ($Z = 394.895$ mm) and back around the pressure side of the blade to the starting trailing edge point. The Z-axis represents axial distance along the shaft from the centerline of the propeller, which was located 355.6 mm (14 inches) behind the leading edge of the shroud. The stator sections between the shroud and hub intersections are ruler surfaces, that is, a straight lines can be drawn from corresponding points at the hub and tip to define intermediate section points. Additional stator sections are available at 10-percent span increments in a data file. Seven identical stator blades are used with a clocking for each blade of $1/7^{\text{th}}$ of 360 degrees.



X	Y	Z (axial)	X	Y	Z (axial)	X	Y	Z (axial)
(0 % Span) Hub Section			(50 % Span) Mid Section			(100 % Span) Tip Section		
11.7451	283.0041	993.0790	-7.2410	1088.7092	993.0790	-26.2243	1894.0372	993.0790
11.9141	282.9975	993.0760	-7.0724	1088.7114	993.0845	-26.0560	1894.0414	993.0930
12.0832	282.9926	993.0621	-6.9030	1088.7139	993.0790	-25.8864	1894.0441	993.0960
12.2521	282.9895	993.0371	-6.7332	1088.7164	993.0621	-25.7157	1894.0453	993.0870
12.4206	282.9881	993.0001	-6.5631	1088.7191	993.0341	-25.5441	1894.0451	993.0680
12.5885	282.9885	992.9522	-6.3933	1088.7219	992.9941	-25.3723	1894.0432	993.0360
12.7557	282.9905	992.8942	-6.2234	1088.7250	992.9441	-25.2000	1894.0400	992.9940
12.9219	282.9943	992.8242	-6.0542	1088.7280	992.8821	-25.0276	1894.0351	992.9400
13.0867	282.9998	992.7442	-5.8858	1088.7313	992.8101	-24.8556	1894.0290	992.8760
13.2501	283.0068	992.6543	-5.7183	1088.7346	992.7282	-24.6839	1894.0215	992.8020
13.4118	283.0153	992.5553	-5.5520	1088.7378	992.6367	-24.5131	1894.0126	992.7180
13.5716	283.0254	992.4463	-5.3871	1088.7415	992.5352	-24.3431	1894.0024	992.6240
13.7293	283.0371	992.3284	-5.2238	1088.7451	992.4246	-24.1742	1893.9910	992.5210
13.8846	283.0502	992.2014	-5.0624	1088.7488	992.3051	-24.0069	1893.9784	992.4090
14.0374	283.0644	992.0674	-4.9033	1088.7524	992.1782	-23.8414	1893.9647	992.2890
14.1876	283.0801	991.9254	-4.7463	1088.7562	992.0427	-23.6777	1893.9498	992.1600
14.3347	283.0968	991.7775	-4.5921	1088.7600	991.9012	-23.5164	1893.9340	992.0250
14.4788	283.1147	991.6225	-4.4405	1088.7639	991.7527	-23.3573	1893.9173	991.8830
14.6195	283.1337	991.4615	-4.2920	1088.7678	991.5983	-23.2012	1893.8997	991.7350
14.7566	283.1536	991.2955	-4.1469	1088.7719	991.4387	-23.0480	1893.8814	991.5820
14.8901	283.1743	991.1256	-4.0052	1088.7758	991.2742	-22.8981	1893.8623	991.4230
15.0197	283.1958	990.9516	-3.8671	1088.7797	991.1057	-22.7517	1893.8426	991.2600
15.1453	283.2179	990.7746	-3.7330	1088.7837	990.9343	-22.6089	1893.8225	991.0940
15.2665	283.2407	990.5946	-3.6029	1088.7876	990.7598	-22.4700	1893.8019	990.9250
15.3834	283.2639	990.4136	-3.4770	1088.7914	990.5839	-22.3354	1893.7810	990.7540
15.4958	283.2876	990.2307	-3.3558	1088.7953	990.4058	-22.2052	1893.7597	990.5810
15.8092	283.3237	989.9027	-3.0264	1088.8033	990.0923	-21.8601	1893.7244	990.2820
16.0933	283.3821	989.4458	-2.7199	1088.8127	989.6478	-21.5311	1893.6712	989.8500
16.3436	283.4600	988.8798	-2.4417	1088.8235	989.0919	-21.2250	1893.6028	989.3040
16.5559	283.5543	988.2258	-2.1967	1088.8353	988.4464	-20.9475	1893.5220	988.6670
16.7270	283.6604	987.5129	-1.9898	1088.8474	987.7385	-20.7048	1893.4321	987.9640
17.9744	285.3178	976.6500	-0.1763	1089.0031	976.9045	-18.3264	1892.0340	977.1590
19.2286	286.9331	965.7781	1.6885	1089.1377	966.0710	-15.8519	1890.6350	966.3640
20.4773	288.5176	954.8923	3.5878	1089.2559	955.2296	-13.3035	1889.2331	955.5670
21.7081	290.0543	943.9895	5.5042	1089.3477	944.3763	-10.7030	1887.8271	944.7630
23.2000	291.9428	930.3377	7.8945	1089.4418	930.7903	-7.4164	1886.0628	931.2430
24.6302	293.7721	916.6579	10.2598	1089.5002	917.1785	-4.1185	1884.2897	917.6990
25.9836	295.5558	902.9462	12.5755	1089.5289	903.5361	-0.8430	1882.5070	904.1260
27.2486	297.3028	889.2044	14.8213	1089.5315	889.8622	2.3808	1880.7143	890.5200
28.4359	298.9941	875.4287	16.9997	1089.4988	876.1533	5.5476	1878.9115	876.8780
29.5128	300.6653	861.6179	19.0708	1089.4473	862.4054	8.6102	1877.0982	863.1930
30.4695	302.2780	847.7740	21.0178	1089.3573	848.6186	11.5449	1875.2747	849.4630
31.2890	303.8811	833.8952	22.8176	1089.2528	834.7896	14.3227	1873.4413	835.6840
31.8248	305.4433	819.9883	24.3179	1089.1138	820.9151	16.7858	1871.5984	821.8420
32.2102	307.0001	806.0493	25.6472	1088.9630	807.0002	19.0576	1869.7476	807.9510
32.4491	308.5070	792.0794	26.8045	1088.7793	793.0462	21.1324	1867.8901	794.0130
32.5349	310.0047	778.0803	27.7791	1088.5833	779.0527	22.9952	1866.0263	780.0250
32.4296	311.4649	764.0553	28.5284	1088.3605	765.0221	24.5993	1864.1577	765.9890
32.1656	312.9056	750.0032	29.0812	1088.1215	750.9541	25.9696	1862.2846	751.9050
31.7440	314.3186	735.9271	29.4350	1087.8634	736.8510	27.0998	1860.4080	737.7750
31.1506	315.6965	721.8269	29.5718	1087.5825	722.7140	27.9683	1858.5288	723.6010
30.1687	317.0843	707.7317	29.2714	1087.2983	708.5609	28.3516	1856.6517	709.3900
29.0127	318.4120	693.6194	28.7471	1086.9818	694.3822	28.4613	1854.7743	695.1450
27.6981	319.7494	679.4961	28.0110	1086.6691	680.1826	28.3066	1852.8969	680.8690
26.2106	320.9974	665.3598	27.0464	1086.3110	665.9614	27.8676	1851.0198	666.5630
24.2977	322.2668	651.2525	25.5984	1085.9623	651.7552	26.8874	1849.1507	652.2580
22.2129	323.4706	637.1412	23.9205	1085.5830	637.5396	25.6192	1847.2831	637.9380
19.9779	324.6367	623.0318	22.0315	1085.1887	623.3189	24.0786	1845.4167	623.6060
17.5864	325.7669	608.9215	19.9223	1084.7803	609.0912	22.2538	1843.5508	609.2610
15.9785	326.4004	600.4723	18.4524	1084.5154	600.5681	20.9231	1842.4344	600.6640
14.3085	327.0283	592.0251	16.8962	1084.2495	592.0450	19.4817	1841.3180	592.0650
12.5874	327.6493	583.5778	15.2639	1083.9822	583.5189	17.9389	1840.2007	583.4600
10.7937	328.2410	575.1266	13.5338	1083.7021	574.9893	16.2729	1839.0828	574.8520
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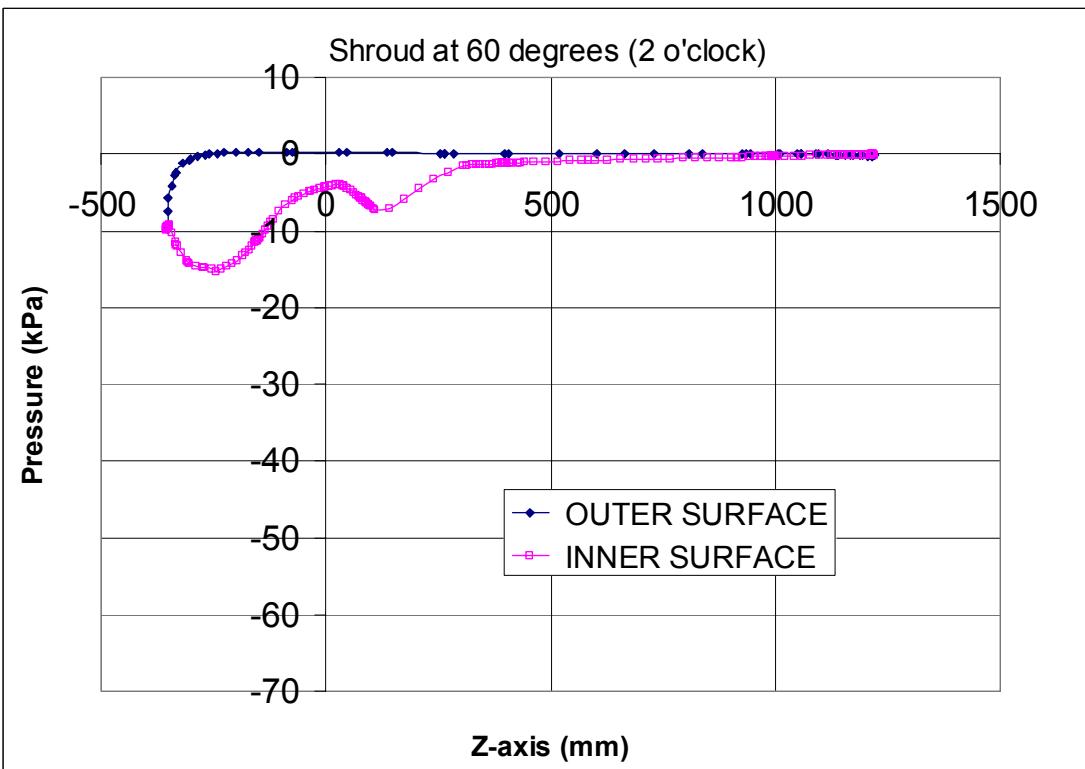
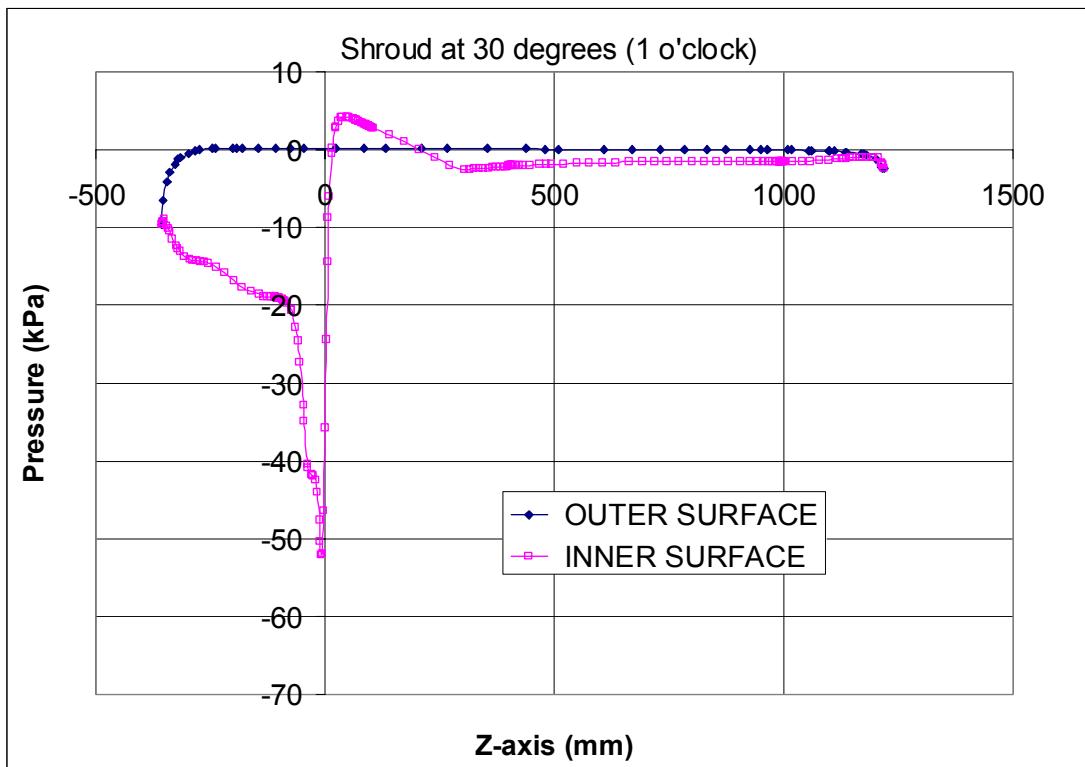
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-4.4997	331.4693	519.1185	-2.0222	1081.5675	518.4423	0.4553	1831.6388	517.7660
-6.3201	331.7240	513.5414	-3.9420	1081.3304	512.8137	-1.5638	1830.8904	512.0860
-8.1930	331.9674	507.9643	-5.9284	1081.0900	507.1877	-3.6632	1830.1402	506.4110
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-18.4622	332.7852	480.1580	-16.9716	1079.7307	479.1850	-15.4763	1826.3656	478.2120
-20.7050	332.8965	474.6220	-19.4076	1079.4436	473.6255	-18.1035	1825.6063	472.6290
-23.1527	332.9721	469.1450	-22.0578	1079.1484	468.1355	-20.9533	1824.8513	467.1260
-25.6649	332.9887	463.6830	-24.7822	1078.8291	462.6685	-23.8866	1824.0947	461.6540
-28.2128	332.9832	458.2301	-27.5511	1078.5032	457.2175	-26.8720	1823.3357	456.2050
-30.8109	332.9531	452.7912	-30.3772	1078.1700	451.7881	-29.9210	1822.5745	450.7850
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-34.9196	332.8455	444.7043	-34.8368	1077.6554	443.7341	-34.7210	1821.4332	442.7640
-36.3291	332.7935	442.0223	-36.3650	1077.4792	441.0677	-36.3638	1821.0520	440.1130
-37.7945	332.7285	439.3614	-37.9493	1077.3007	438.4252	-38.0622	1820.6717	437.4890
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-42.3225	332.4737	431.4185	-42.8306	1076.7489	430.5603	-43.2800	1819.5296	429.7020
-43.9213	332.3443	428.8026	-44.5444	1076.5502	427.9778	-45.1018	1819.1497	427.1530
-45.5486	332.2029	426.1977	-46.2851	1076.3485	425.4094	-46.9484	1818.7695	424.6210
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-50.6490	331.6939	418.4979	-51.7224	1075.7219	417.8205	-52.6949	1817.6271	417.1430
-52.4651	331.4866	415.9890	-53.6502	1075.5044	415.3500	-54.7231	1817.2468	414.7110
-54.2973	331.2663	413.4871	-55.5930	1075.2832	412.8890	-56.7642	1816.8654	412.2910
-56.1835	331.0264	410.9942	-57.5838	1075.0579	410.4480	-58.8461	1816.4845	409.9020
-57.7943	330.8078	409.0332	-59.2726	1074.8724	408.5376	-60.6007	1816.1818	408.0420
-59.4524	330.5724	407.0843	-61.0034	1074.6835	406.6451	-62.3906	1815.8793	406.2060
-61.1414	330.3225	405.1444	-62.7588	1074.4911	404.7647	-64.1980	1815.5764	404.3850
-62.9035	330.0490	403.2315	-64.5809	1074.2935	402.9143	-66.0640	1815.2739	402.5970
-64.3322	329.8152	401.8466	-66.0525	1074.1392	401.5748	-67.5649	1815.0482	401.3030
-65.8194	329.5624	400.4957	-67.5830	1073.9801	400.2638	-69.1238	1814.8220	400.0320
-67.3583	329.2911	399.1757	-69.1663	1073.8162	398.9779	-70.7354	1814.5949	398.7800
-69.0151	328.9859	397.9158	-70.8698	1073.6448	397.7489	-72.4673	1814.3687	397.5820
-70.2514	328.7477	397.1378	-72.1395	1073.5231	396.9914	-73.7562	1814.2197	396.8450
-71.5685	328.4858	396.3998	-73.4920	1073.3961	396.2734	-75.1284	1814.0714	396.1470
-72.9498	328.2032	395.6928	-74.9107	1073.2646	395.5859	-76.5673	1813.9232	395.4790
-74.4166	327.8939	395.0279	-76.4175	1073.1270	394.9409	-78.0951	1813.7757	394.8540
-74.5922	327.8562	394.9569	-76.5977	1073.1107	394.8719	-78.2778	1813.7590	394.7870
-74.7681	327.8181	394.8889	-76.7784	1073.0946	394.8064	-78.4608	1813.7428	394.7240
-74.9442	327.7799	394.8249	-76.9592	1073.0786	394.7449	-78.6440	1813.7271	394.6650
-75.1205	327.7413	394.7649	-77.1401	1073.0630	394.6874	-78.8272	1813.7119	394.6100
-75.2969	327.7026	394.7079	-77.3211	1073.0475	394.6329	-79.0104	1813.6971	394.5580
-75.4933	327.6592	394.6489	-77.5225	1073.0302	394.5769	-79.2142	1813.6812	394.5050
-75.6897	327.6155	394.5959	-77.7241	1073.0133	394.5265	-79.4181	1813.6660	394.4570
-75.8862	327.5715	394.5479	-77.9255	1072.9966	394.4810	-79.6218	1813.6514	394.4140
-76.0826	327.5272	394.5049	-78.1268	1072.9801	394.4410	-79.8253	1813.6375	394.3770
-76.2788	327.4827	394.4679	-78.3279	1072.9639	394.4070	-80.0284	1813.6245	394.3460
-76.4749	327.4379	394.4369	-78.5286	1072.9480	394.3785	-80.2312	1813.6121	394.3200
-76.6704	327.3930	394.4109	-78.7288	1072.9324	394.3555	-80.4333	1813.6005	394.3000
-76.8656	327.3478	394.3899	-78.9285	1072.9171	394.3380	-80.6348	1813.5897	394.2860
-77.0600	327.3025	394.3759	-79.1275	1072.9020	394.3265	-80.8356	1813.5796	394.2770
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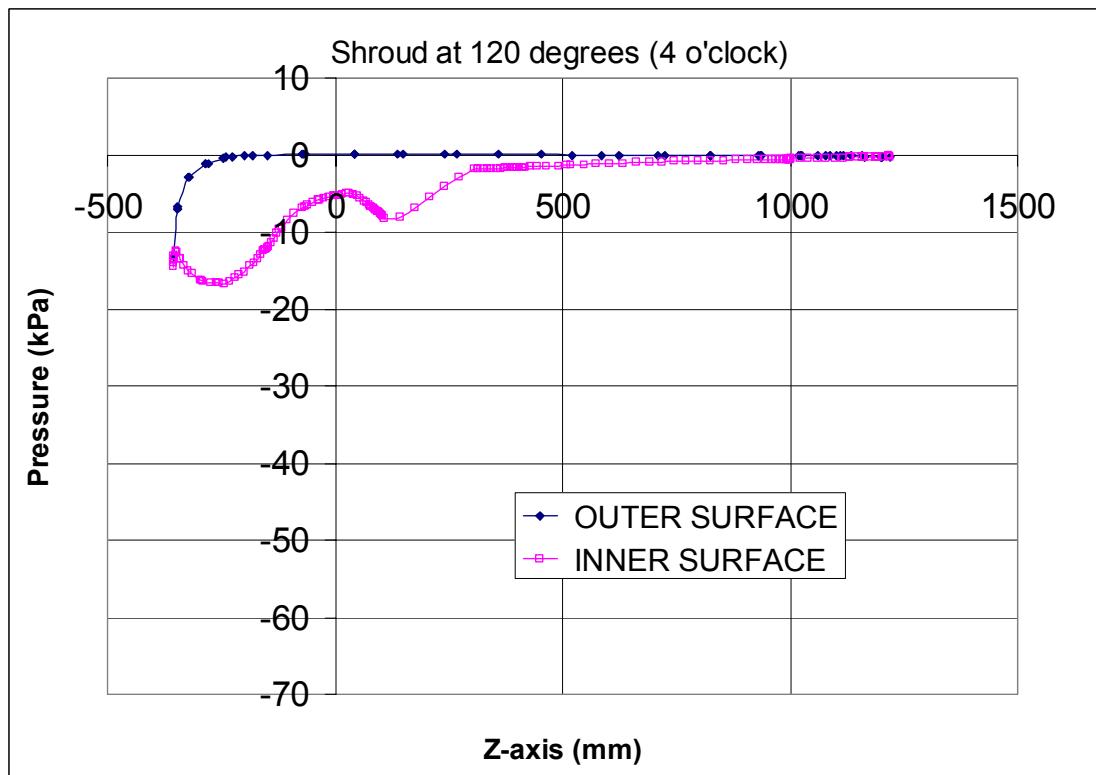
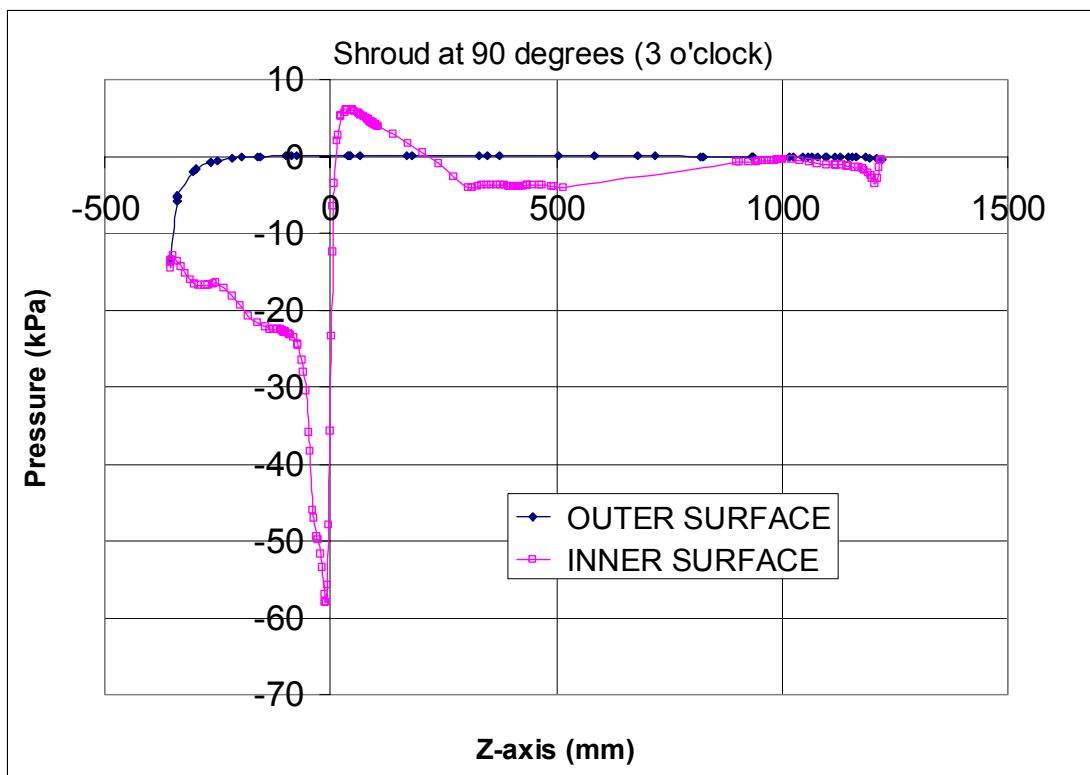
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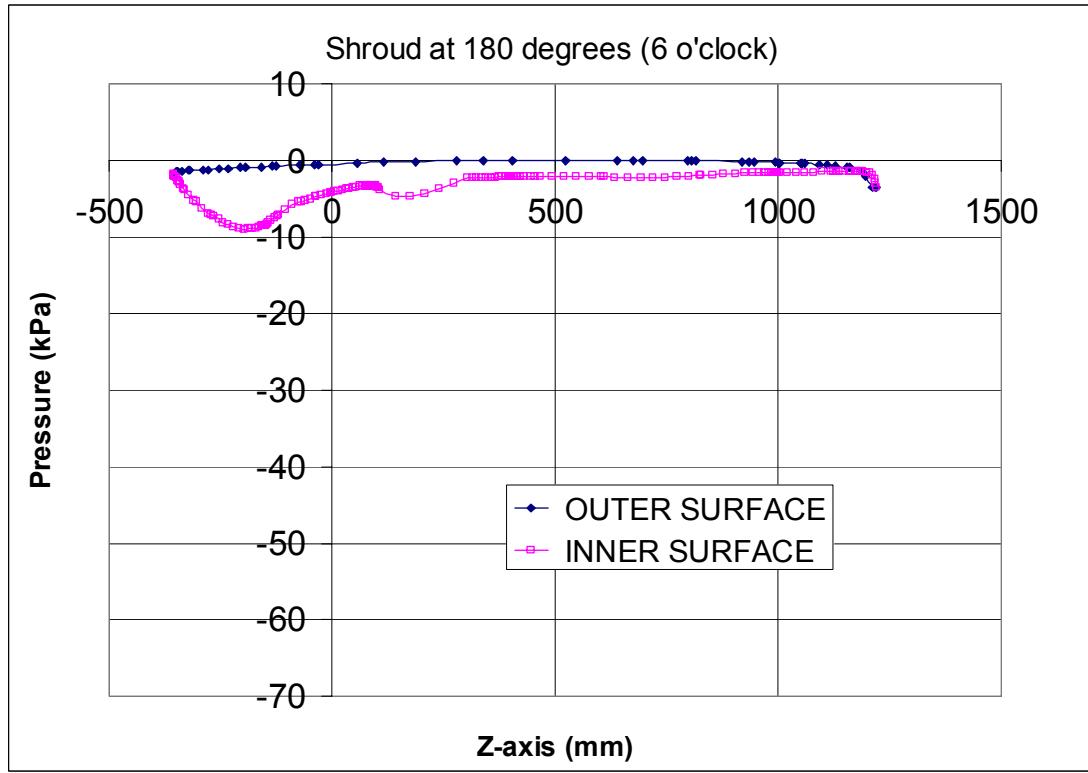
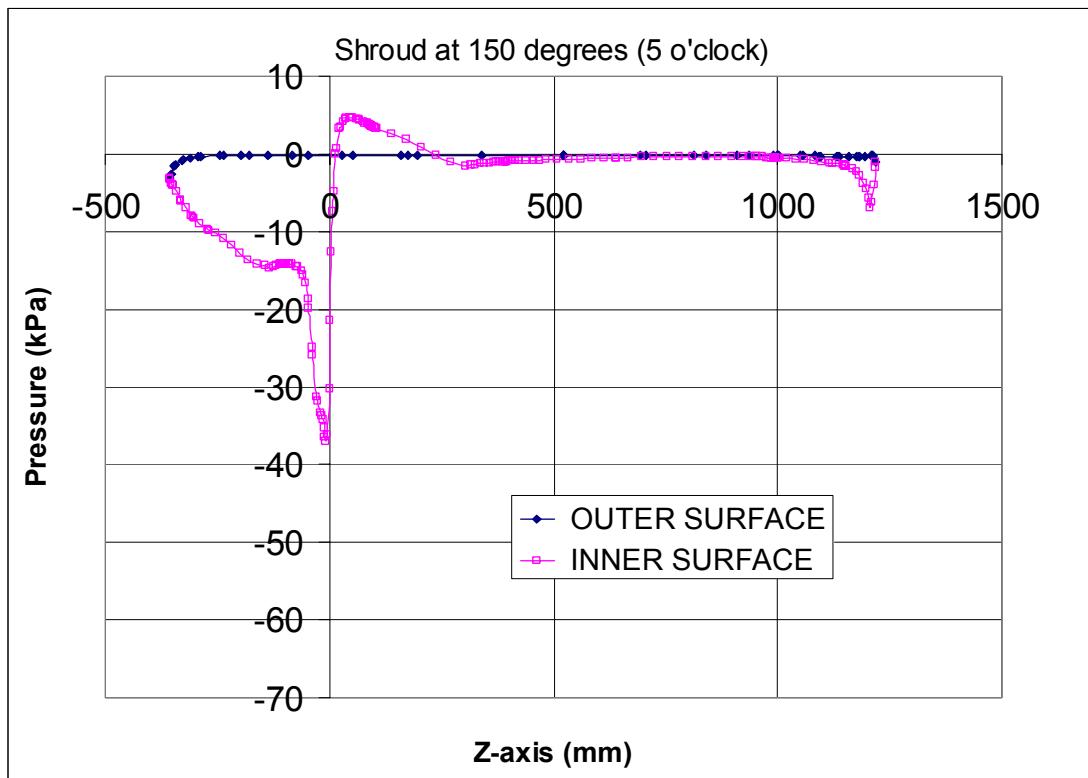
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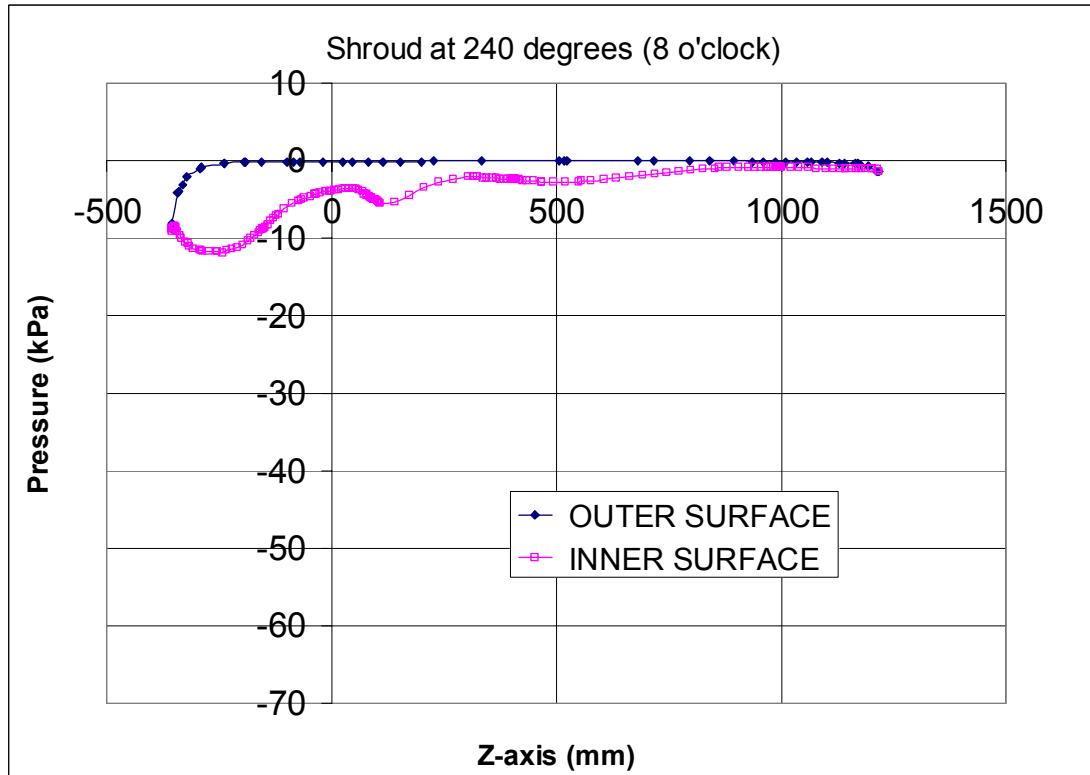
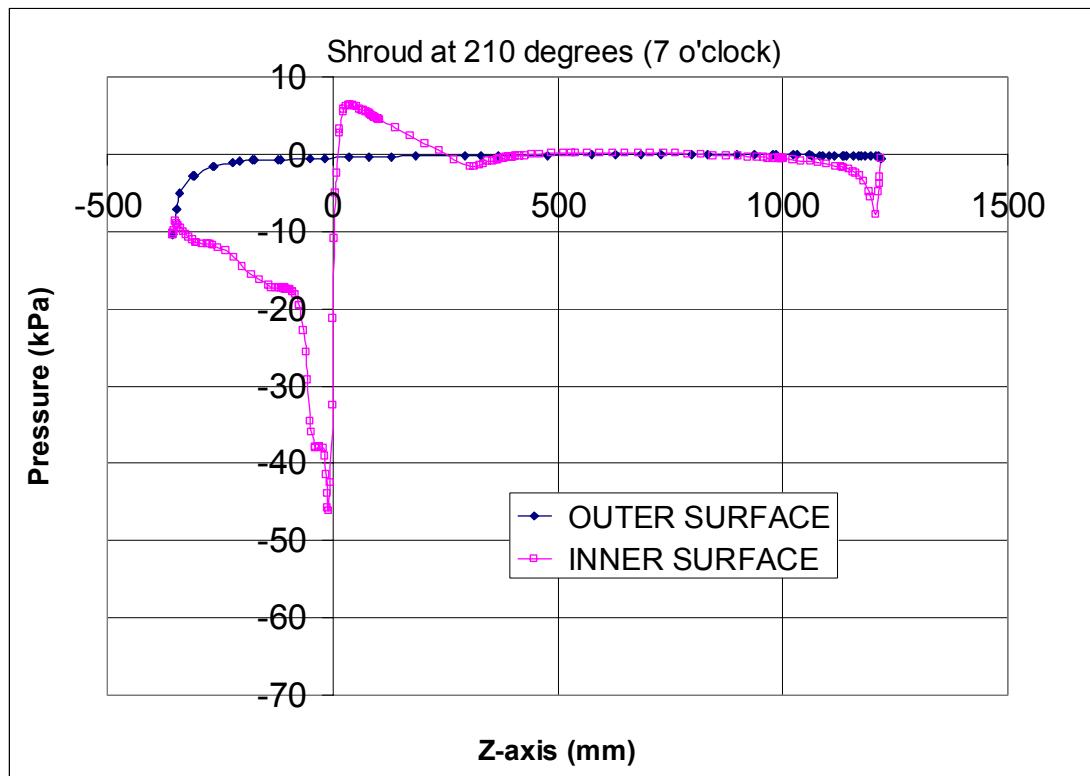
APPENDIX C
SHROUD PRESSURE DISTRIBUTIONS

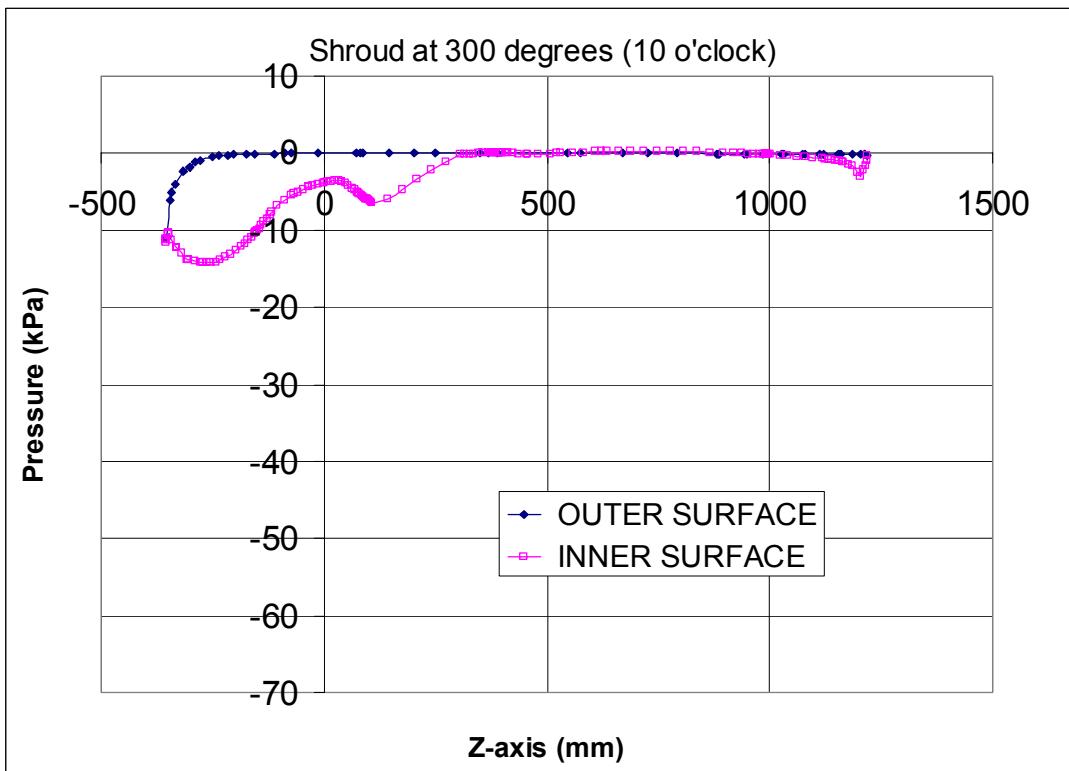
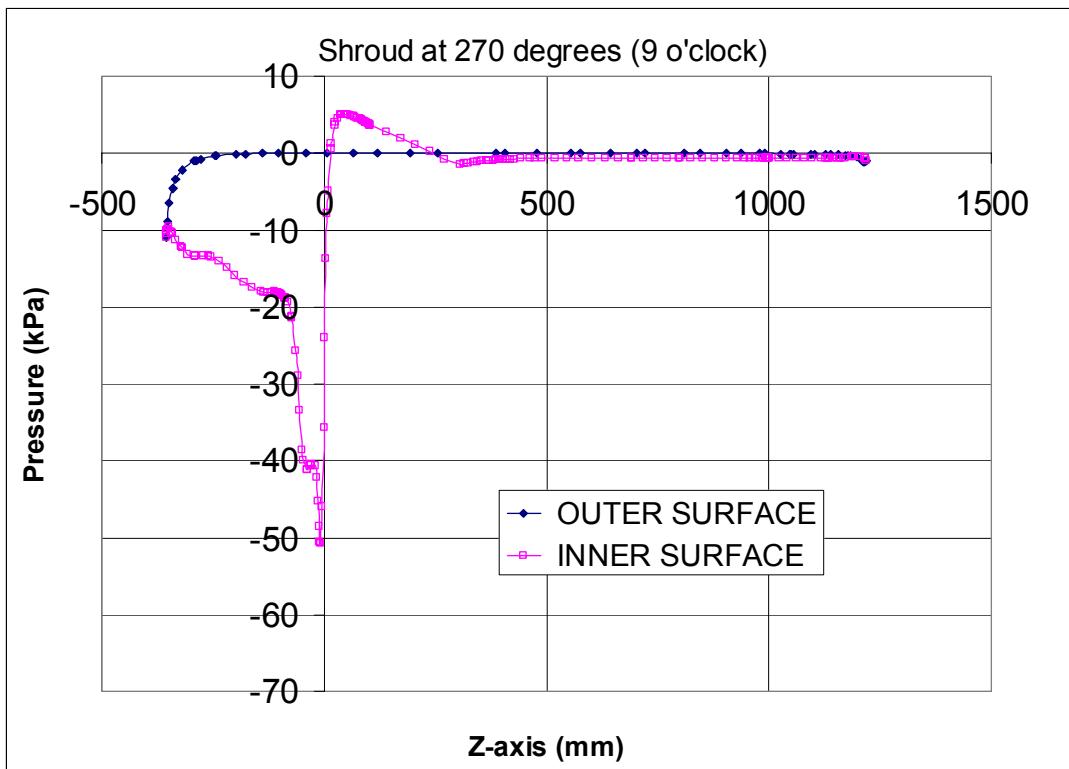
Shroud pressure distributions were obtained from CFD analysis for the maximum power design case. Cutting planes through the shroud were taken at increments of 30 degrees around the full circumference of the shroud (12 places). Pressure distributions at each increment were extracted and are shown in the figures and tables below. Pressure distributions at each increment were separated into the inner and outer surface of the shroud with the leading and trailing edge of the shroud being the dividing point between the two shroud surfaces. The pressure distribution for the inner surface of the shroud is influenced by the location of the propeller. With a six bladed propeller and twelve shroud increments, the inner surface pressure distribution plots show a pressure spike every other increments due to the proximity of the blade. The remaining pressure distribution plots represent the inner surface pressures at locations between two adjacent propeller blades and no pressure spike is occurring.

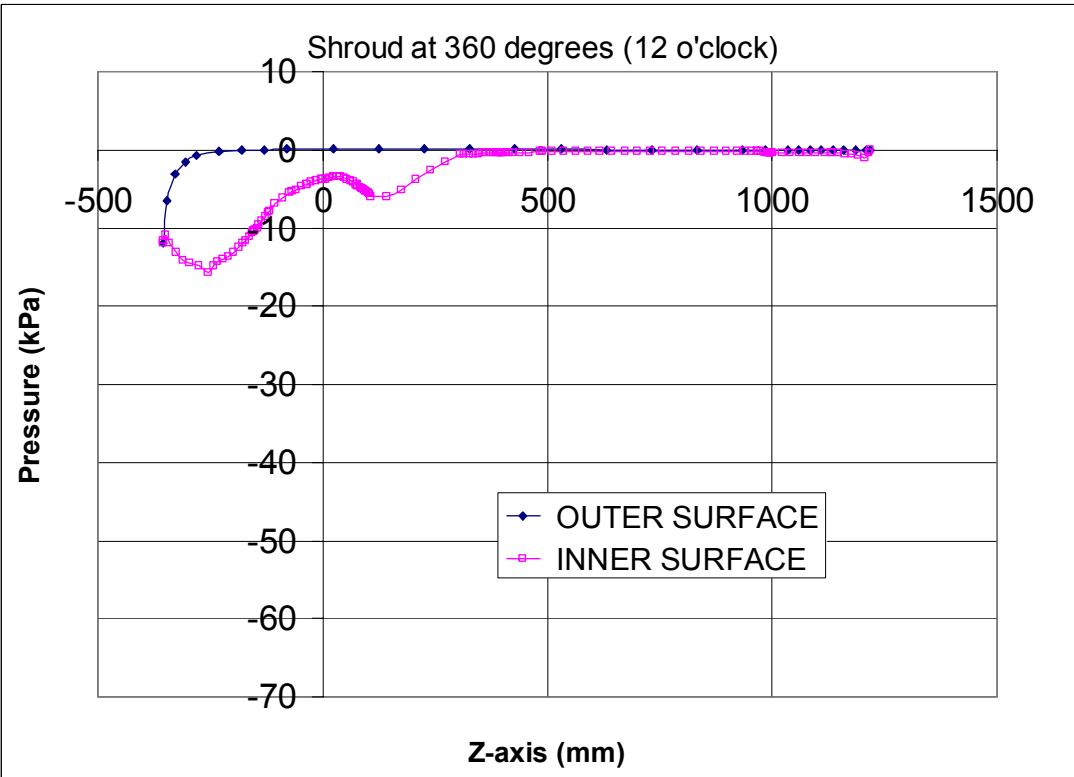
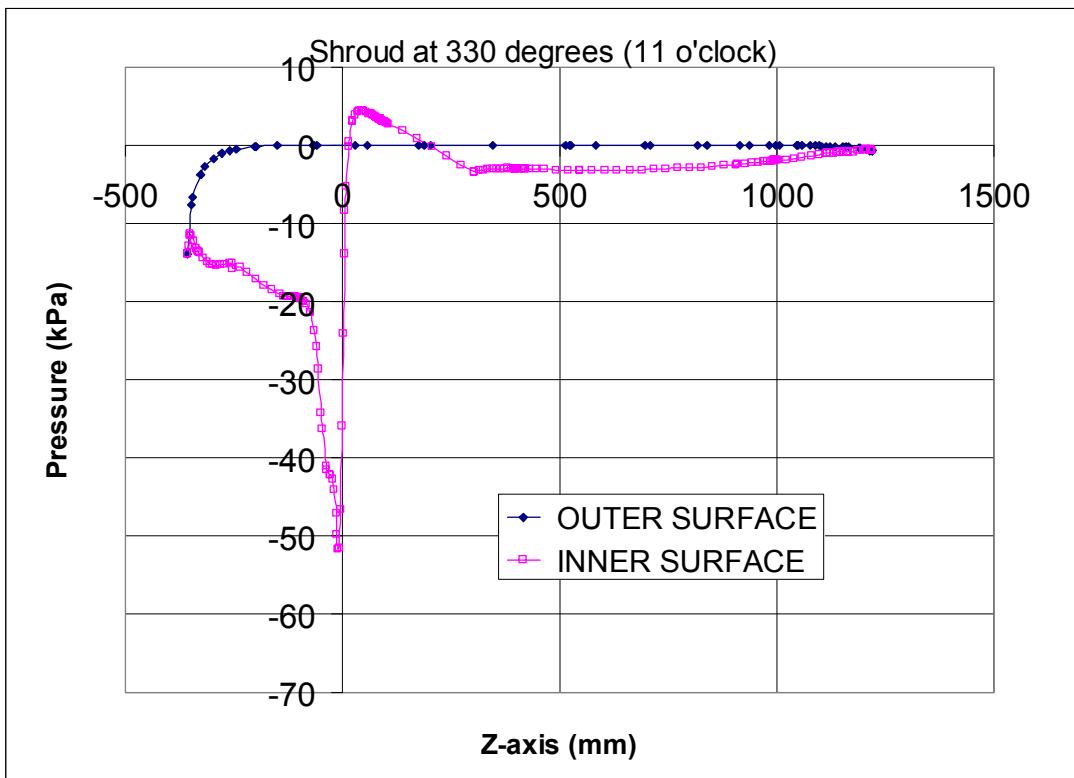








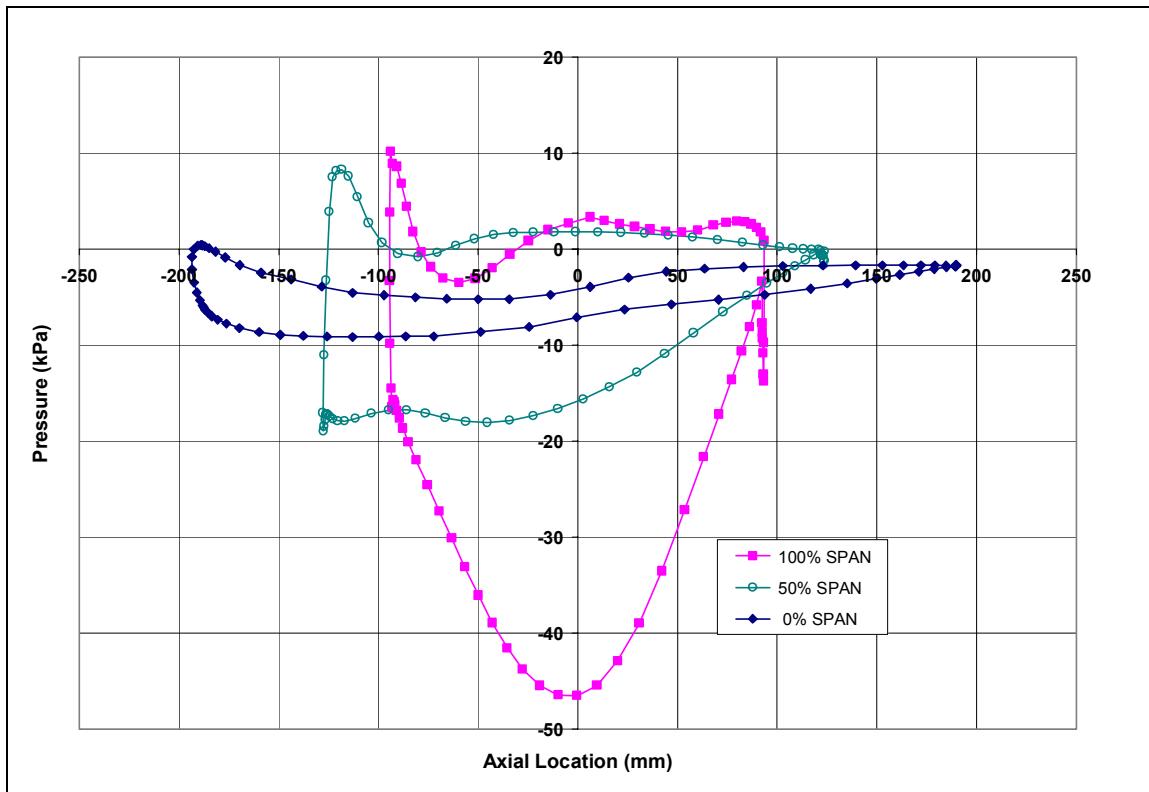




150 degrees OUTER SURFACE			150 degrees INNER SURFACE			INNER SURFACE					
X5 [mm]	Y5 [mm]	Z5 [mm]	Pressure [kPa]	X5 [mm]	Y5 [mm]	Z5 [mm]	Pressure [kPa]	X5 [mm]	Y5 [mm]	Z5 [mm]	Pressure [kPa]
-1730.9585	999.5676	-355.6000	-3.1759	-1730.9585	999.5676	-355.6000	-3.1759	-1561.6952	901.8240	2.0228	-21.4216
-1737.9110	1003.5825	-352.6393	-2.6344	-1727.8279	997.7598	-355.3416	-3.3005	-1561.6957	901.8242	4.2316	-12.6620
-1750.5306	1010.8698	-347.3810	-1.5917	-1712.3516	988.8228	-354.0694	-3.8025	-1561.6919	901.8220	6.1663	-7.5130
-1753.8467	1012.7847	-343.7629	-1.3934	-1694.3826	978.4463	-349.7434	-4.0096	-1561.6987	901.8261	8.9072	-4.8727
-1764.0504	1018.6771	-329.8531	-0.8461	-1693.3605	977.8560	-349.4993	-4.0205	-1561.7056	901.8299	14.0499	0.0255
-1766.7282	1020.2233	-325.3687	-0.7026	-1692.4622	977.3373	-349.1452	-4.0638	-1561.7028	901.8284	15.9148	0.6796
-1772.0791	1023.3133	-309.9512	-0.5547	-1675.0607	967.2885	-342.3165	-4.9138	-1561.7087	901.8318	23.1062	3.2098
-1777.8717	1026.6582	-293.6329	-0.3887	-1659.8438	958.5012	-333.4211	-5.8908	-1561.7095	901.8323	23.6892	3.4097
-1779.3969	1027.5391	-286.4093	-0.3528	-1657.9661	957.4170	-332.3310	-6.0124	-1561.6935	901.8230	30.4599	4.0513
-1784.0276	1030.2131	-245.2481	-0.2473	-1655.9775	956.2687	-330.8145	-6.1365	-1561.7051	901.8297	35.4415	4.5416
-1784.3364	1030.3914	-234.9126	-0.2326	-1641.9277	948.1554	-320.1764	-6.9999	-1561.6968	901.8248	37.9446	4.5716
-1784.2961	1030.3682	-196.7654	-0.2034	-1630.2545	941.4146	-308.8410	-7.8881	-1561.6912	901.8217	45.4913	4.6364
-1784.2343	1030.3325	-178.1952	-0.1934	-1627.2196	939.6620	-305.9171	-8.1250	-1561.7042	901.8292	49.4114	4.6726
-1783.8693	1030.1217	-137.8490	-0.1788	-1624.3668	938.0146	-302.4420	-8.3247	-1561.6879	901.8198	53.0592	4.5982
-1784.1190	1030.2659	-84.3242	-0.1166	-1613.6967	931.8530	-289.5522	-9.0343	-1561.6827	901.8168	60.6281	4.4608
-1784.1169	1030.2648	-46.6043	-0.1544	-1602.9177	925.6285	-273.4145	-9.6859	-1561.7050	901.8295	65.5946	4.3866
-1784.2067	1030.3165	27.1250	-0.1417	-1600.6918	924.3431	-270.1312	-9.8285	-1561.6897	901.8207	68.1989	4.3041
-1784.2135	1030.3204	51.0416	-0.1365	-1599.5647	923.6923	-268.0573	-9.8910	-1561.6660	901.8072	74.3196	4.1184
-1784.1349	1030.2750	160.3327	-0.1279	-1591.9658	919.3042	-254.0000	-10.1671	-1561.6747	901.8121	79.2801	3.9938
-1783.9893	1030.1909	175.1556	-0.1261	-1584.0839	914.7526	-236.6385	-10.9026	-1561.6990	901.8261	83.3060	3.9042
-1783.7056	1030.0271	196.0778	-0.1255	-1577.4421	910.9172	-218.6355	-11.8378	-1561.7105	901.8328	84.8010	3.8707
-1780.1204	1027.9568	338.8274	-0.1264	-1571.9741	907.7597	-200.1110	-12.8542	-1561.6941	901.8233	86.5851	3.8079
-1773.1387	1023.9252	521.4812	-0.1448	-1567.6816	905.2809	-181.1735	-13.6320	-1561.6735	901.8115	89.2604	3.7154
-1773.1277	1023.9188	521.7250	-0.1449	-1564.5612	903.4789	-161.9272	-14.1190	-1561.6614	901.8044	91.4421	3.6437
-1773.1196	1023.9142	521.8618	-0.1449	-1562.6090	902.3516	-142.4712	-14.3682	-1561.6544	901.8004	93.2200	3.5883
-1772.9769	1023.8317	523.9285	-0.1453	-1562.0977	902.0564	-134.2654	-14.6765	-1561.6533	901.7998	93.9366	3.5674
-1761.0333	1016.9346	693.0848	-0.1806	-1561.7759	901.8706	-126.0543	-14.5857	-1561.6520	901.7989	95.1660	3.5318
-1759.9888	1016.3317	705.3793	-0.1828	-1561.7189	901.8377	-117.8418	-14.3662	-1561.6498	901.7977	97.2786	3.4723
-1750.0212	1010.5757	814.7572	-0.2083	-1561.6758	901.8127	-111.5736	-14.1892	-1561.6480	901.7967	100.0575	3.3964
-1747.8546	1009.3245	839.5737	-0.2129	-1561.6538	901.8000	-107.9143	-14.2028	-1561.6476	901.7964	102.8501	3.3199
-1741.2095	1005.4870	910.6636	-0.2318	-1561.6489	901.7971	-105.7753	-14.2341	-1561.6482	901.7968	105.6497	3.2878
-1738.9801	1004.1997	935.7054	-0.2376	-1561.6417	901.7931	-104.6511	-14.2309	-1561.6480	901.7967	138.9626	2.5770
-1734.1692	1001.4216	989.7726	-0.2529	-1561.6348	901.7890	-103.1068	-14.1994	-1561.6480	901.7967	172.2137	1.8014
-1733.3340	1000.9393	998.5383	-0.2556	-1561.6303	901.7864	-100.9906	-14.1599	-1561.6480	901.7967	205.4155	0.8792
-1732.9795	1000.7335	1002.6122	-0.2568	-1561.6335	901.7883	-98.1119	-14.1316	-1561.6480	901.7967	238.5802	-0.1317
-1728.4674	998.1290	1051.4049	-0.2702	-1561.6563	901.8015	-94.1895	-14.1354	-1562.0059	902.0034	271.7156	-1.1203
-1728.0526	997.8894	1056.0310	-0.2718	-1561.7094	901.8322	-89.5687	-14.1901	-1563.3109	902.7570	304.8000	-1.6155
-1725.5693	996.4554	1083.1926	-0.2807	-1561.7033	901.8286	-88.9044	-14.1869	-1563.9482	903.1251	315.2435	-1.4709
-1724.6289	995.9124	1093.2893	-0.2840	-1561.6539	901.8001	-81.8238	-14.2131	-1564.6660	903.5395	325.6793	-1.3697
-1724.4454	995.8065	1095.2798	-0.2848	-1561.7083	901.8315	-72.9244	-14.5501	-1565.7074	904.1408	336.0799	-1.2875
-1721.0143	993.8250	1132.5316	-0.2969	-1561.7067	901.8306	-72.5094	-14.5641	-1566.7484	904.7421	346.4802	-1.2183
-1720.9358	993.7797	1133.3809	-0.2972	-1561.6762	901.8130	-63.4993	-15.1244	-1567.7897	905.3432	356.8799	-1.1574
-1720.7225	993.6567	1135.7280	-0.2983	-1561.6925	901.8223	-59.6756	-15.6290	-1568.9463	906.0111	367.2628	-1.1028
-1718.5548	992.4049	1159.0040	-0.3080	-1561.6799	901.8152	-54.6754	-16.5806	-1569.6375	906.4104	373.4574	-1.0736
-1716.9605	991.4843	1176.4783	-0.3188	-1561.6957	901.8242	-48.3942	-18.7325	-1570.3366	906.8140	379.6594	-1.0465
-1716.5690	991.2582	1180.6647	-0.3216	-1561.6932	901.8229	-45.8579	-19.9257	-1571.0433	907.2222	385.8661	-1.0211
-1716.4213	991.1727	1182.1368	-0.3246	-1561.7015	901.8276	-38.3521	-24.9772	-1571.5891	907.5374	390.6137	-1.0027
-1712.1788	988.7230	1195.1836	-0.2865	-1561.7018	901.8278	-36.8972	-25.9766	-1571.9102	907.7227	393.3898	-0.9921
-1707.8219	986.2070	1209.7656	-0.2515	-1561.7018	901.8278	-29.1905	-31.2127	-1572.0988	907.8317	395.0145	-0.9860
-1707.1399	985.8132	1211.1361	-0.2475	-1561.7034	901.8287	-31.6351	-31.8676	-1572.3412	907.9717	397.1815	-0.9780
-1705.6345	984.9438	1211.8125	-0.3152	-1561.6975	901.8253	-20.7834	-33.3723	-1572.7793	908.2245	401.0392	-0.9648
-1689.2765	975.4977	1219.2000	-0.10730	-1561.7024	901.8281	-17.9788	-33.7905	-1573.4144	908.5914	406.6117	-0.9473
			-1561.6948	901.8237	-13.1285	-34.2715	-1574.2937	909.0992	414.3058	-0.9255	-1686.6095
			-1561.6951	901.8239	-11.7332	-35.2554	-1575.4851	909.7870	424.7013	-0.8993	975.4977
			-1561.6945	901.8236	-9.8786	-36.4085	-1577.0924	910.7153	438.6848	-0.8685	1219.2000
			-1561.6943	901.8234	-7.4077	-36.9337	-1579.2677	911.9714	457.4900	-0.8322	-1.0730
			-1561.6942	901.8234	-4.0962	-36.1641	-1580.4507	912.6546	467.6678	-0.8145	-0.4091
			-1561.6947	901.8236	-0.6866	-30.2174	-1582.1481	913.6347	482.9141	-0.7883	-0.5463

APPENDIX D
PROPELLER PRESSURE DISTRIBUTION

Averaged pressure distributions around the propeller blade at different spans for the full power (5056.6 kW) design condition at 25-knot craft speed and relative air speed of 9-knots is shown in the following figure and tabulations.



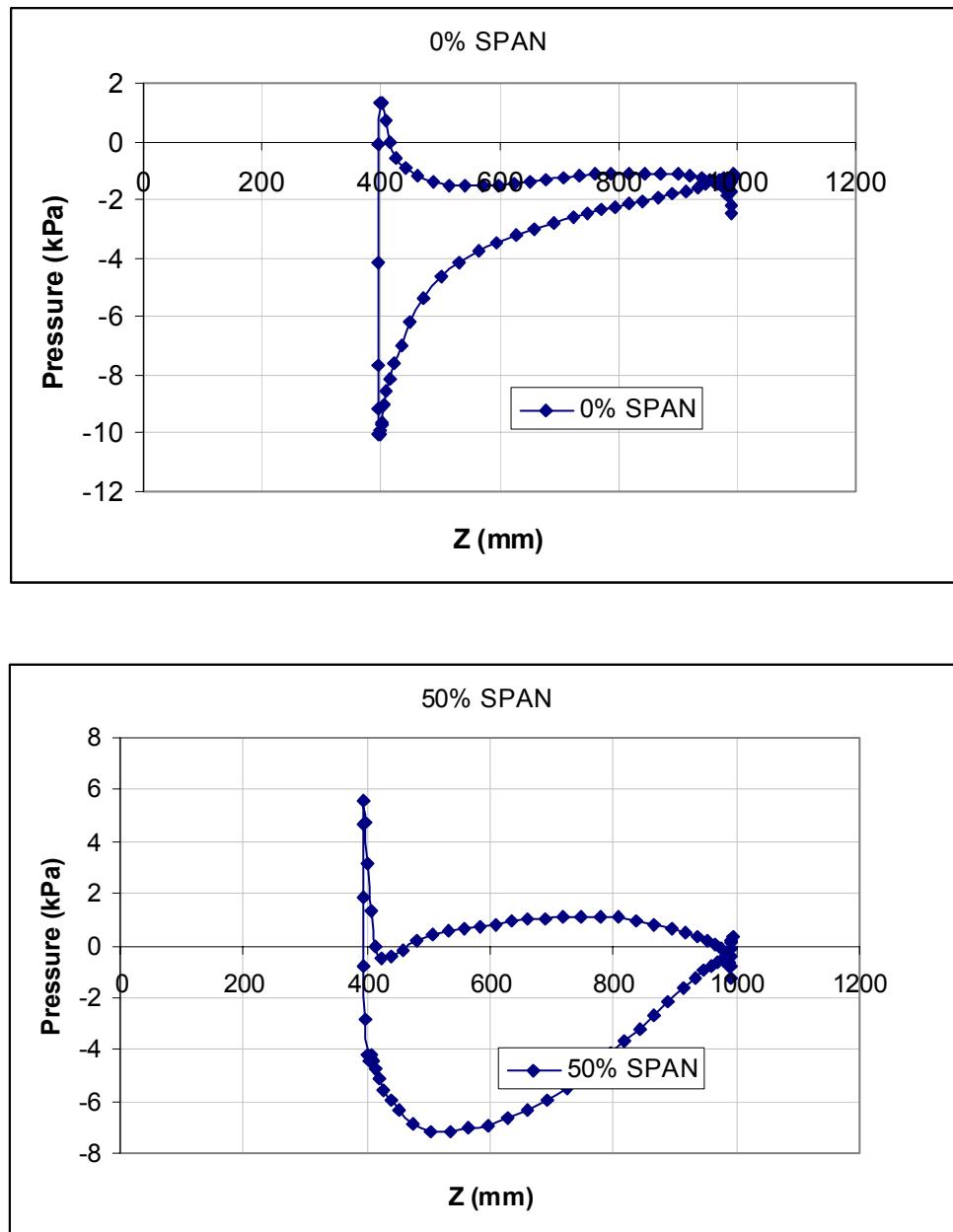
0% SPAN	10% SPAN	20% SPAN	30% SPAN				
Z (mm)	Pressure (kPa)						
184.768	-1.711	172.472	-1.325	160.028	-0.848	146.862	-0.538
179.111	-1.699	167.132	-1.346	155.219	-0.838	142.602	-0.500
172.101	-1.694	160.514	-1.371	149.260	-0.835	137.320	-0.469
163.414	-1.690	152.313	-1.384	141.871	-0.824	130.765	-0.429
152.658	-1.689	142.157	-1.385	132.705	-0.798	122.618	-0.362
139.373	-1.693	129.610	-1.385	121.370	-0.756	112.527	-0.269
123.005	-1.716	114.174	-1.403	107.415	-0.712	100.083	-0.164
102.933	-1.778	95.274	-1.454	90.328	-0.697	84.829	-0.068
83.147	-1.883	76.690	-1.543	73.540	-0.745	69.833	-0.032
63.667	-2.049	58.439	-1.720	57.081	-0.867	55.128	-0.053
44.457	-2.353	40.481	-2.114	40.937	-1.102	40.719	-0.125
25.395	-2.990	22.683	-2.863	24.990	-1.472	26.533	-0.238
6.217	-3.943	4.762	-3.648	9.083	-1.828	12.509	-0.384
-13.632	-4.759	-13.904	-3.821	-7.120	-1.911	-1.419	-0.511
-34.366	-5.223	-33.438	-3.587	-23.833	-1.877	-15.353	-0.652
-49.911	-5.228	-46.964	-3.468	-36.213	-1.929	-26.967	-0.785
-65.612	-5.194	-60.482	-3.580	-48.588	-2.098	-38.634	-0.881
-81.390	-5.028	-74.011	-3.882	-61.040	-2.397	-50.452	-1.032
-97.164	-4.819	-87.643	-4.111	-73.608	-2.757	-62.329	-1.354
-112.878	-4.575	-101.502	-3.968	-86.392	-2.918	-74.244	-1.785
-128.472	-3.917	-115.633	-3.485	-99.579	-2.758	-86.372	-2.043
-143.736	-3.168	-129.892	-2.784	-113.242	-2.303	-99.120	-1.908
-158.826	-2.482	-144.191	-1.720	-127.292	-1.336	-112.570	-1.083
-169.519	-1.678	-154.558	-0.489	-137.913	0.041	-123.098	0.318
-176.810	-0.860	-161.898	0.638	-145.811	1.541	-131.341	2.316
-181.696	-0.267	-166.977	1.422	-151.526	2.736	-137.621	4.159
-184.949	0.087	-170.444	1.818	-155.586	3.334	-142.266	5.064
-187.031	0.288	-172.708	1.951	-158.353	3.463	-145.525	5.173
-188.204	0.390	-174.196	1.938	-160.215	3.365	-147.721	4.900
-189.032	0.442	-175.210	1.855	-161.518	3.124	-149.212	4.314
-190.688	0.345	-178.028	1.076	-164.261	1.676	-151.397	1.635
-192.545	-0.034	-180.314	-0.590	-166.748	-1.104	-153.378	-2.575
-193.539	-0.823	-181.194	-3.028	-168.286	-4.770	-154.834	-7.420
-193.663	-2.122	-179.814	-5.588	-168.071	-8.253	-155.106	-11.582
-192.404	-3.513	-176.686	-7.686	-166.394	-10.425	-154.318	-13.700
-190.939	-4.554	-173.117	-9.109	-164.308	-11.636	-153.195	-14.434
-189.531	-5.327	-169.855	-9.801	-162.272	-12.266	-152.018	-14.778
-188.273	-5.882	-167.171	-10.077	-160.539	-12.346	-150.969	-14.670
-187.656	-6.095	-166.312	-10.154	-159.778	-12.274	-150.295	-14.564
-186.763	-6.334	-165.129	-10.300	-158.717	-12.298	-149.336	-14.527
-185.459	-6.644	-163.472	-10.544	-157.211	-12.492	-147.961	-14.624
-183.518	-7.008	-161.130	-10.846	-155.060	-12.836	-145.978	-14.858
-180.556	-7.361	-157.813	-11.192	-151.963	-13.317	-143.070	-15.230
-176.173	-7.752	-153.050	-11.539	-147.416	-13.823	-136.751	-15.651
-169.662	-8.248	-146.186	-11.798	-140.706	-14.097	-132.243	-15.806
-159.869	-8.666	-136.284	-12.019	-130.810	-14.109	-122.515	-15.530
-149.120	-8.938	-125.840	-12.407	-120.227	-14.246	-112.090	-15.362
-137.663	-9.039	-114.899	-12.882	-109.077	-14.519	-101.111	-15.389
-125.590	-9.119	-103.423	-13.249	-97.329	-14.777	-89.581	-15.526
-112.925	-9.168	-91.421	-13.204	-84.972	-14.809	-77.442	-15.360
-99.759	-9.120	-78.995	-12.822	-72.068	-13.920	-64.773	-14.648
-86.203	-9.084	-66.186	-12.285	-58.720	-13.028	-51.716	-13.748
-72.324	-9.088	-53.090	-11.682	-45.037	-12.078	-38.335	-12.904
-48.565	-8.623	-33.074	-10.103	-27.539	-10.816	-23.148	-11.949
-24.477	-8.112	-12.827	-8.575	-9.778	-9.394	-7.631	-11.086
-0.441	-7.128	7.542	-7.381	8.242	-8.412	8.253	-10.225
23.429	-6.296	28.006	-6.422	26.512	-7.376	24.492	-9.091
47.105	-5.763	48.513	-5.519	44.951	-6.242	41.015	-7.718
70.567	-5.290	69.022	-4.661	63.519	-5.163	57.773	-6.355
93.785	-4.754	89.508	-3.836	82.189	-4.167	74.731	-5.092
116.738	-4.135	109.950	-3.050	100.931	-3.228	91.842	-3.899
135.069	-3.586	126.403	-2.468	116.089	-2.512	105.727	-2.983
149.755	-3.099	139.654	-2.038	128.329	-1.965	116.960	-2.272
161.553	-2.690	150.334	-1.729	138.206	-1.559	126.027	-1.732
171.055	-2.361	158.949	-1.514	146.174	-1.266	133.339	-1.332
178.720	-2.100	165.898	-1.363	152.803	-1.050	139.240	-1.024
184.905	-1.893	171.505	-1.252	157.791	-0.875	144.003	-0.761
189.894	-1.747	176.028	-1.197	161.978	-0.762	147.846	-0.564
189.849	-1.713	176.086	-1.213	162.129	-0.802	148.040	-0.669
189.791	-1.689	176.163	-1.201	162.329	-0.780	148.294	-0.652
189.714	-1.662	176.265	-1.173	162.592	-0.748	148.629	-0.642
189.613	-1.681	176.400	-1.178	162.939	-0.769	149.071	-0.682
189.512	-1.749	176.535	-1.225	163.287	-0.824	149.513	-0.735
189.435	-1.796	176.639	-1.261	163.552	-0.885	149.849	-0.858
189.377	-1.815	176.718	-1.365	163.753	-1.091	150.104	-1.068
189.332	-1.807	176.778	-1.383	163.906	-0.978	150.298	-0.714

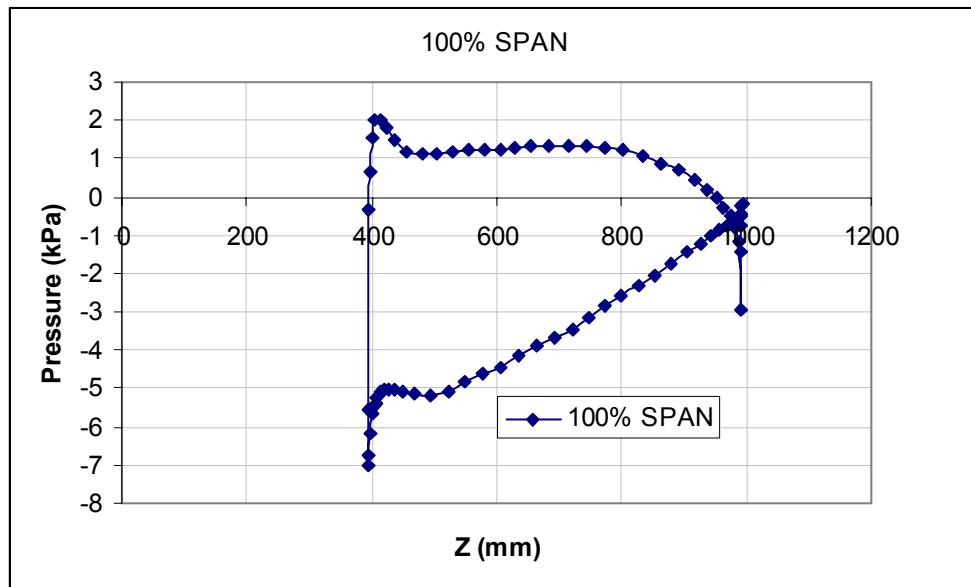
40% SPAN		50% SPAN		60% SPAN		70% SPAN	
Z (mm)	Pressure (kPa)						
133.363	-0.290	120.930	-0.089	110.507	0.060	102.413	0.144
129.596	-0.253	117.488	-0.058	107.253	0.097	99.318	0.178
124.924	-0.212	113.217	-0.012	103.213	0.157	95.471	0.258
119.120	-0.150	107.906	0.064	98.187	0.258	90.676	0.406
111.892	-0.045	101.283	0.199	91.915	0.443	84.685	0.678
102.918	0.100	93.046	0.407	84.115	0.747	77.227	1.115
91.823	0.269	82.828	0.676	74.406	1.163	67.923	1.698
78.185	0.442	70.223	0.974	62.398	1.625	56.412	2.338
64.746	0.574	57.780	1.226	50.561	2.007	45.074	2.865
51.538	0.669	45.533	1.438	38.944	2.322	33.975	3.286
38.570	0.729	33.508	1.606	27.578	2.574	23.155	3.613
25.837	0.757	21.742	1.720	16.508	2.754	12.666	3.854
13.335	0.738	10.259	1.778	5.763	2.863	2.530	4.015
1.069	0.669	-0.921	1.789	-4.611	2.917	-7.202	4.109
-10.970	0.557	-11.806	1.777	-14.627	2.950	-18.539	4.150
-22.035	0.460	-22.097	1.758	-23.974	2.947	-25.116	4.103
-33.040	0.410	-32.178	1.706	-33.050	2.838	-33.383	3.884
-44.066	0.296	-42.066	1.503	-41.831	2.504	-41.322	3.394
-55.008	-0.033	-51.708	1.050	-50.297	1.885	-48.940	2.601
-65.804	-0.604	-61.110	0.365	-58.490	1.093	-56.282	1.677
-76.575	-1.200	-70.396	-0.373	-66.540	0.309	-63.455	0.819
-87.779	-1.444	-79.910	-0.815	-74.756	-0.149	-70.749	0.346
-99.779	-0.881	-90.079	-0.459	-83.429	0.169	-78.395	0.684
-109.334	0.387	-98.207	0.655	-90.348	1.192	-84.458	1.639
-117.102	2.589	-104.915	2.724	-96.047	3.090	-89.426	3.443
-123.313	5.035	-110.435	5.422	-100.751	5.756	-93.487	6.014
-128.139	6.578	-114.912	7.573	-104.663	8.160	-96.896	8.494
-131.723	6.883	-118.454	8.268	-107.889	9.238	-99.772	9.909
-134.228	6.623	-121.106	8.158	-110.472	9.258	-102.175	10.148
-135.927	5.862	-122.969	7.467	-112.389	8.660	-104.095	9.667
-137.830	2.309	-124.626	3.899	-113.733	5.483	-105.370	7.072
-139.471	-3.441	-126.098	-3.275	-114.933	-2.745	-106.373	-1.613
-140.722	-9.670	-127.219	-11.019	-115.795	-11.639	-107.003	-10.985
-141.099	-14.740	-127.726	-17.062	-116.253	-18.665	-107.298	-19.058
-140.617	-16.893	-127.565	-18.965	-116.255	-20.772	-107.211	-22.104
-139.843	-17.069	-127.148	-18.471	-116.044	-19.691	-106.970	-20.841
-139.042	-16.930	-126.695	-17.800	-115.791	-18.492	-106.718	-19.289
-138.327	-16.615	-126.283	-17.210	-115.557	-17.462	-106.496	-18.013
-137.748	-16.508	-125.850	-17.144	-115.228	-17.352	-106.184	-17.823
-136.906	-16.488	-125.198	-17.237	-114.717	-17.436	-105.713	-17.837
-135.678	-16.557	-124.197	-17.458	-113.916	-17.707	-104.993	-18.056
-133.882	-16.678	-122.683	-17.711	-112.666	-18.069	-103.875	-18.420
-131.209	-16.815	-120.376	-17.891	-110.714	-18.392	-102.130	-18.818
-127.228	-16.926	-116.878	-17.913	-107.686	-18.530	-99.434	-19.111
-121.190	-16.819	-111.540	-17.663	-103.050	-18.391	-95.326	-19.227
-112.115	-16.331	-103.479	-17.115	-96.025	-18.052	-89.092	-19.207
-102.441	-15.993	-94.839	-16.765	-88.398	-17.880	-82.304	-19.266
-92.307	-15.973	-85.774	-16.758	-80.357	-17.978	-75.145	-19.526
-81.718	-16.203	-76.321	-17.090	-71.958	-18.409	-67.648	-20.098
-70.601	-16.368	-66.442	-17.579	-63.180	-19.095	-59.793	-20.952
-58.990	-16.161	-56.102	-17.962	-53.974	-19.849	-51.541	-21.951
-46.963	-15.671	-45.292	-18.066	-44.286	-20.425	-42.837	-22.873
-34.560	-15.081	-34.031	-17.858	-34.117	-20.641	-33.675	-23.513
-21.291	-14.388	-22.229	-17.365	-23.440	-20.464	-23.852	-23.761
-7.613	-13.610	-9.963	-16.628	-12.281	-19.923	-13.541	-23.551
6.483	-12.684	2.764	-15.633	-0.620	-18.996	-2.716	-22.791
21.005	-11.532	15.974	-14.374	11.555	-17.625	8.638	-21.383
35.910	-10.050	29.651	-12.841	24.249	-15.922	20.518	-19.424
51.145	-8.363	43.749	-10.913	37.423	-13.762	32.900	-16.986
66.662	-6.647	58.212	-8.743	51.016	-11.186	45.737	-14.011
82.398	-4.996	72.966	-6.557	64.956	-8.480	58.958	-10.758
95.213	-3.743	85.045	-4.868	76.432	-6.330	69.877	-8.103
105.598	-2.777	94.860	-3.542	85.789	-4.591	78.808	-5.919
113.984	-2.040	102.791	-2.523	93.358	-3.227	86.045	-4.165
120.746	-1.485	109.185	-1.754	99.461	-2.198	91.887	-2.810
126.203	-1.047	114.347	-1.149	104.389	-1.395	96.608	-1.742
130.610	-0.663	118.517	-0.612	108.372	-0.691	100.427	-0.824
134.167	-0.367	121.883	-0.194	111.588	-0.127	103.513	-0.054
134.343	-0.579	122.027	-0.602	111.710	-0.809	103.623	-1.079
134.575	-0.572	122.217	-0.602	111.870	-0.837	103.768	-1.164
134.881	-0.577	122.466	-0.609	112.081	-0.845	103.959	-1.180
135.283	-0.611	122.794	-0.637	112.359	-0.864	104.210	-1.253
135.685	-0.662	123.122	-0.710	112.636	-1.009	104.461	-1.450
135.991	-0.901	123.372	-1.086	112.847	-1.505	104.652	-2.098
136.223	-1.093	123.561	-1.249	113.008	-1.612	104.797	-2.202
136.399	-0.463	123.705	-0.256	113.129	-0.122	104.907	-0.041

80% SPAN		90% SPAN		100% SPAN	
Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)
96.831	0.031	93.391	-0.267	91.815	1.746
93.872	0.077	90.666	-0.146	89.723	2.243
90.185	0.201	87.258	0.085	87.106	2.604
85.578	0.433	82.991	0.456	83.837	2.815
79.813	0.825	77.644	1.000	79.739	2.876
72.619	1.397	70.961	1.713	74.597	2.765
63.660	2.126	62.649	2.577	68.167	2.475
52.592	2.949	52.377	3.577	60.165	1.945
41.652	3.652	42.168	4.476	52.187	1.766
30.929	4.209	32.091	5.211	44.268	1.823
20.498	4.625	22.244	5.760	36.411	2.049
10.418	4.931	12.686	6.160	28.642	2.318
0.708	5.158	3.457	6.461	21.004	2.603
-8.602	5.307	-5.424	6.671	13.523	2.929
-17.509	5.354	-13.952	6.729	6.230	3.322
-25.578	5.229	-22.204	6.533	-4.597	2.684
-33.312	4.845	-30.090	5.983	-14.906	1.983
-40.709	4.143	-37.636	5.033	-24.690	0.853
-47.795	3.138	-44.860	3.726	-33.962	-0.555
-54.626	2.005	-51.817	2.270	-42.771	-1.980
-61.283	0.961	-58.591	0.860	-51.195	-3.082
-67.976	0.356	-65.278	-0.121	-59.336	-3.519
-74.968	0.659	-72.154	-0.092	-67.443	-3.055
-80.474	1.551	-77.533	0.661	-73.840	-1.890
-84.964	3.308	-81.862	2.316	-78.493	-0.325
-88.616	5.860	-85.394	5.013	-82.460	1.797
-91.665	8.483	-88.319	8.082	-85.705	4.406
-94.253	10.317	-90.789	10.644	-88.405	6.838
-96.454	10.959	-92.911	12.041	-90.758	8.590
-98.332	10.731	-94.822	12.329	-92.981	8.878
-99.535	9.381	-95.750	13.255	-93.840	10.167
-100.349	1.534	-96.336	7.527	-94.222	3.825
-100.820	-7.408	-96.678	-0.230	-94.297	-3.317
-101.000	-16.306	-96.810	-9.410	-94.050	-9.826
-100.851	-21.741	-96.715	-18.068	-93.595	-14.521
-100.599	-22.302	-96.545	-22.483	-93.160	-16.454
-100.358	-20.942	-96.373	-22.812	-92.792	-16.579
-100.150	-19.171	-96.223	-20.431	-92.494	-15.720
-99.847	-18.894	-95.934	-20.161	-92.100	-15.888
-99.390	-18.825	-95.489	-20.052	-91.525	-16.266
-98.698	-18.975	-94.800	-20.187	-90.683	-18.827
-97.636	-19.289	-93.747	-20.505	-89.449	-17.605
-95.996	-19.682	-92.138	-20.921	-87.646	-18.669
-93.489	-20.056	-89.673	-21.349	-84.992	-20.105
-89.659	-20.341	-85.885	-21.678	-81.099	-21.998
-83.851	-20.514	-80.192	-21.879	-75.410	-24.561
-77.577	-20.739	-74.102	-22.179	-69.417	-27.284
-70.960	-21.182	-67.685	-22.779	-63.170	-30.131
-64.011	-21.946	-60.938	-23.753	-56.658	-33.100
-56.721	-23.005	-53.864	-25.066	-49.867	-36.065
-49.063	-24.259	-46.442	-26.642	-42.781	-38.933
-40.994	-25.538	-38.645	-28.360	-35.388	-41.543
-32.503	-26.669	-30.448	-30.069	-27.668	-43.767
-23.265	-27.496	-21.556	-31.857	-18.967	-45.456
-13.540	-27.838	-12.209	-32.918	-9.863	-48.458
-3.310	-27.476	-2.386	-33.551	-0.315	-46.525
7.431	-26.198	7.902	-33.189	9.689	-45.443
18.667	-24.061	18.644	-31.604	20.060	-42.884
30.403	-21.214	29.862	-28.943	30.893	-38.972
42.615	-17.584	41.544	-24.907	42.134	-33.535
55.243	-13.450	53.635	-19.457	53.681	-27.162
65.666	-10.072	63.589	-14.353	63.156	-21.860
74.193	-7.394	71.712	-10.334	70.882	-17.199
81.113	-5.297	78.304	-7.438	77.159	-13.603
86.708	-3.658	83.638	-5.302	82.245	-10.638
91.238	-2.321	87.964	-3.567	86.375	-8.100
94.907	-1.151	91.477	-2.039	89.730	-5.830
97.880	-0.099	94.334	-0.566	92.455	-3.382
97.985	-1.611	94.431	-2.989	92.537	-7.701
98.123	-1.792	94.559	-3.332	92.645	-8.683
98.304	-1.807	94.728	-3.305	92.786	-9.303
98.543	-1.926	94.949	-3.456	92.973	-10.845
98.782	-2.218	95.171	-3.761	93.159	-13.039
98.963	-3.108	95.339	-4.895	93.301	-13.795
99.101	-3.282	95.467	-5.101	93.408	-9.751
99.206	-0.154	95.565	-0.502	93.490	0.901

APPENDIX E
STATOR PRESSURE DISTRIBUTION

Averaged pressure distributions around the stator are shown below for different blade spans in the figures and tables below. The 0-percent span represents the hub intersection and the 100-percent span is the shroud intersection.





0% SPAN	10% SPAN	20% SPAN	30% SPAN				
Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)
984.504	-1.271	984.511	-1.328	985.518	-0.948	984.543	-0.934
978.032	-1.249	978.146	-1.325	986.890	-0.957	978.244	-0.908
970.009	-1.284	970.257	-1.358	988.209	-0.868	970.439	-0.959
960.063	-1.365	960.476	-1.416	984.801	-0.851	960.765	-1.055
947.725	-1.471	948.348	-1.498	981.278	-0.893	948.772	-1.196
932.423	-1.582	933.303	-1.603	976.848	-0.869	933.901	-1.384
913.432	-1.694	914.635	-1.730	970.577	-0.926	915.453	-1.628
889.857	-1.819	891.460	-1.896	962.723	-0.997	892.552	-1.965
866.241	-1.940	868.244	-2.075	953.004	-1.161	869.609	-2.342
842.585	-2.051	844.985	-2.248	941.758	-1.282	846.620	-2.704
818.890	-2.145	821.677	-2.391	928.846	-1.478	823.569	-2.999
795.167	-2.241	798.330	-2.526	914.450	-1.624	800.466	-3.284
771.430	-2.350	774.961	-2.683	898.365	-2.019	777.332	-3.602
747.684	-2.471	751.580	-2.868	880.757	-2.218	754.171	-3.946
723.943	-2.579	728.196	-3.014	862.070	-2.432	730.999	-4.204
691.650	-2.778	695.922	-3.244	840.724	-2.684	698.421	-4.600
659.435	-2.989	663.712	-3.435	818.522	-2.928	665.895	-4.877
627.367	-3.225	631.659	-3.645	793.952	-3.211	633.528	-5.164
595.434	-3.491	599.735	-3.926	771.597	-3.442	601.303	-5.517
563.765	-3.764	568.065	-4.171	746.437	-3.704	569.349	-5.698
532.423	-4.127	536.766	-4.524	722.171	-3.925	537.814	-5.913
501.444	-4.637	505.809	-5.002	695.795	-4.182	506.662	-6.168
471.157	-5.396	475.518	-5.501	670.371	-4.403	476.251	-6.258
449.512	-6.210	453.904	-5.821	642.944	-4.665	454.592	-6.167
434.022	-6.996	436.391	-6.143	619.544	-4.841	439.060	-6.134
423.000	-7.600	427.320	-6.414	593.602	-5.051	427.982	-6.075
415.212	-8.112	419.416	-6.594	571.289	-5.184	420.063	-5.925
409.662	-8.574	413.803	-6.698	547.074	-5.321	414.453	-5.778
405.756	-9.037	409.771	-6.793	526.942	-5.429	410.399	-5.654
402.981	-9.645	406.914	-7.040	506.563	-5.544	407.537	-5.603
401.863	-9.676	405.359	-7.305	490.482	-5.621	405.956	-5.755
400.444	-9.895	403.361	-7.633	474.301	-5.691	403.911	-5.939
398.674	-10.042	400.890	-8.010	459.862	-5.775	401.364	-6.044
396.684	-10.027	397.932	-7.774	446.104	-5.849	398.302	-5.322
395.076	-9.181	395.634	-6.751	435.849	-5.963	395.875	-3.935
394.383	-7.670	394.448	-4.645	426.145	-5.966	394.531	-1.461
394.548	-4.160	394.549	-0.735	419.989	-5.824	394.437	2.363
395.123	-0.108	395.370	2.408	414.445	-5.361	395.140	4.425
397.288	1.310	397.644	2.923	409.573	-4.607	397.230	4.197
401.344	1.315	401.664	2.480	405.971	-3.716	401.117	3.278
407.209	0.742	407.417	1.343	403.642	-2.849	406.737	1.796
415.363	-0.026	415.390	0.319	402.193	-2.153	414.540	0.575
426.435	-0.574	426.207	-0.247	401.809	-1.807	425.119	0.007
441.390	-0.938	440.807	-0.562	402.286	-1.618	439.382	-0.165
461.570	-1.208	460.500	-0.768	404.005	-1.361	458.600	-0.182
486.870	-1.408	487.127	-0.913	406.947	-1.154	484.562	-0.134
516.080	-1.498	513.690	-0.977	411.665	-0.725	510.488	-0.071
543.247	-1.520	540.217	-0.989	418.646	-0.307	536.411	-0.022
570.414	-1.511	566.741	-0.953	427.468	-0.280	562.325	0.065
597.621	-1.515	593.350	-0.968	438.624	-0.247	588.386	0.077
624.779	-1.475	619.900	-0.964	451.523	0.074	614.385	0.080
651.966	-1.371	646.483	-0.850	466.611	0.357	640.449	0.209
679.230	-1.312	673.167	-0.799	483.241	0.213	666.614	0.276
706.666	-1.260	701.086	-0.763	501.956	0.092	695.077	0.305
734.171	-1.202	729.100	-0.720	522.026	0.005	723.636	0.334
761.696	-1.145	757.132	-0.678	544.211	-0.096	752.239	0.368
789.293	-1.099	785.246	-0.636	567.520	-0.158	780.928	0.400
816.945	-1.096	813.418	-0.649	592.635	-0.171	809.683	0.374
844.629	-1.111	841.630	-0.699	618.488	-0.166	838.482	0.270
872.321	-1.120	869.851	-0.741	645.624	-0.155	867.291	0.172
900.032	-1.140	898.090	-0.789	672.789	-0.150	896.113	0.081
922.394	-1.190	920.877	-0.866	700.265	-0.141	919.368	-0.037
940.434	-1.257	939.259	-0.973	727.779	-0.149	938.126	-0.182
954.983	-1.334	954.084	-1.091	754.818	-0.162	953.257	-0.330
966.715	-1.429	966.041	-1.228	781.258	-0.196	965.458	-0.484
976.178	-1.569	975.683	-1.414	806.693	-0.237	975.301	-0.671
983.811	-1.873	983.460	-1.772	831.124	-0.299	983.239	-0.987
989.808	-2.182	989.614	-2.039	854.206	-0.385	989.537	-1.263
990.515	-2.451	990.343	-2.069	876.001	-0.512	990.284	-1.304
991.449	-1.700	991.323	-1.542	895.941	-0.643	991.288	-0.749
992.491	-1.244	992.426	-1.306	913.793	-0.782	992.418	-0.492
993.072	-1.101	993.076	-1.221	929.822	-0.777	993.077	-0.378
992.304	-1.149	992.304	-1.248	943.770	-0.799	992.294	-0.516
991.228	-1.231	991.201	-1.293	955.825	-0.773	991.187	-0.744
990.289	-1.401	990.243	-1.371	965.584	-0.794	990.239	-1.079
989.608	-1.358	989.642	-1.355	973.398	-0.798	989.526	-0.975

40% SPAN	50% SPAN	60% SPAN	70% SPAN				
Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)
984.603	-0.749	984.633	-0.682	984.591	-0.673	984.541	-0.702
978.257	-0.698	978.160	-0.598	977.959	-0.561	977.780	-0.568
970.394	-0.753	970.142	-0.653	969.745	-0.616	969.410	-0.619
960.651	-0.867	960.207	-0.780	959.571	-0.752	959.043	-0.756
948.575	-1.039	947.895	-0.979	946.965	-0.968	946.203	-0.979
933.603	-1.271	932.634	-1.251	931.343	-1.270	930.294	-1.293
915.032	-1.578	913.707	-1.618	911.970	-1.688	910.569	-1.734
891.981	-2.011	890.217	-2.139	887.929	-2.286	886.093	-2.373
868.888	-2.495	866.682	-2.717	863.841	-2.940	861.568	-3.067
845.744	-2.958	843.090	-3.253	839.687	-3.524	836.971	-3.665
822.533	-3.336	819.419	-3.698	815.439	-4.017	812.263	-4.175
799.262	-3.699	795.685	-4.134	791.126	-4.524	787.485	-4.732
775.955	-4.113	771.905	-4.631	766.758	-5.099	762.647	-5.361
752.617	-4.561	748.091	-5.171	742.352	-5.694	737.764	-5.967
729.263	-4.877	724.259	-5.526	717.924	-6.089	712.853	-6.399
696.823	-5.318	692.282	-5.973	686.530	-6.549	681.917	-6.888
664.435	-5.619	660.363	-6.304	655.203	-6.913	651.054	-7.286
632.200	-5.931	628.586	-6.634	624.011	-7.267	620.324	-7.680
600.113	-6.269	596.970	-6.935	592.996	-7.537	589.781	-7.950
568.313	-6.398	565.658	-7.040	562.305	-7.639	559.585	-8.071
536.937	-6.561	534.761	-7.167	532.020	-7.760	529.790	-8.227
505.975	-6.712	504.315	-7.193	502.219	-7.681	500.506	-8.105
475.781	-6.603	474.651	-6.874	473.221	-7.191	472.056	-7.529
454.305	-6.305	453.581	-6.363	452.641	-6.511	451.870	-6.764
438.912	-6.091	436.486	-5.962	437.910	-5.954	437.439	-6.115
427.937	-5.869	427.726	-5.566	427.409	-5.414	427.150	-5.490
420.090	-5.572	420.030	-5.114	419.894	-4.836	419.785	-4.833
414.534	-5.303	414.582	-4.724	414.575	-4.348	414.572	-4.275
410.514	-5.082	410.636	-4.435	410.722	-4.000	410.795	-3.872
407.678	-4.938	407.849	-4.242	407.993	-3.755	408.113	-3.570
406.098	-5.042	406.265	-4.332	406.409	-3.807	406.528	-3.560
404.051	-5.197	404.210	-4.428	404.350	-3.813	404.463	-3.535
401.494	-5.170	401.633	-4.173	401.762	-3.361	401.865	-3.049
398.417	-4.184	398.531	-2.825	398.641	-1.775	398.723	-1.387
395.957	-2.530	396.032	-0.796	396.108	0.532	396.161	1.124
394.571	0.047	394.609	1.821	394.653	3.173	394.682	3.875
394.398	3.502	394.370	4.673	394.342	5.618	394.320	6.236
395.043	4.979	394.970	5.575	394.887	6.209	394.832	6.678
397.056	4.436	396.918	4.736	396.761	5.160	396.658	5.492
400.889	3.227	400.704	3.167	400.498	3.262	400.359	3.403
406.467	1.595	406.243	1.302	406.003	1.123	405.835	1.073
414.214	0.340	413.942	-0.039	413.661	-0.389	413.456	-0.618
424.715	-0.193	424.373	-0.511	424.032	-0.851	423.772	-1.136
438.865	-0.240	438.420	-0.428	437.991	-0.668	437.649	-0.899
457.919	-0.126	457.325	-0.160	456.766	-0.253	456.300	-0.363
483.652	0.047	482.850	0.165	482.109	0.237	481.472	0.279
509.360	0.188	508.362	0.398	507.448	0.572	506.651	0.711
535.082	0.282	533.899	0.538	532.824	0.759	531.875	0.941
560.802	0.397	559.449	0.681	558.230	0.925	557.147	1.124
586.680	0.434	585.163	0.743	583.805	1.009	582.594	1.224
612.517	0.451	610.858	0.777	609.383	1.057	608.064	1.282
638.424	0.586	636.628	0.917	635.039	1.202	633.616	1.430
664.437	0.665	662.508	1.009	660.811	1.309	659.290	1.548
693.077	0.698	691.304	1.047	689.752	1.352	688.367	1.597
721.816	0.725	720.202	1.073	718.796	1.375	717.551	1.616
750.611	0.756	749.166	1.102	747.915	1.399	746.816	1.634
779.491	0.785	778.215	1.125	777.118	1.415	776.163	1.641
808.438	0.750	807.330	1.080	806.386	1.359	805.572	1.575
837.429	0.628	836.488	0.938	835.694	1.200	835.016	1.399
866.430	0.508	865.657	0.795	865.012	1.031	864.470	1.204
895.442	0.399	894.833	0.664	894.336	0.874	893.926	1.018
918.850	0.267	918.373	0.515	917.994	0.702	917.890	0.819
937.730	0.112	937.362	0.345	937.079	0.511	936.862	0.603
952.961	-0.039	952.680	0.184	952.478	0.333	952.332	0.403
965.245	-0.188	965.037	0.032	964.901	0.170	964.816	0.222
975.155	-0.358	975.006	-0.130	974.924	0.006	974.889	0.048
983.148	-0.635	983.048	-0.387	983.011	-0.261	983.018	-0.250
989.494	-0.906	989.441	-0.651	989.440	-0.511	989.475	-0.494
990.244	-0.959	990.191	-0.785	990.191	-0.796	990.227	-1.028
991.251	-0.365	991.200	-0.106	991.197	0.036	991.226	0.037
992.390	-0.083	992.349	0.221	992.344	0.439	992.363	0.568
993.080	0.040	993.083	0.362	993.085	0.593	993.087	0.746
992.336	-0.159	992.395	0.116	992.423	0.286	992.430	0.353
991.251	-0.503	991.338	-0.389	991.388	-0.418	991.411	-0.574
990.309	-1.043	990.406	-1.241	990.467	-1.632	990.503	-2.136
989.612	-0.814	989.728	-0.777	989.804	-0.800	989.851	-0.866

80% SPAN		90% SPAN		100% SPAN	
Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)	Z (mm)	Pressure (kPa)
984.609	-0.819	984.624	-1.111	984.010	-0.621
977.784	-0.650	977.728	-0.889	976.791	-0.676
969.337	-0.682	969.196	-0.870	967.860	-0.766
958.878	-0.804	958.634	-0.936	956.806	-0.879
945.926	-1.011	945.558	-1.075	943.124	-1.020
929.883	-1.306	929.363	-1.284	926.176	-1.201
909.996	-1.719	909.290	-1.582	905.170	-1.436
885.320	-2.320	884.382	-2.023	879.096	-1.743
860.594	-2.972	859.419	-2.500	852.945	-2.047
835.792	-3.524	834.369	-2.883	826.666	-2.319
810.862	-3.991	809.177	-3.207	800.238	-2.578
785.856	-4.522	783.897	-3.597	773.700	-2.864
760.781	-5.137	758.534	-4.063	747.062	-3.178
735.654	-5.721	733.108	-4.489	720.349	-3.442
710.490	-6.146	707.634	-4.791	693.583	-3.666
679.766	-6.631	677.193	-5.162	664.511	-3.897
649.114	-7.069	646.818	-5.574	635.522	-4.161
618.590	-7.522	616.563	-6.071	606.646	-4.465
588.259	-7.823	586.503	-6.412	577.980	-4.631
558.291	-8.010	556.818	-6.722	549.725	-4.817
528.723	-8.251	527.528	-7.143	521.806	-5.078
499.691	-8.176	498.795	-7.314	494.488	-5.171
471.518	-7.633	470.940	-7.140	468.047	-5.140
451.543	-6.889	451.199	-6.718	449.283	-5.067
437.272	-6.237	437.106	-6.276	435.897	-5.039
427.098	-5.591	427.059	-5.783	426.350	-5.042
419.814	-4.901	419.865	-5.224	419.507	-5.048
414.660	-4.315	414.773	-4.783	414.663	-5.084
410.926	-3.908	411.084	-4.541	411.150	-5.162
408.266	-3.622	408.451	-4.396	408.636	-5.258
406.663	-3.629	406.820	-4.585	407.007	-5.418
404.571	-3.734	404.695	-4.655	404.888	-5.449
401.934	-3.337	402.006	-4.021	402.188	-5.634
398.746	-1.497	398.759	-1.734	398.933	-6.151
396.147	1.389	396.117	1.271	396.245	-6.769
394.656	4.296	394.614	3.882	394.697	-7.002
394.320	6.561	394.327	6.039	394.288	-5.575
394.881	6.703	394.965	6.362	394.821	-0.341
396.746	5.380	396.898	5.186	396.643	0.632
400.451	3.281	400.620	3.219	400.311	1.546
405.905	0.967	406.058	1.039	405.730	1.991
413.486	-0.768	413.603	-0.692	413.260	2.031
423.740	-1.337	423.798	-1.325	423.431	1.820
437.524	-1.086	437.495	-1.117	437.090	1.490
456.043	-0.459	455.887	-0.469	455.413	1.196
481.029	0.300	480.698	0.321	480.123	1.104
506.031	0.816	505.531	0.861	504.856	1.146
531.089	1.080	530.430	1.124	529.665	1.178
556.210	1.273	555.406	1.298	554.564	1.216
581.514	1.381	580.570	1.385	579.655	1.228
606.857	1.443	605.790	1.424	604.815	1.233
632.288	1.588	631.103	1.539	630.071	1.280
657.846	1.711	656.541	1.644	655.450	1.323
687.953	1.758	686.880	1.680	684.900	1.334
716.372	1.774	715.331	1.685	714.459	1.331
745.775	1.785	744.866	1.686	744.097	1.321
775.257	1.782	774.472	1.674	773.795	1.294
804.795	1.704	804.126	1.591	803.525	1.215
834.362	1.510	833.797	1.393	833.257	1.066
863.936	1.290	863.469	1.166	862.978	0.889
893.508	1.075	893.130	0.938	892.676	0.687
917.364	0.849	917.054	0.692	916.623	0.458
936.612	0.606	936.356	0.423	935.941	0.208
952.145	0.379	951.934	0.165	951.530	-0.041
964.681	0.174	964.507	-0.073	964.112	-0.281
974.798	-0.019	974.655	-0.301	974.269	-0.516
982.963	-0.360	982.846	-0.689	982.467	-0.812
989.454	-0.624	989.369	-1.032	989.033	-1.171
990.205	-1.569	990.126	-2.732	989.843	-2.959
991.200	-0.173	991.131	-0.757	990.935	-1.411
992.339	0.561	992.292	0.265	992.207	-0.474
993.089	0.813	993.090	0.574	993.093	-0.168
992.475	0.318	992.513	-0.045	992.438	-0.251
991.484	-0.846	991.535	-1.459	991.365	-0.459
990.595	-2.716	990.657	-3.551	990.441	-0.779
989.954	-0.997	990.015	-1.332	989.723	-0.551

APPENDIX F
SHROUD OFFSETS

The inner and outer surface points for generating the shroud are shown in the table below. The length along the shroud is shown as overall length and as length from the centerline of the propeller, which is located 355.6 mm (14 inches) behind the shroud leading edge. The inner and outer surfaces of the shroud use the leading and trailing edge points of the shroud as the dividing points. All dimensions are in millimeters.

Overall Shroud Length	Length from Prop Centerline	Shroud Inner Radius	Shroud Outer R Radius
0.0000	-355.6000	1998.8530	1998.8530
1.1287	-354.4713	1979.7689	2007.4116
3.9304	-351.6696	1963.3468	2014.7273
8.1546	-347.4454	1947.9410	2021.5071
13.6865	-341.9135	1933.2895	2027.8398
20.4518	-335.1482	1919.3003	2033.7378
28.3960	-327.2040	1905.9420	2039.1854
37.4767	-318.1233	1893.2120	2044.1513
47.6592	-307.9408	1881.1245	2048.5926
58.9144	-296.6856	1869.7050	2052.4558
71.2172	-284.3828	1858.9874	2055.6752
84.5457	-271.0543	1849.0128	2058.1705
98.8806	-256.7194	1839.8296	2059.8415
114.2044	-241.3956	1831.4931	2060.5593
130.5015	-225.0985	1824.0667	2060.5750
147.7574	-207.8426	1817.6231	2060.5750
165.9590	-189.6410	1812.2460	2060.5750
185.0943	-170.5057	1808.0327	2060.5750
205.1519	-150.4481	1805.0979	2060.5750
226.1213	-129.4787	1803.5787	2060.5750
247.9927	-107.6073	1803.4000	2060.5750
270.7570	-84.8430	1803.4000	2060.5750
294.4054	-61.1946	1803.4000	2060.5750
318.9298	-36.6702	1803.4000	2060.5750
344.3223	-11.2777	1803.4000	2060.5750
355.6000	0.0000	1803.4000	2060.5750
370.5757	14.9757	1803.4000	2060.5750
397.6829	42.0829	1803.4000	2060.5750
425.6373	70.0373	1803.4000	2060.5750
454.4325	98.8325	1803.4000	2060.5750
484.0624	128.4624	1803.4000	2060.5748
514.5211	158.9211	1803.4000	2060.4900
545.8031	190.2031	1803.4000	2060.2417
577.9029	222.3029	1803.4000	2059.8172
610.8155	255.2155	1803.4010	2059.2034
644.5357	288.9357	1804.2121	2058.3869
679.0589	323.4589	1806.6152	2057.3542
714.3802	358.7802	1810.7067	2056.0915
750.4954	394.8954	1815.4616	2054.5846
787.4000	431.8000	1820.3205	2052.8194
824.3046	468.7046	1825.1794	2050.8260
860.4198	504.8198	1829.9343	2048.6542
895.7411	540.1411	1834.5848	2046.3183
930.2643	574.6643	1839.1301	2043.8326
963.9845	608.3845	1843.5697	2041.2111
996.8971	641.2971	1847.9030	2038.4677
1028.9969	673.3969	1852.1293	2035.6160
1060.2789	704.6789	1856.2479	2032.6694
1090.7376	735.1376	1860.2581	2029.6413
1120.3675	764.7675	1864.1592	2026.5454
1149.1627	793.5627	1867.9504	2023.4772
1177.1171	821.5171	1871.6309	2020.4987
1204.2243	848.6243	1875.1998	2017.6103
1230.4777	874.8777	1878.6563	2014.8130
1255.8702	900.2702	1881.9995	2012.1074
1280.3946	924.7946	1885.2284	2009.4942
1304.0430	948.4430	1888.3420	2006.9745
1326.8073	971.2073	1891.3391	2004.5489
1348.6787	993.0787	1894.2187	2002.2185
1369.6481	1014.0481	1896.9796	1999.9641
1389.7057	1034.1057	1899.6204	1997.8470
1408.8410	1053.2410	1902.1397	1995.8081
1427.0426	1071.4426	1904.5362	1993.8686
1444.2985	1088.6985	1906.8081	1992.0300
1460.5956	1104.9956	1908.9538	1990.2935
1475.9194	1120.3194	1910.9713	1988.6607
1490.2543	1134.6543	1912.8586	1987.1333
1503.5828	1147.9828	1914.6135	1985.7132
1515.8856	1160.2656	1916.2333	1984.4023
1527.1408	1171.5408	1917.7151	1983.2030
1537.3233	1181.7233	1919.0568	1982.1180
1546.4040	1190.8040	1920.2513	1981.1505
1554.3482	1198.7482	1921.9369	1979.5031
1561.1135	1205.5135	1925.2836	1976.1564
1566.6454	1211.0454	1929.9689	1971.4711
1570.8696	1215.2696	1935.7484	1965.6916
1573.6713	1218.0713	1942.5022	1958.9378
1574.8000	1219.2000	1950.7200	1950.7200

APPENDIX G
PROP/STATOR HUB OFFSETS

The surface points for the prop/stator hub are shown in the table below. The length along the hub is shown from the centerline of the propeller, which is located 355.6 mm (14 inches) behind the shroud leading edge. There is a straight section of hub extending 355.6 mm (10 inches) forward and aft of the centerline of the propeller. The shape of the hub region forward of the propeller hub straight section is a faired surface that will mate to the propeller shaft and will be refined later. The shape of the hub aft of the prop is defined in the table below since this shape must match the defined stator 0% span section. All dimensions are in millimeters.

Hub Length from Prop Centerline	Hub Section Radius
-254.0000	338.0740
0.0000	338.0740
254.0000	338.0740
288.9357	337.9614
323.4589	337.6289
358.7802	337.0602
394.8954	336.2386
431.8000	335.1465
468.7046	333.7966
504.8198	332.2228
540.1411	330.4385
574.6643	328.4564
608.3845	326.2889
641.2971	323.9485
673.3969	321.4472
704.6789	318.7969
735.1376	316.0096
764.7675	313.0971
793.5627	310.0714
821.5171	306.9444
848.6243	303.7284
874.8777	300.4355
900.2702	297.0783
924.7946	293.6696
948.4430	290.2224
971.2073	286.7502
993.0787	283.2666
1014.0481	279.7859
1034.1057	276.3228
1053.2410	272.8923
1071.4426	269.5101
1088.6985	266.1923
1104.9956	262.9556
1120.3194	259.8173
1134.6543	256.7953
1147.9828	253.9078
1160.2856	251.1740
1171.5408	248.6134
1181.7233	246.2463
1190.8040	244.0937
1198.7482	242.1773
1205.5135	240.5204
1211.0454	239.1482
1215.2696	238.0896
1218.0713	237.3823
1219.2000	237.0962
1220.4266	236.7845
1224.8359	235.6573
1232.9518	233.5548
1245.0956	230.3396
1261.5085	225.8569
1282.3868	219.9122
1307.8971	212.2492
1338.1849	202.5155
1373.3797	190.2040
1413.5989	174.5374
1453.8181	156.6759
1489.0129	138.5819
1519.3007	120.3486
1544.8110	102.0714
1565.6893	83.8518
1582.1022	65.8028
1594.2460	48.0610
1602.3619	30.8138
1606.7712	14.3869
1607.9978	0.0000